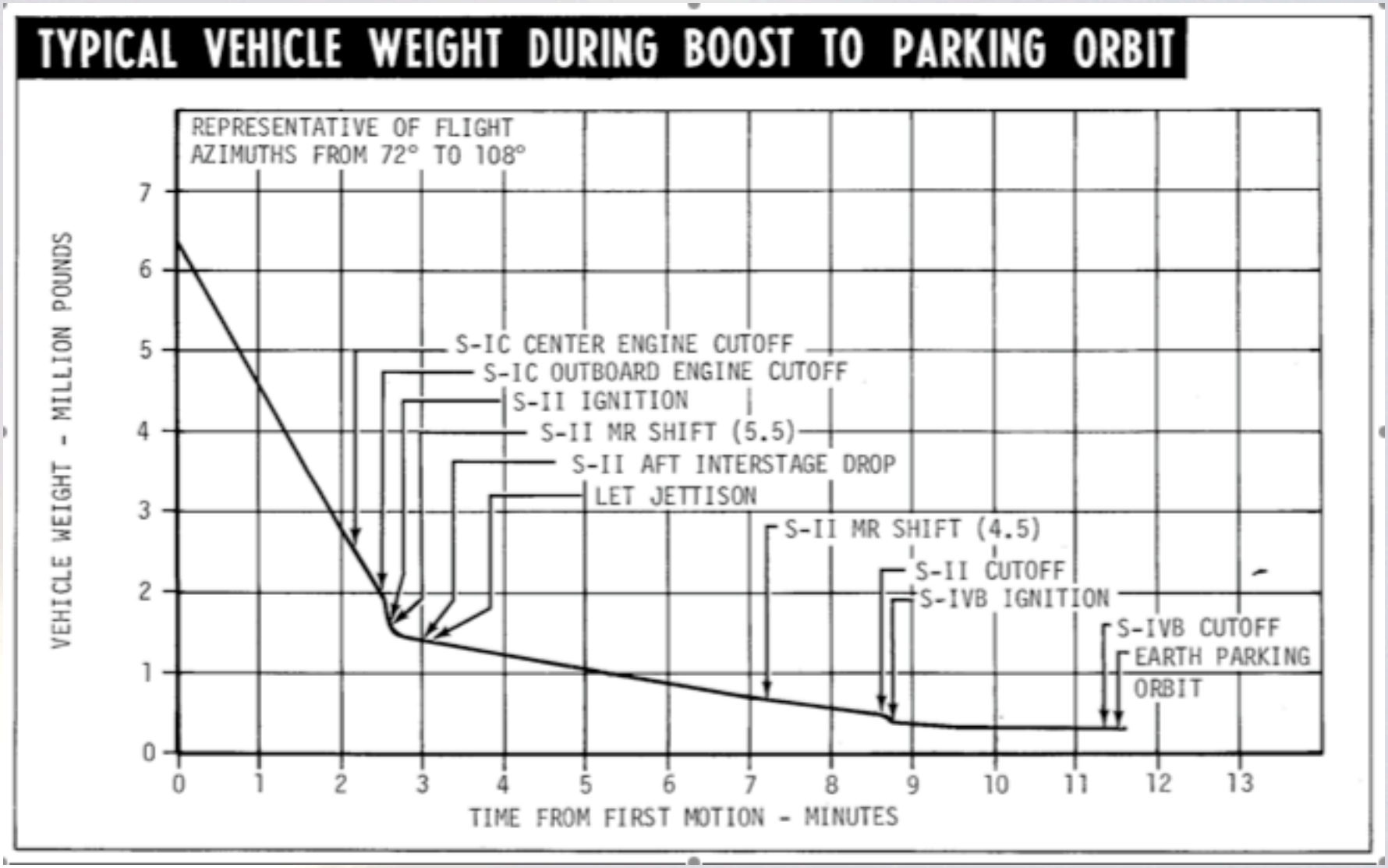
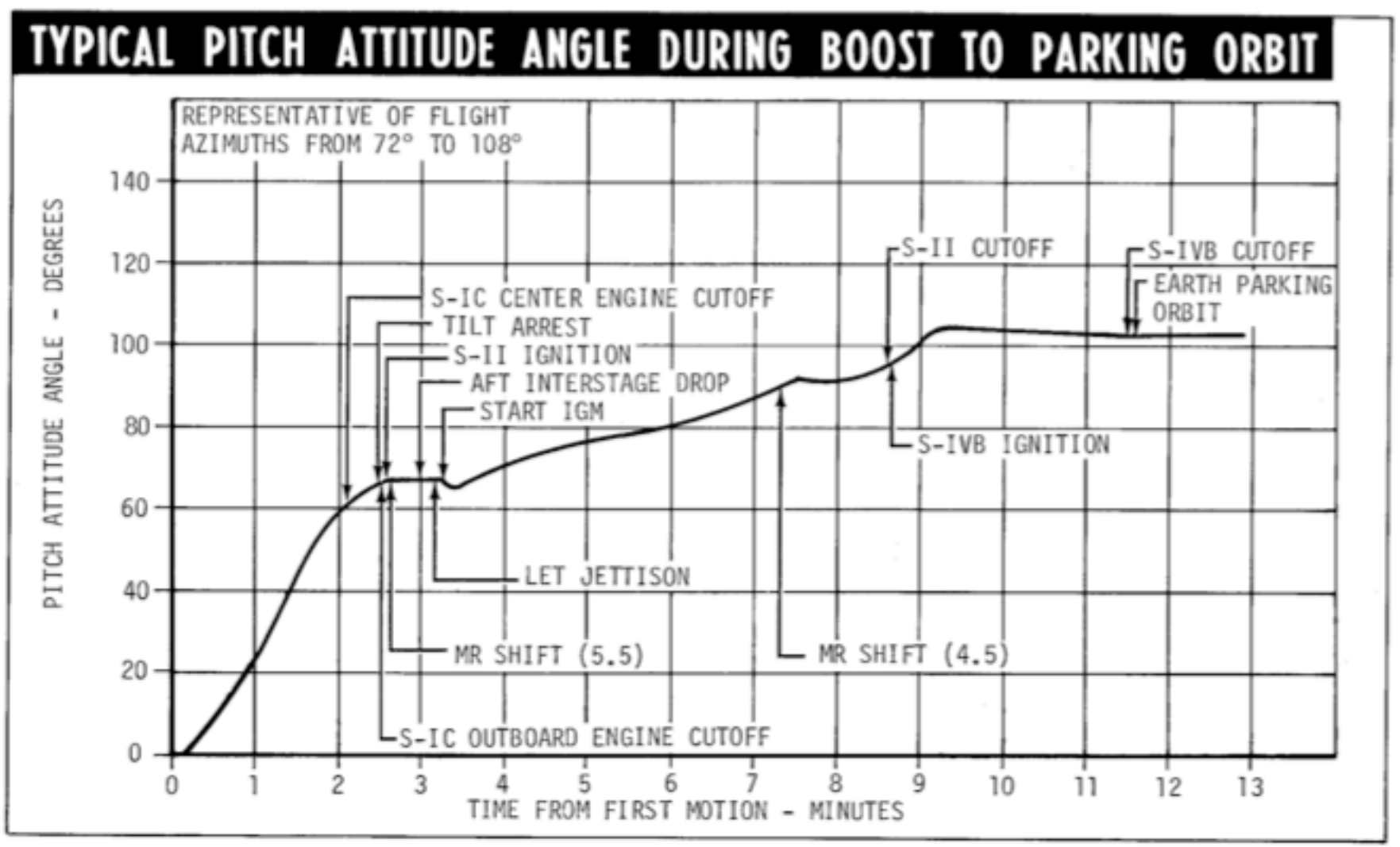
Launch Vehicle Systems

- Case Study: Saturn V
  - Data is from SA-503 Saturn V Flight Manual, MSFC-MAN-503, NASA TM-X-72151, November 1968
- Trajectory and dynamics
- Onboard systems
- Ground systems

