## Spacecraft Design for Habitability (part 1)

- Lecture \#13 - October 10, 2023
- Required crew volumes
- Interior layouts
- Workstation design
- Habitat optimization


## Designing for Living in Space

- How much room do you need?
- How do you design the habitat shape?
- How do you design the habitat interior?
- Where do you put everything?
- How do you make it livable?
- How do you make it functional?
- How do you make it comfortable?


## How Much Room Do You Need?

- How do we define "room"?
- Floor area (in appreciable gravity)
- Volume (in microgravity)
- How do we define "need"?
- Survival
- Critical functionality
- Comfort
- How does the mission affect the answers?


## Bounding the Problem

| Environment | Relative Gravity |
| :---: | :---: |
| Earth | 1 |
| Mars | 0.38 |
| Moon | 0.16 |
| Minor Bodies | $10^{-2}-10^{-4}$ |
| Orbit | $10^{-5}-10^{-6}$ |

## Mercury Spacecraft Interior Layout



Spacecraft Design for Habitability ENAE 483/788D - Principles of Space Systems Design

## Habitat Design Size

- Typically estimated per crew member
- Microgravity habitat figure of merit $\mathrm{m}^{3} /$ crew
- Partial gravity habitat figure of merit $\mathrm{m}^{2}$ / crew
- Historical analysis
- Most available habitat data is for partial gravity missions
- Classic parametric function: "Celentano curves"

$$
\frac{\text { volume }}{\text { crew member }}=A\left(1-e^{-\frac{\text { duration }}{B}}\right)
$$

- Standard form uses A=5 ("tolerable"), 10 ("performance"), 20 ("optimum") m³/ crew; B=20 days


## Required Space per Crew Member


from Celentano, Amorelli, and Freeman, "Establishing a Habitability Index for Space Stations and Planetary Bases" AIAA 63-139, AIAA/ASMA Manned Space Laboratory Conference, Los Angeles, California, May 2, 1963

## Gemini Spacecraft Cutaway



Spacecraft Design for Habitability MARYLAND

## Gemini 4 Crew Cabin



Spacecraft Design for Habitability

## Early Looks at Performance Data



## Total S/C Pressurized Volume Data



## A Closer Look at Volume Rqmnts

SAE TECHNICAL PAPER SERIES

2008-01-2027

# Testing the Celentano Curve: An Empirical Survey of Predictions for Human Spacecraft Pressurized Volume 

Marc M. Cohen
Northrop Grumman Corporation

## Data from Space Flight Experience

| Spacecraft Type | Category | Number of missions | Max. <br> Mission <br> Duration <br> Days | Min. Mission Duration Days | Max. Volume Per Crew $\mathrm{m}^{3}$ | Min. Volume Per Crew $\mathrm{m}^{3}$ | Max. Crew | Min. Crew |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Mercury | Capsule | 6 | 1.43 | 0.02 | 1.70 | 1.70 | 1 | 1 |
| Gemini | Capsule | 10 | 14.00 | 0.21 | 1.28 | 1.28 | 2 | 1 |
| Apollo CM with and w/o LM | Capsule | 11 | 12.75 | 6.00 | 4.27 | 2.22 | 3 | 3 |
| Apollo LM | Lander | 7 | 3.21 | 1.00 | 3.33 | 3.33 | 2 | 2 |
| Apollo-Soyuz | Capsule | 1 | 9.04 | 9.04 | 3.33 | 3.33 | 5 | 5 |
| Vostok | Capsule | 6 | 5.00 | 0.07 | 5.73 | 5.73 | 1 | 1 |
| Voskhod | Capsule | 2 | 1.08 | 1.00 | 2.87 | 1.91 | 3 | 2 |
| Soyuz | Capsule | 42 | 14.00 | 0.43 | 1.28 | 1.28 | 2 | 2 |
| Shenzhou | Capsule | 2 | 5.00 | 1.00 | 17.00 | 8.50 | 2 | 1 |
| Space Shuttle | Shuttle | 89 | 17.67 | 2.25 | 35.75 | 8.94 | 8 | 2 |
| Shuttle- <br> Spacelab/SpaceHab | Shuttle | 25 | 16.90 | 4.00 | 42.70 | 14.66 | 8 | 5 |
| Skylab | Station | 3 | 84.00 | 28.00 | 120.33 | 120.33 | 3 | 3 |
| Salyut | Station | 17 | 237.00 | 16.00 | 55.25 | 33.50 | 3 | 2 |
| Mir | Station | 25 | 437.75 | 72.82 | 181.35 | 45.00 | 3 | 2 |
| ISS | Station | 12 | 195.82 | 128.86 | 201.13 | 85.17 | 3 | 2 |

U N I V E R S I TrymCohen, "Testing the Celentano Curve:..."SAE 2008-01-2027
Spacecraft Design for Habitability
MARYLAND

## Correlation to Space Flight Experience



## Straight Power-Law Curve Fit (2008)


from Cohen, "Testing the Celentano Curve:..." SAE 2008-01-2027
Spacecraft Design for Habitability
UNIVERSITY OF MARYLAND

## Cohen's Fit to Data Maxima

Pressurized Volume Per Crew Member Versus Mission Duration: Maxima for Mission Durations for Every Crew Size in Each Spacecraft Configuration


UNIVERSITY OF MARYLAND

## Historical Data Fitted to Celentano Curves



## Historical Trends - Habitat Volume



## Data from 24 Lunar Mission Concepts



## Focusing on Smaller Habitats



Spacecraft Design for Habitability ENAE 483/788D - Principles of Space Systems Design

## Curve Fit to Small Hab Data Only



Spacecraft Design for Habitability ENAE 483/788D - Principles of Space Systems Design

## Required Habitat Volume vs. Crew Load



## Restricting Data to Durations $\leq 180$ Days



## Restricting Data to Crew Loads $\leq 200$



## Apollo Command Module Interior

## APOLLO COMMAND MODULE INTERIOR

## LEFT SIDE



## Apollo Command Module Interior



## Apollo Command Module Interior



## Apollo Spacecraft (Rescue Configuration)



Spacecraft Design for Habitability ENAE 483/788D - Principles of Space Systems Design

## Soyuz Interior



Spacecraft Design for Habitability

## Space Shuttle Flight Deck



Spacecraft Design for Habitability

## Space Shuttle Mid-Deck Panorama



Spacecraft Design for Habitability

## Space Shuttle Mid-Deck



Spacecraft Design for Habitability ENAE 483/788D - Principles of Space Systems Design

## Orion Seating Arrangement



## Orion (Mockup) Interior



## Orion Developmental Seat System



Spacecraft Design for Habitability

## Boeing CST 100 Notional Interior



## Starliner (Mockup) Interior



## Starliner Interior (Training)



## Crew Dragon Interior (Prototype)



## Crew Dragon Interior (Crew-2)



Spacecraft Design for Habitability

## Dragon Hull and External Items



Spacecraft Design for Habitability I ENAE 483/788D - Principles of Space Systems Design

## Limitations to Internal Outfitting

- Never put anything inside a pressurized crew volume that could hurt the crew or threaten the mission
- Toxic substances (e.g., propellants, ammonia coolant)
- Cryogenic fluids (e.g., LOX)
- Pressurized gases that would asphyxiate the crew or overpressure the crew compartment (e.g., GN2)
- Fire or explosion risks (e.g., batteries)
- Experience indicates that it is easier to build and test items external to the pressure hull anyway


## CRS-7 Falcon 9 In-Flight Failure

## Orion Control Panel Concept



## Starliner Control Panel



## Crew Dragon Control Panel



## Space Shuttle Flight Deck Interfaces



## International Space Station



## Inflatable Lunar Habitat Concept (Vertical Layout)



## Horizontal Habitat Interior



## Pressure Vessel Shape and Orientation

- Pressure hull - assumed to be cylindrical with ellipsoidal end caps
- Toroidal configurations modeled as low L/D cylindrical
- Orientation of internal outfitting could be "horizontal" (floors parallel to long axis) or "vertical" (floors perpendicular to long axis)
- Assumption of consistent internal orientation
- Enforced by physics for partial gravity systems
- Standard practice for microgravity systems due to strong crew


## Challenge to Maximize Habitable Volume

- Assume "habitable volume" involves standing headroom
- Human volume is rectangular; pressure vessels are curved



## Habitat Layout - Vertical or Horizontal?

- Geometric modeling of "packing factor" to fit humans into cylindrical shapes

- Mass estimation for human-rated pressurized volumes from JSC-26096 (converted to metric)

$$
M<k g>=13.94\left(A_{\text {surface }}<m^{2}>\right)^{1.15}
$$

Spacecraft Design for Habitability

## Habitat Layout Trades - Floor Area



## Habitat Layout Trades - Useful Volume



## Habitat Layout Trades - Accessible Wall


——Vertical, 1 floor -Horizontal, L=3 m - Horizontal, L=5.5 m -Horizontal, L=8 m - Vertical, 2 floors $\rightarrow$ Design Point

Spacecraft Design for Habitability MARYLAND

## Habitat Layout Trades - Total Volume



## Internal Layout for Horizontal Habitats




## Endcap Effect on Available Floor Area



## Effect of Endcap Shape on Floor Area



—Horiz. 1 floor —Horiz. 2 floors-Horiz. 3 floors—Horiz. 4 floors
-Vert. 1 floor —Vert. 2 floors —Vert. 3 floors —Vert. 4 floors



## Endcap and Cylinder Effects on Mass