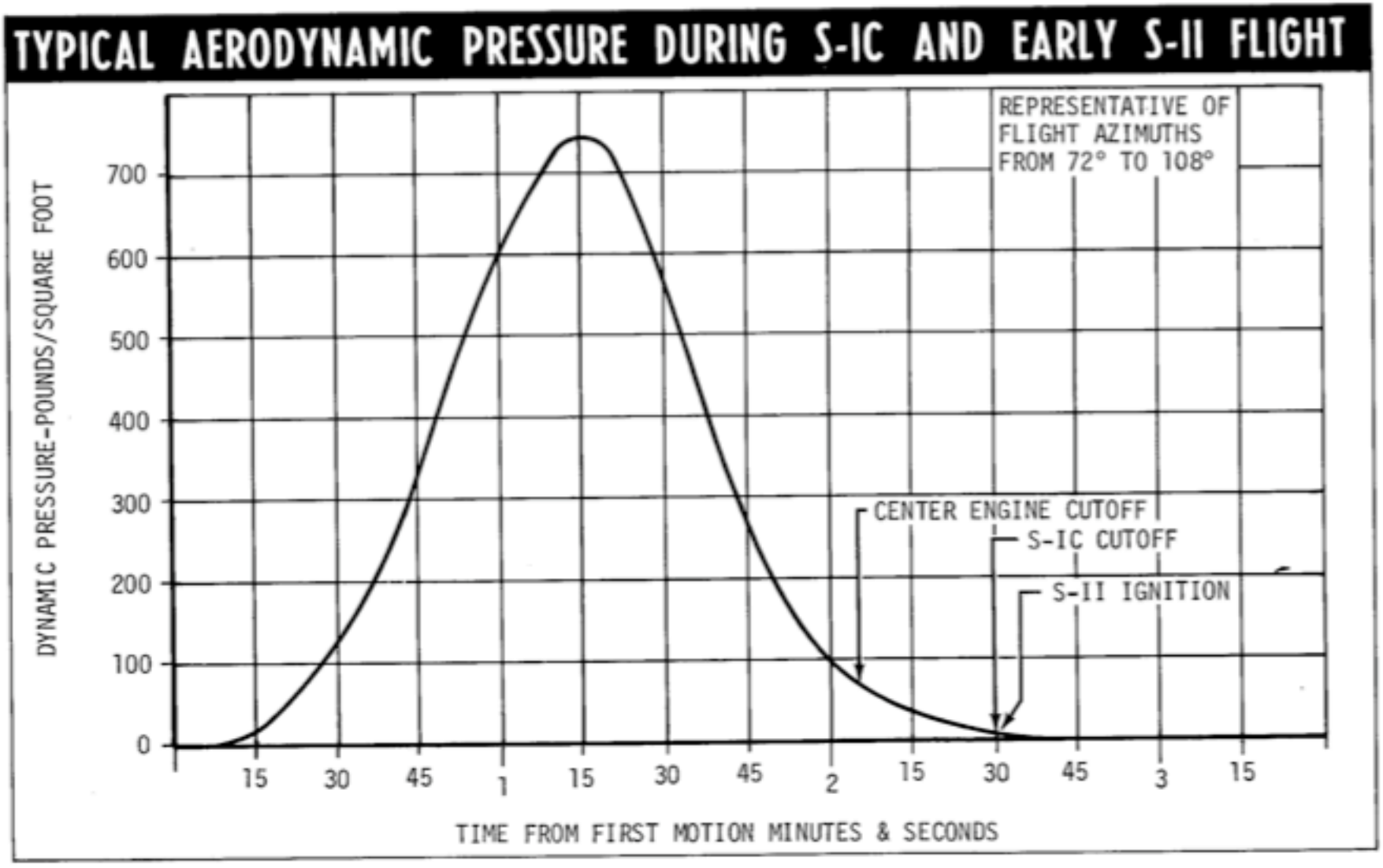
Mass Changes During Launch



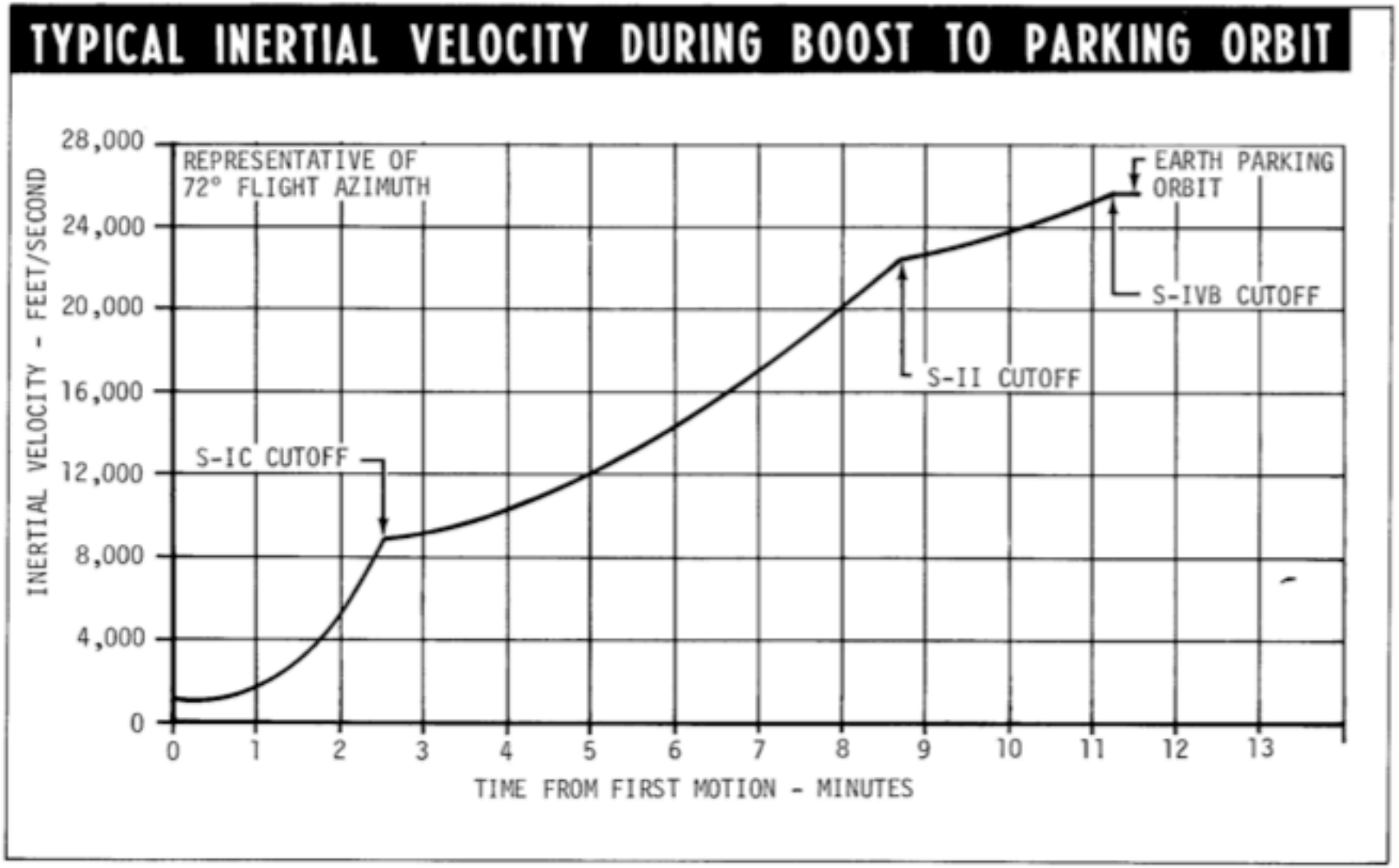
Pitch Attitude Angle During Launch



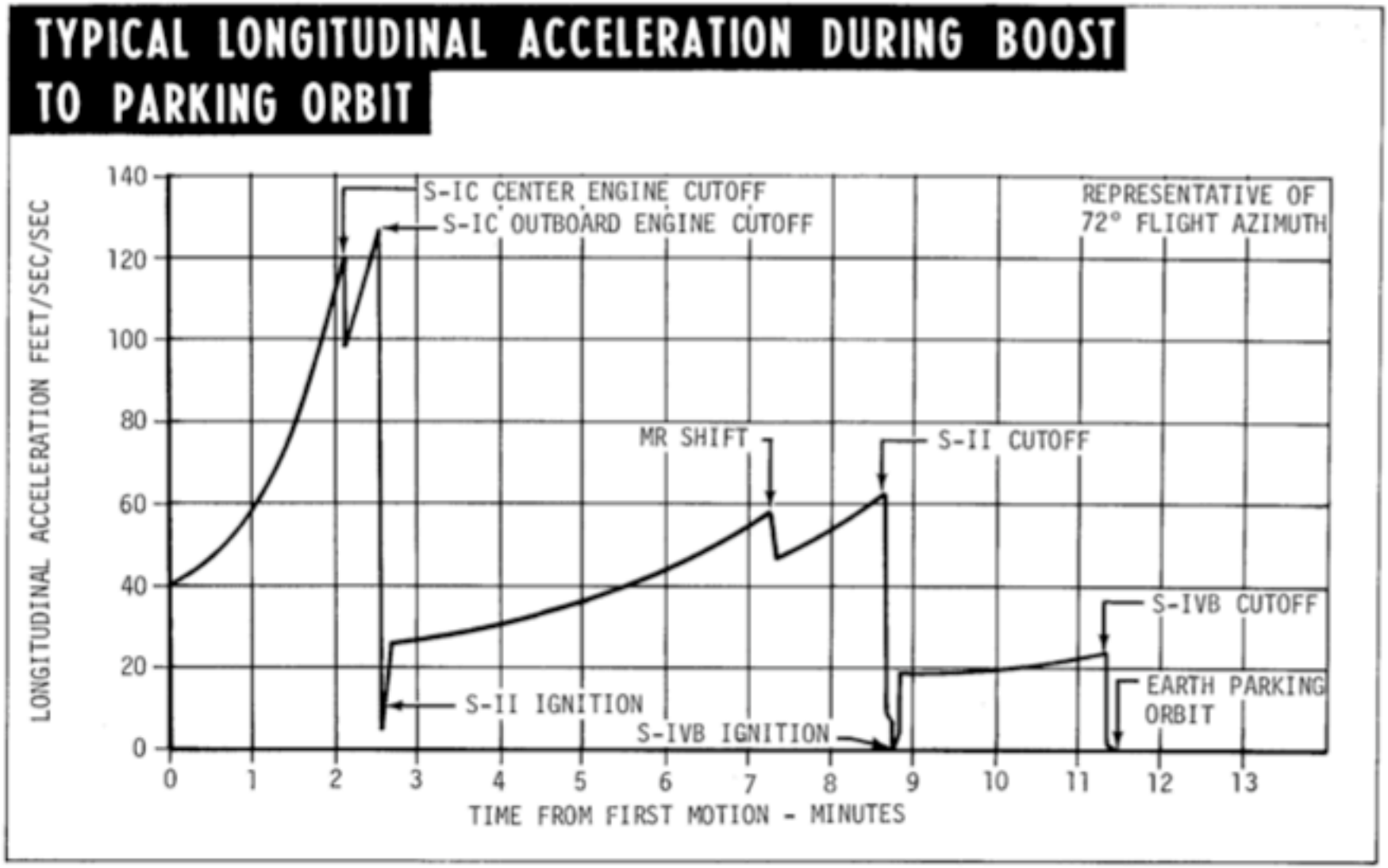
Aerodynamic Pressure During Launch



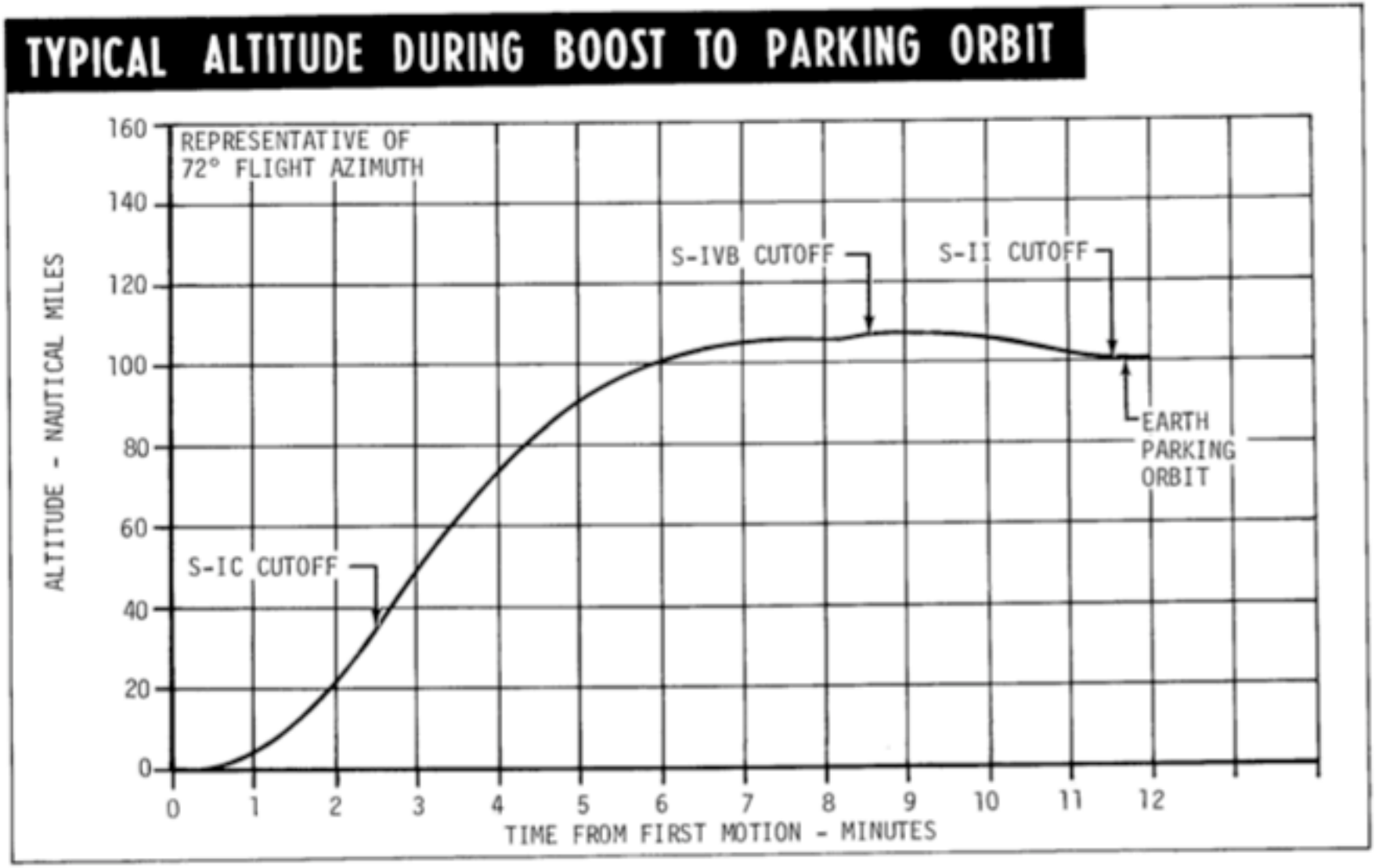
Velocity as a Function of Time



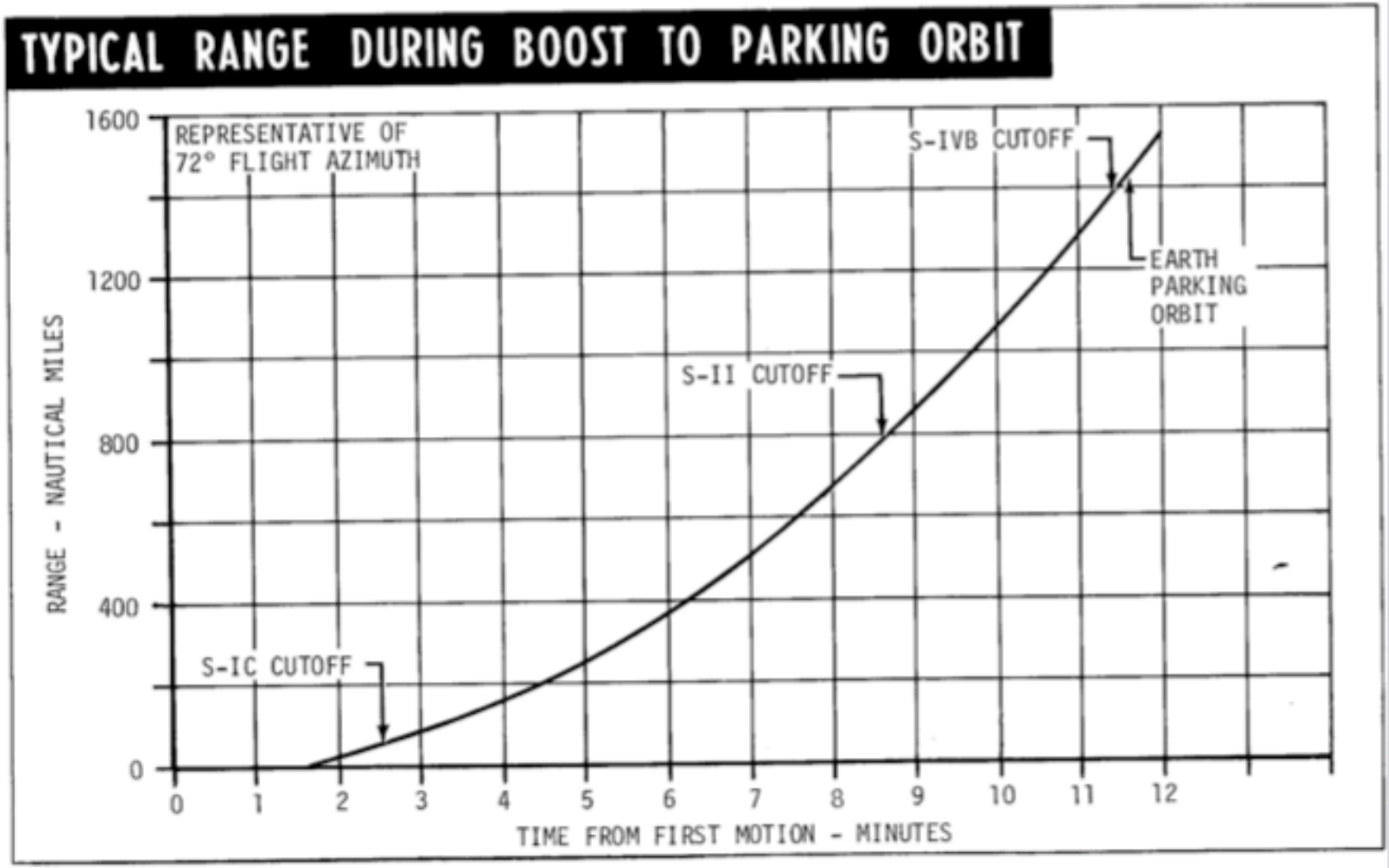
Acceleration as a Function of Time



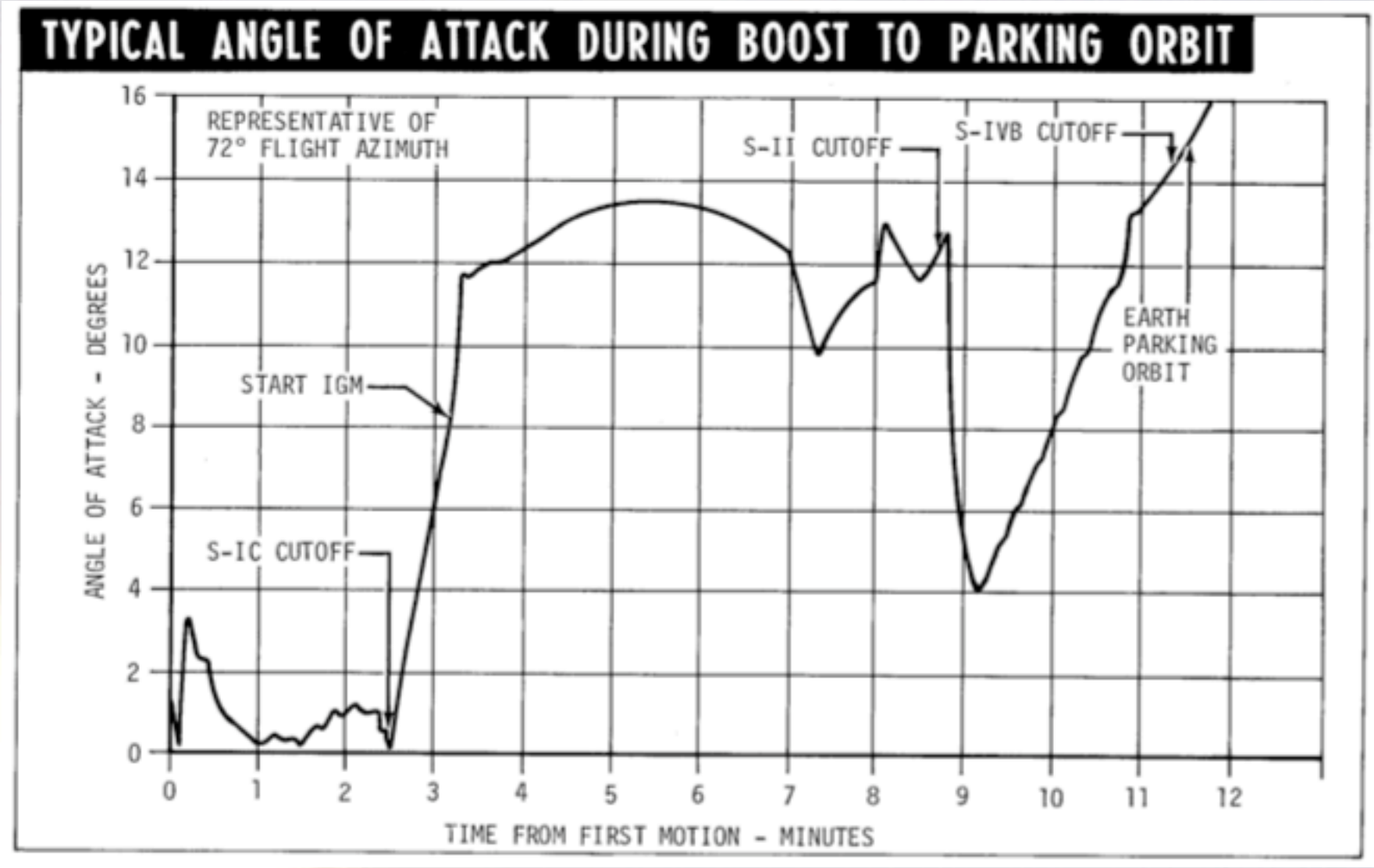
Altitude vs. Time



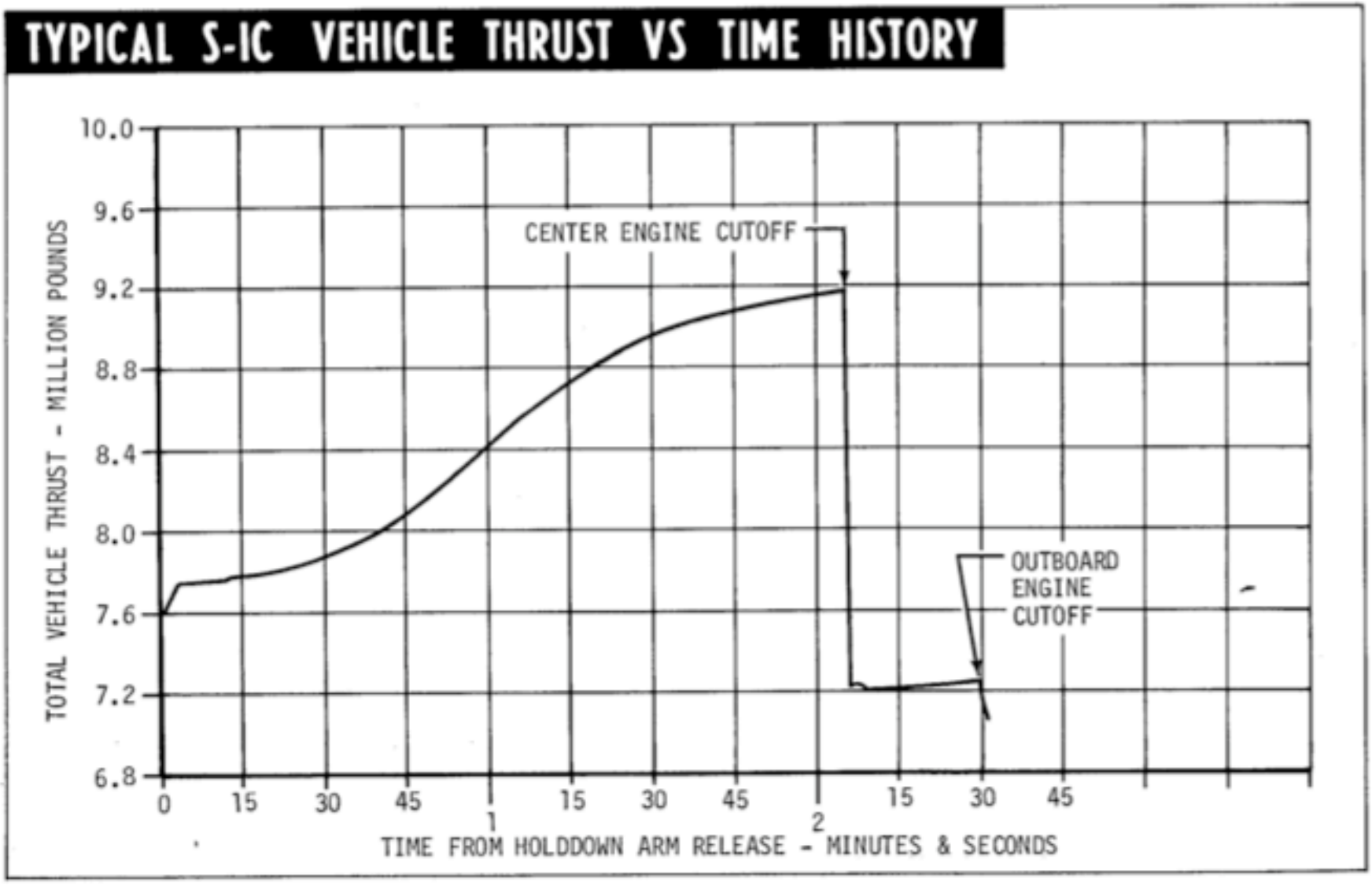
Down-Range Distance vs. Time



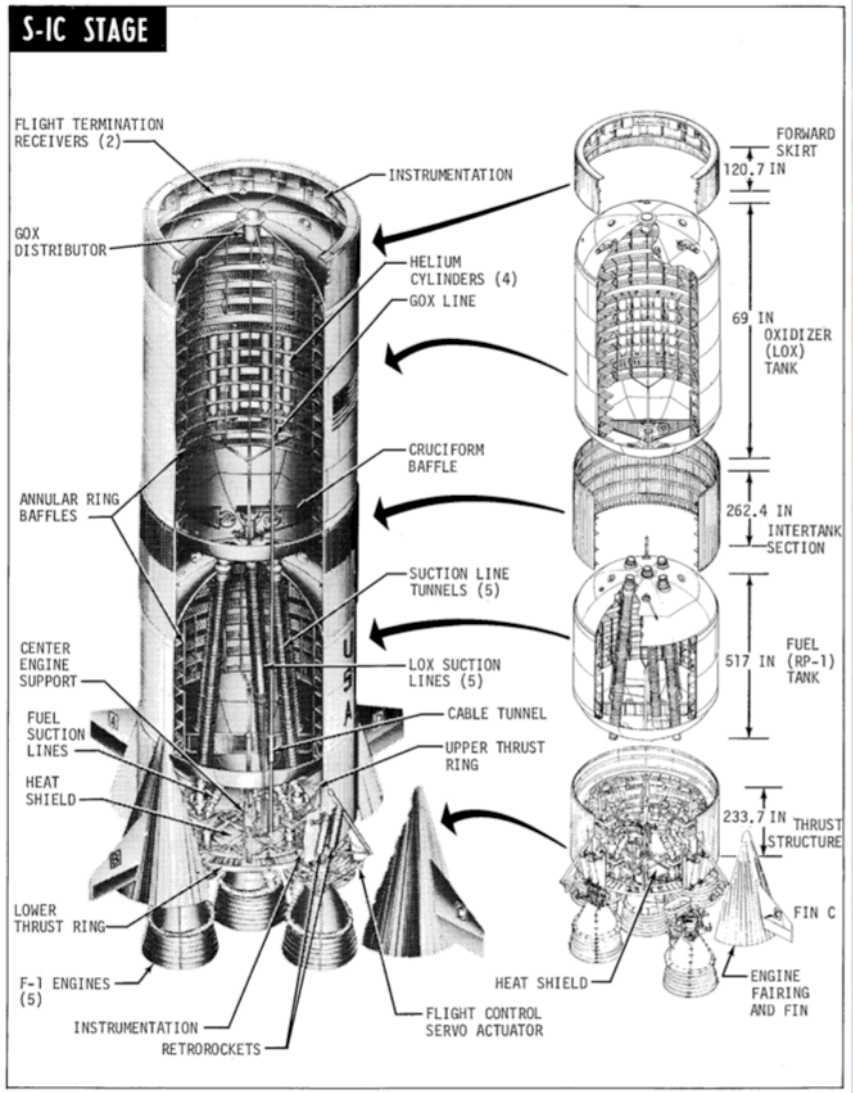
Angle of Attack in Trajectory



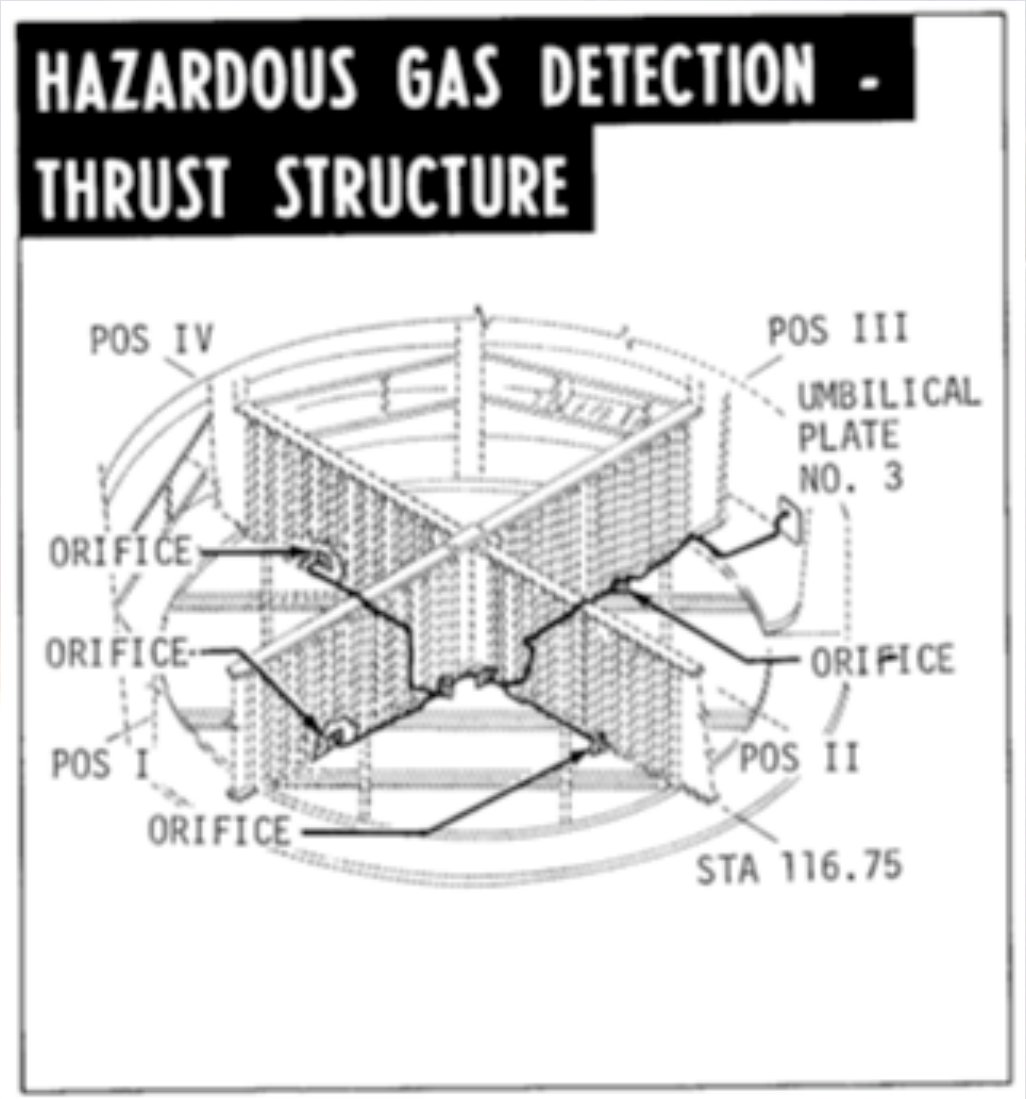
First Stage Thrust vs. Time



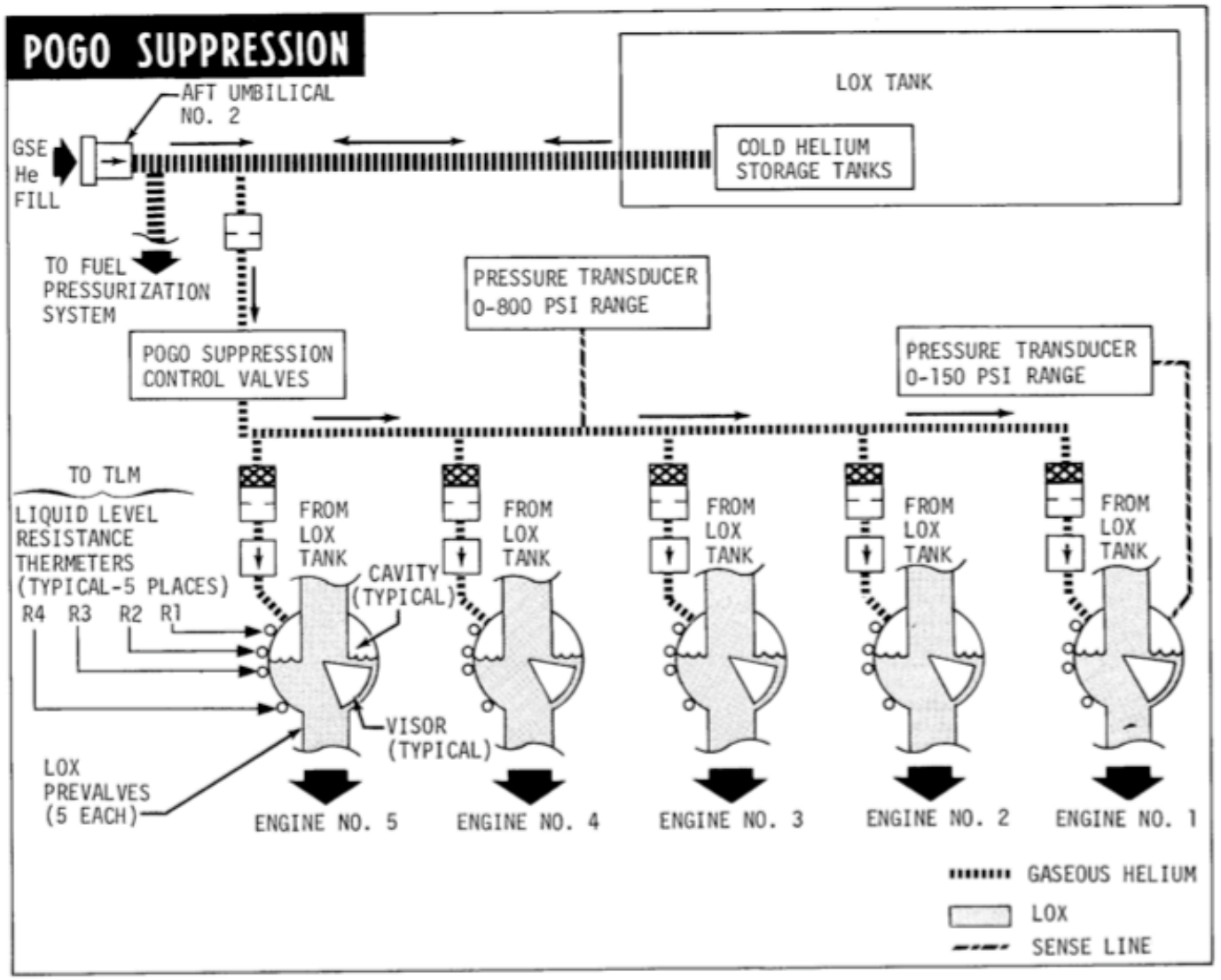
S-IC First Stage Internal Configuration



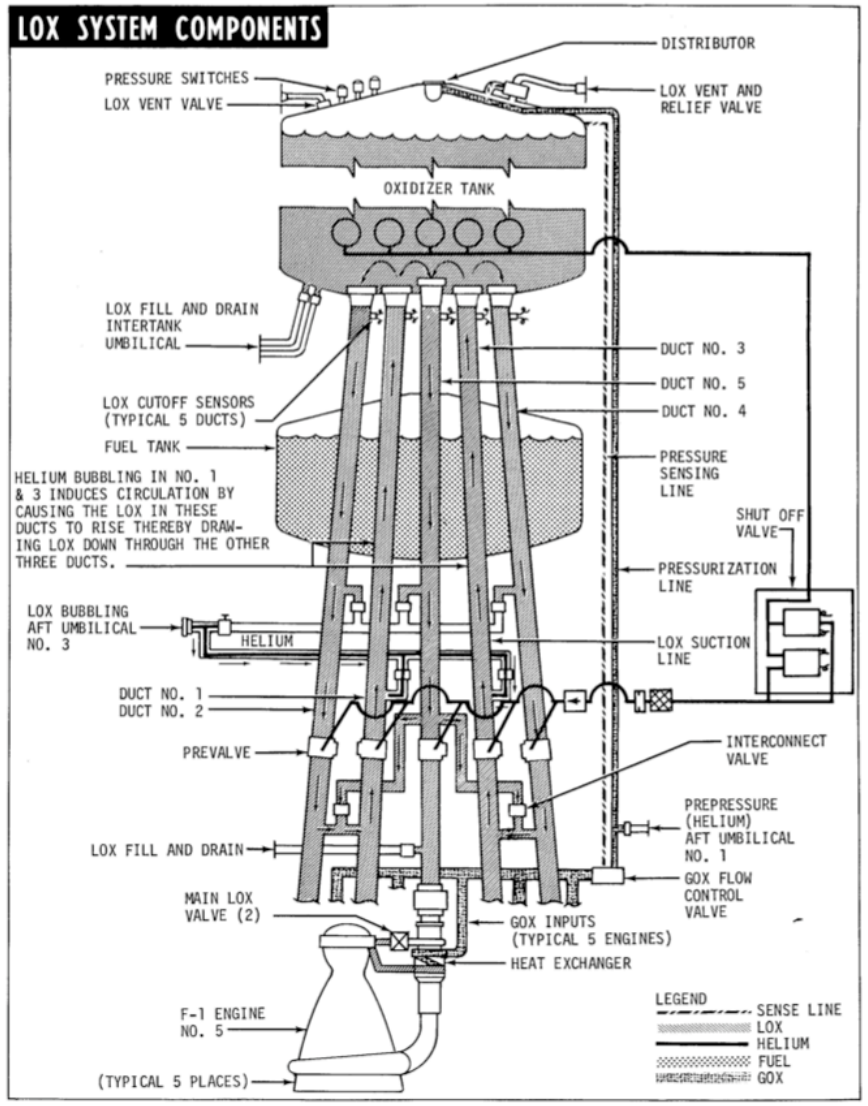
Hydrogen Leak Sensors



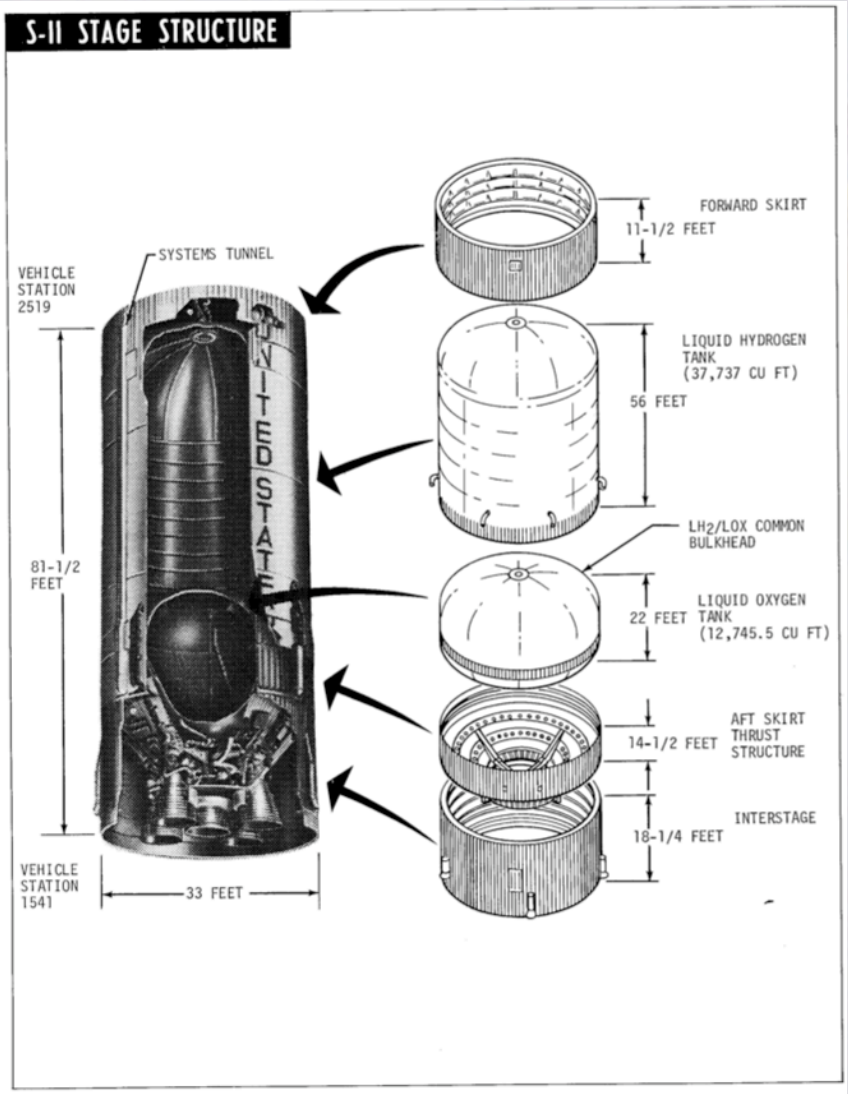
Pogo Suppression System



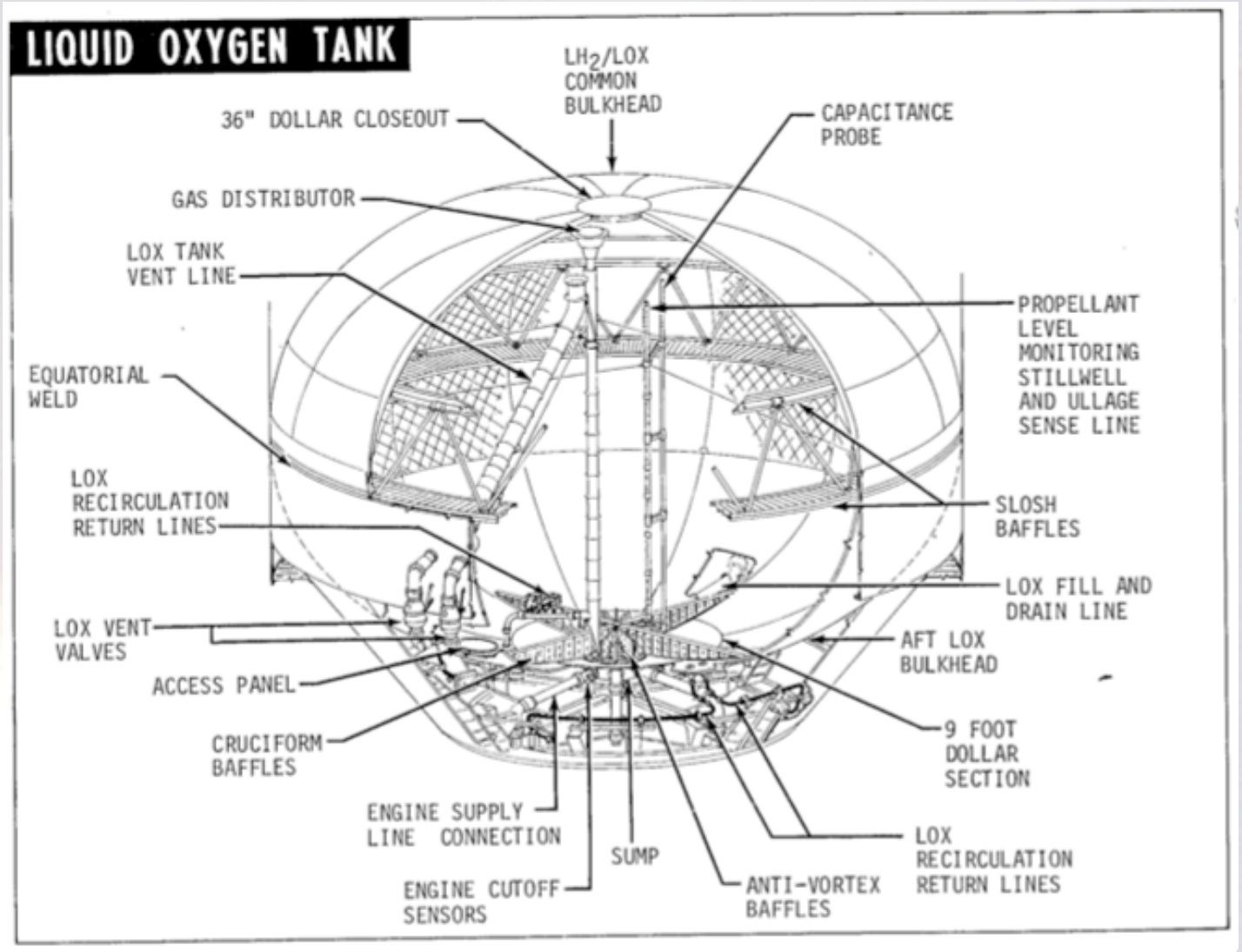
S-IC LOX Feed System



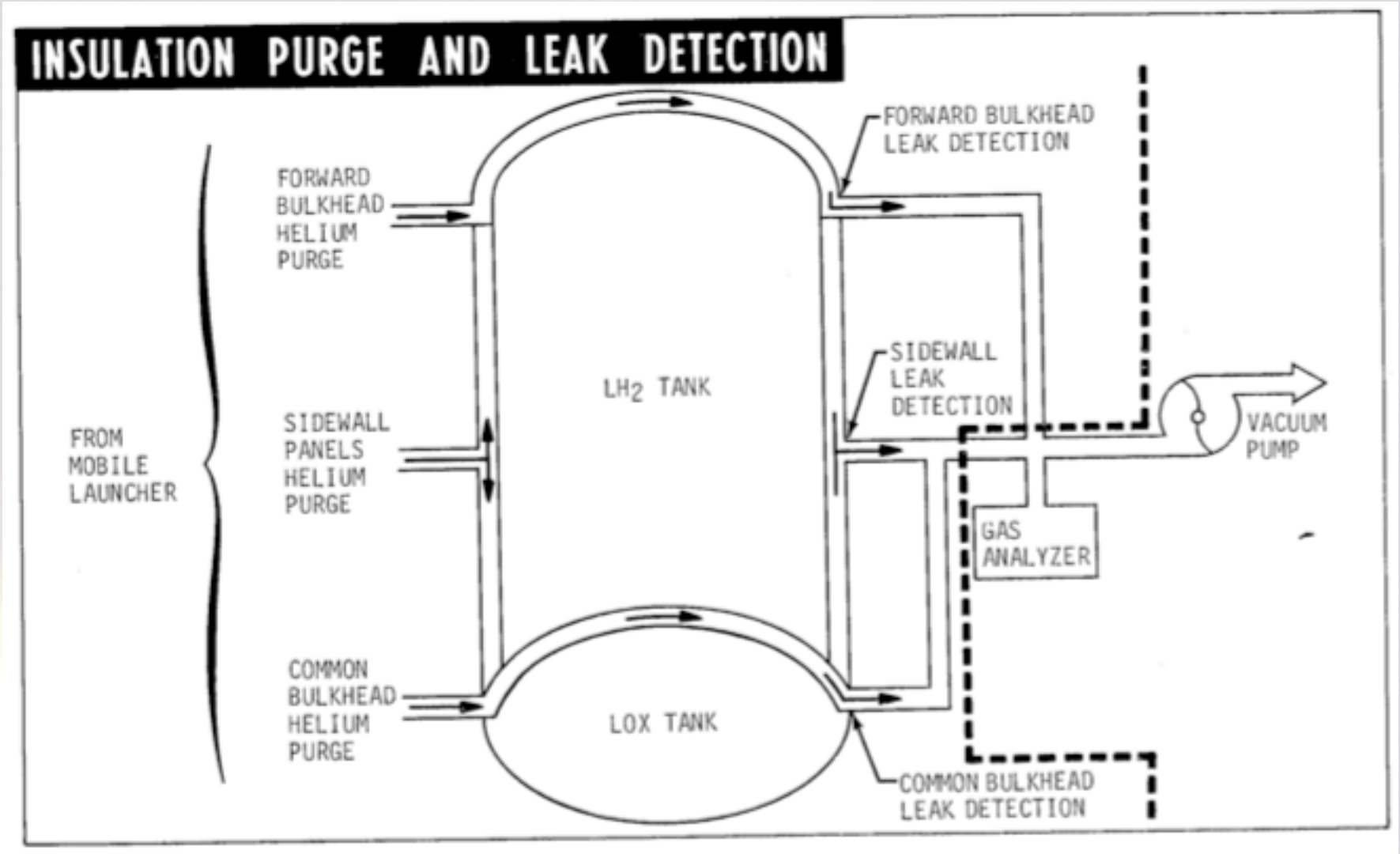
S-II Stage Structure



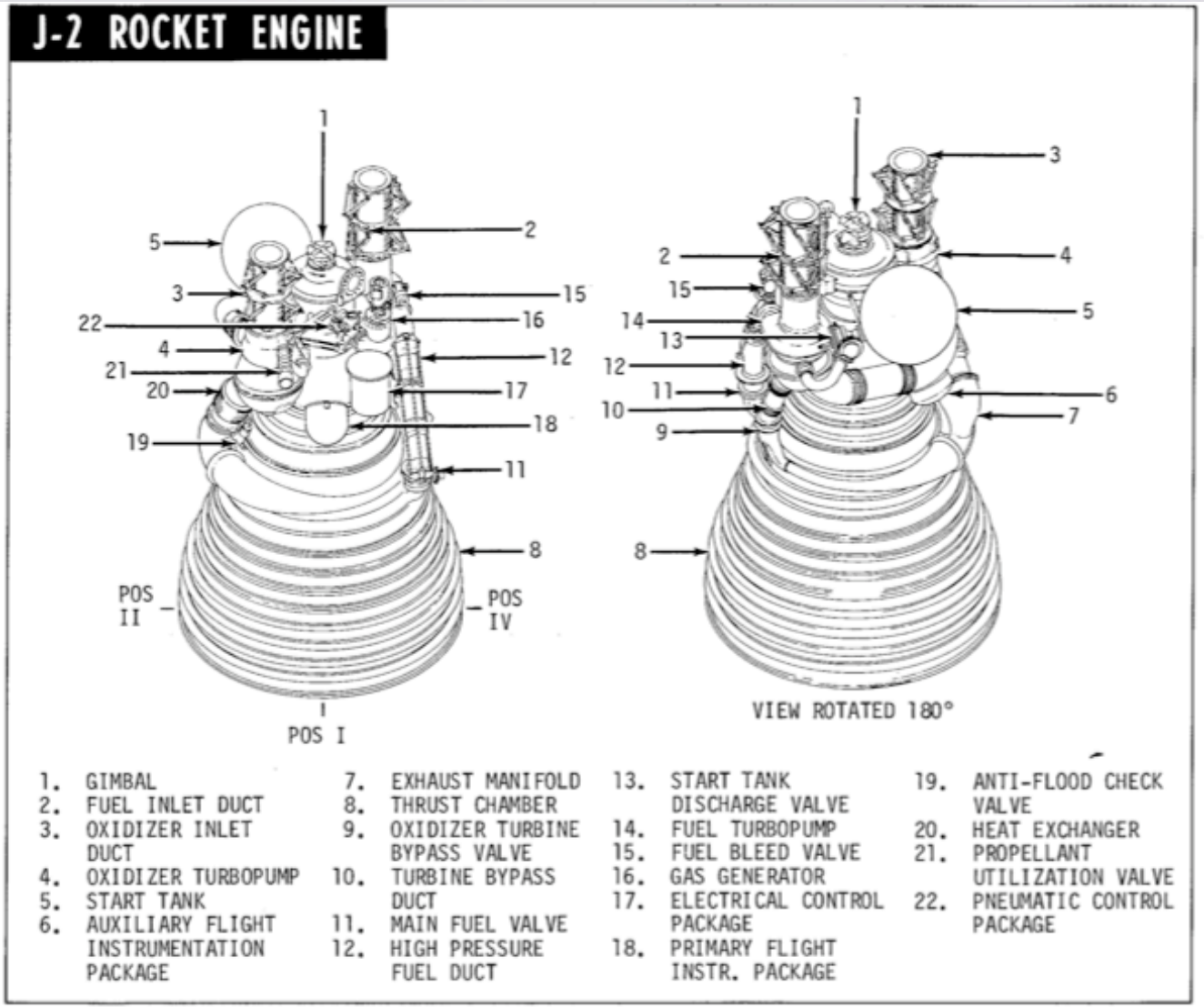
S-II LOX Tank Configuration



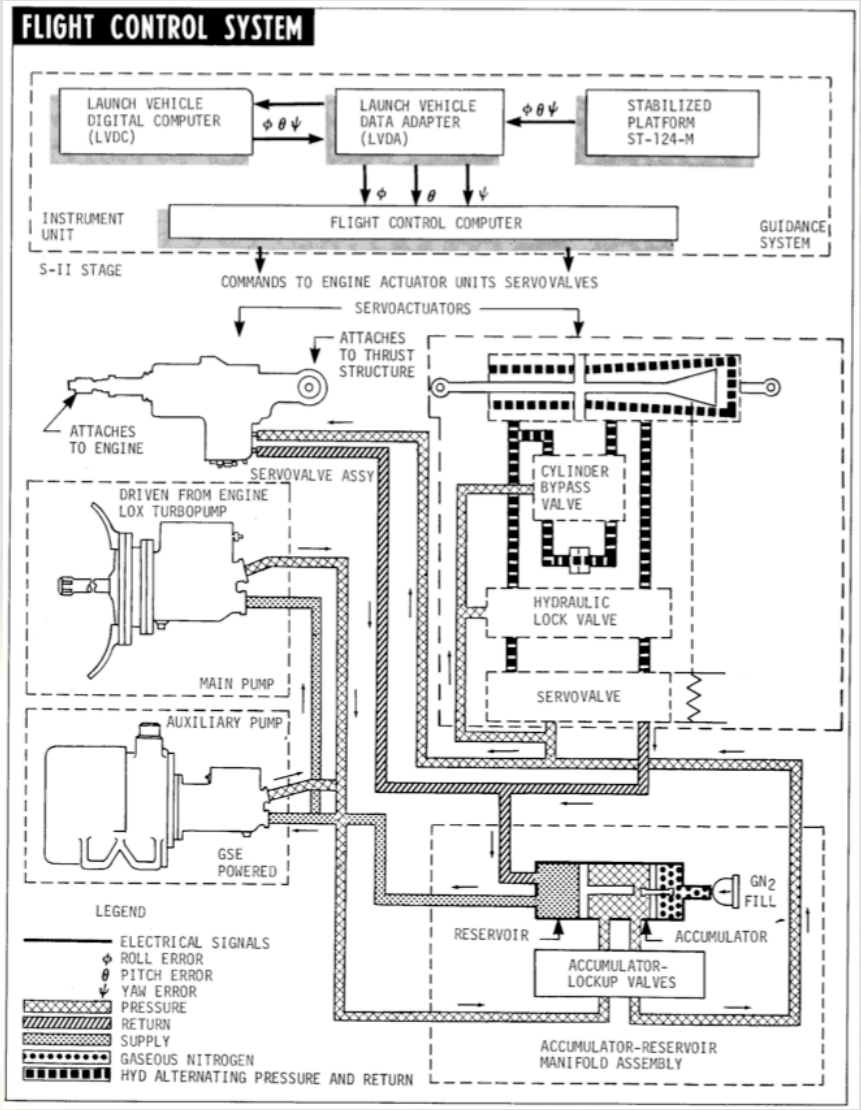
S-II Tank Purge and Leak Detection



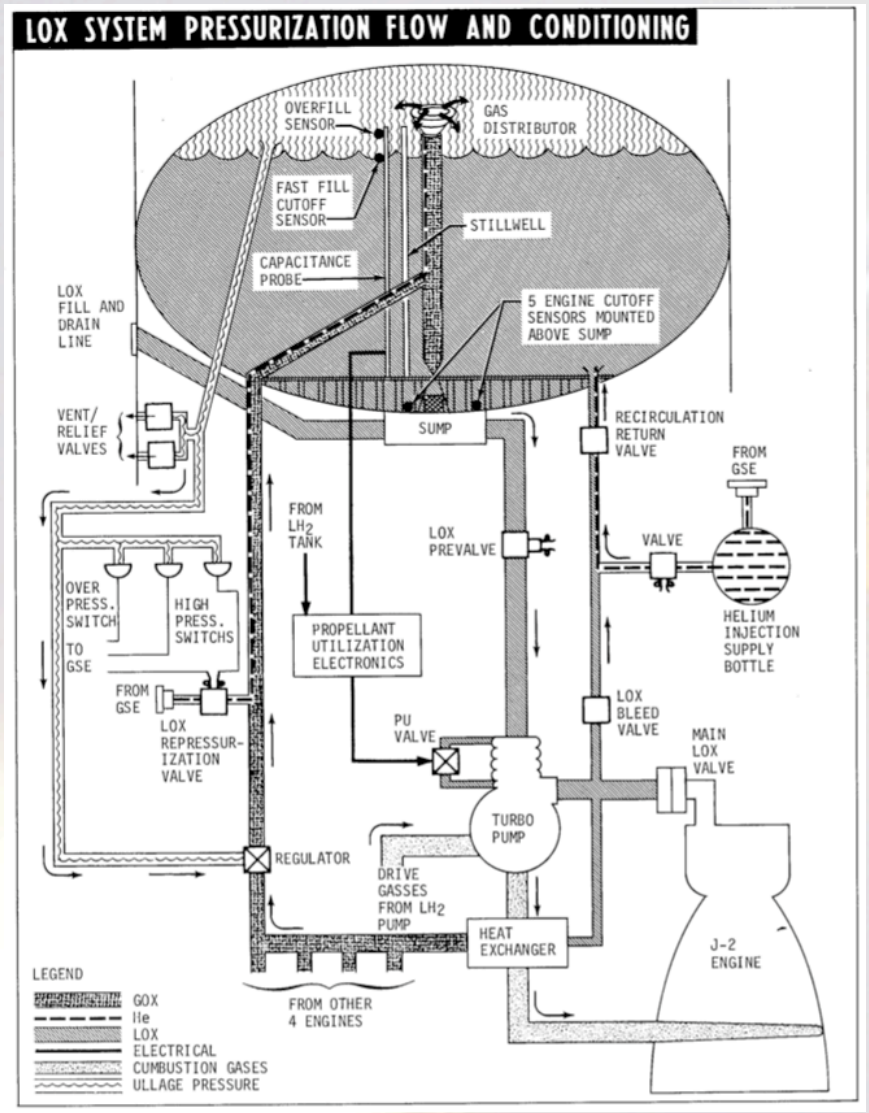
J-2 Rocket Engine (S-II Stage)

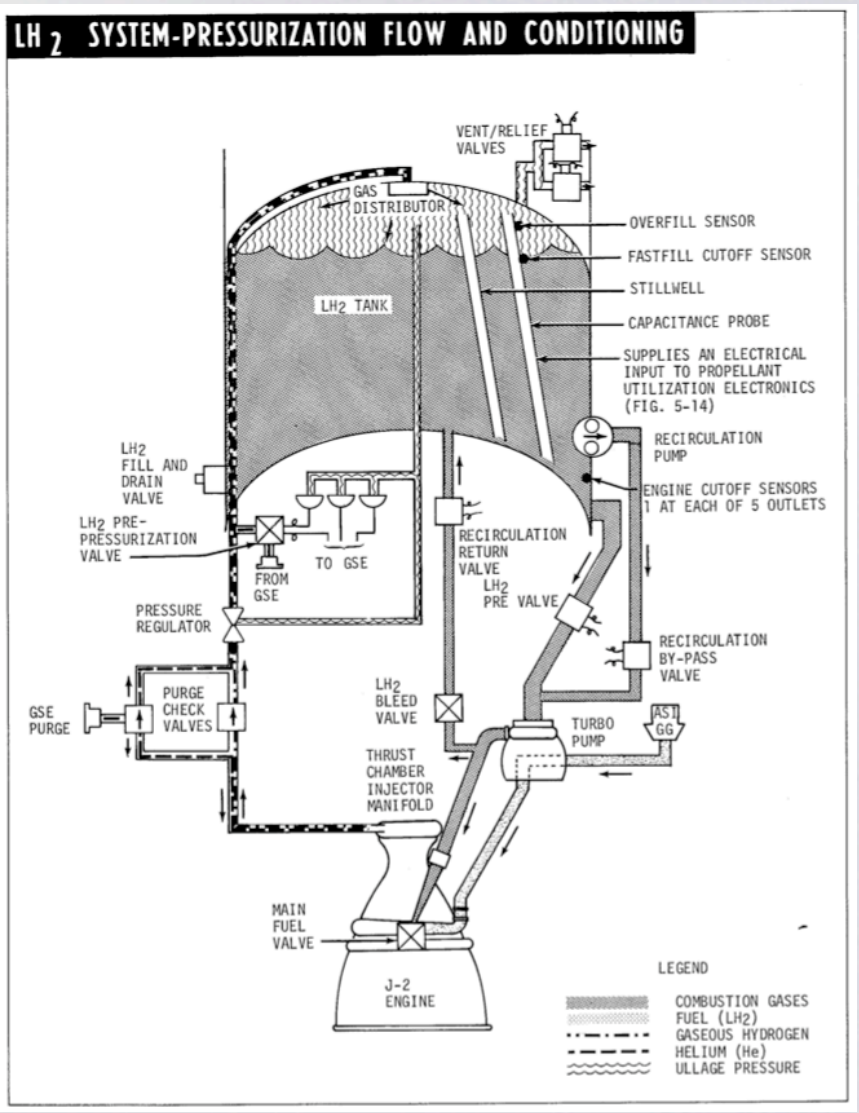


S-II Flight Control System Block Diagram

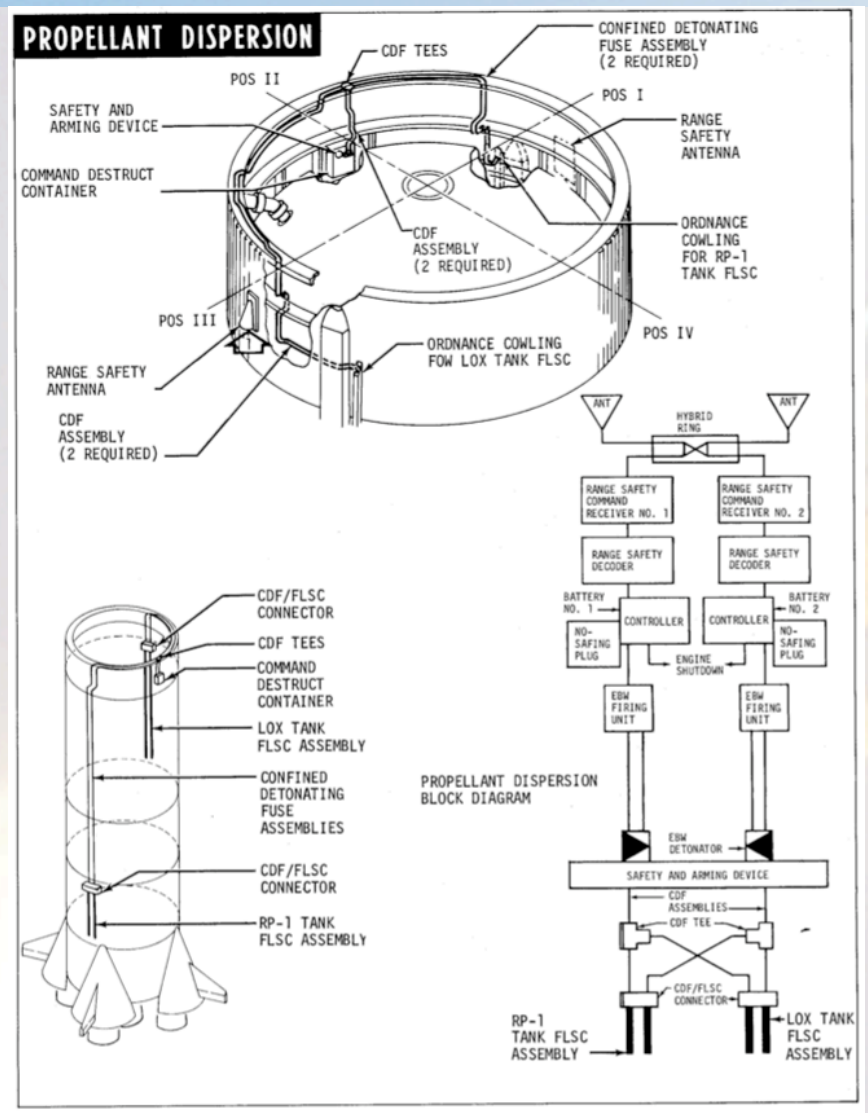


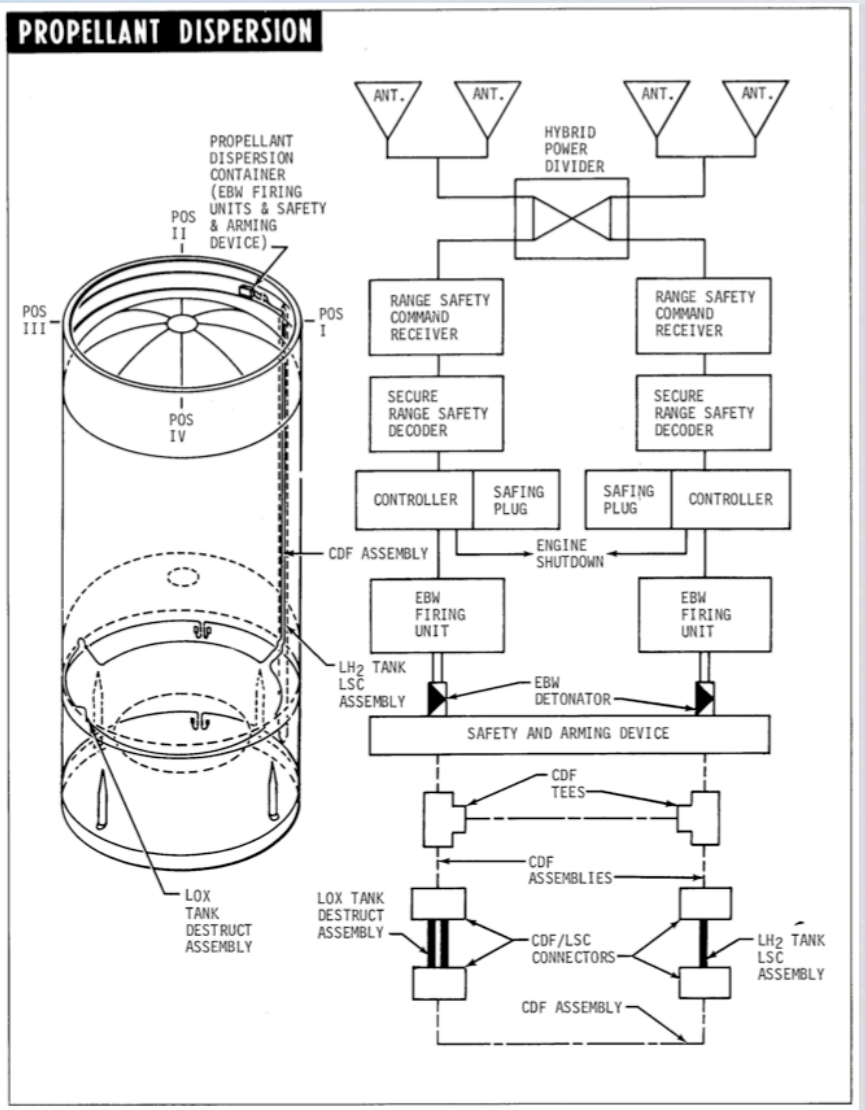
S-II Propellant Pressurization Systems



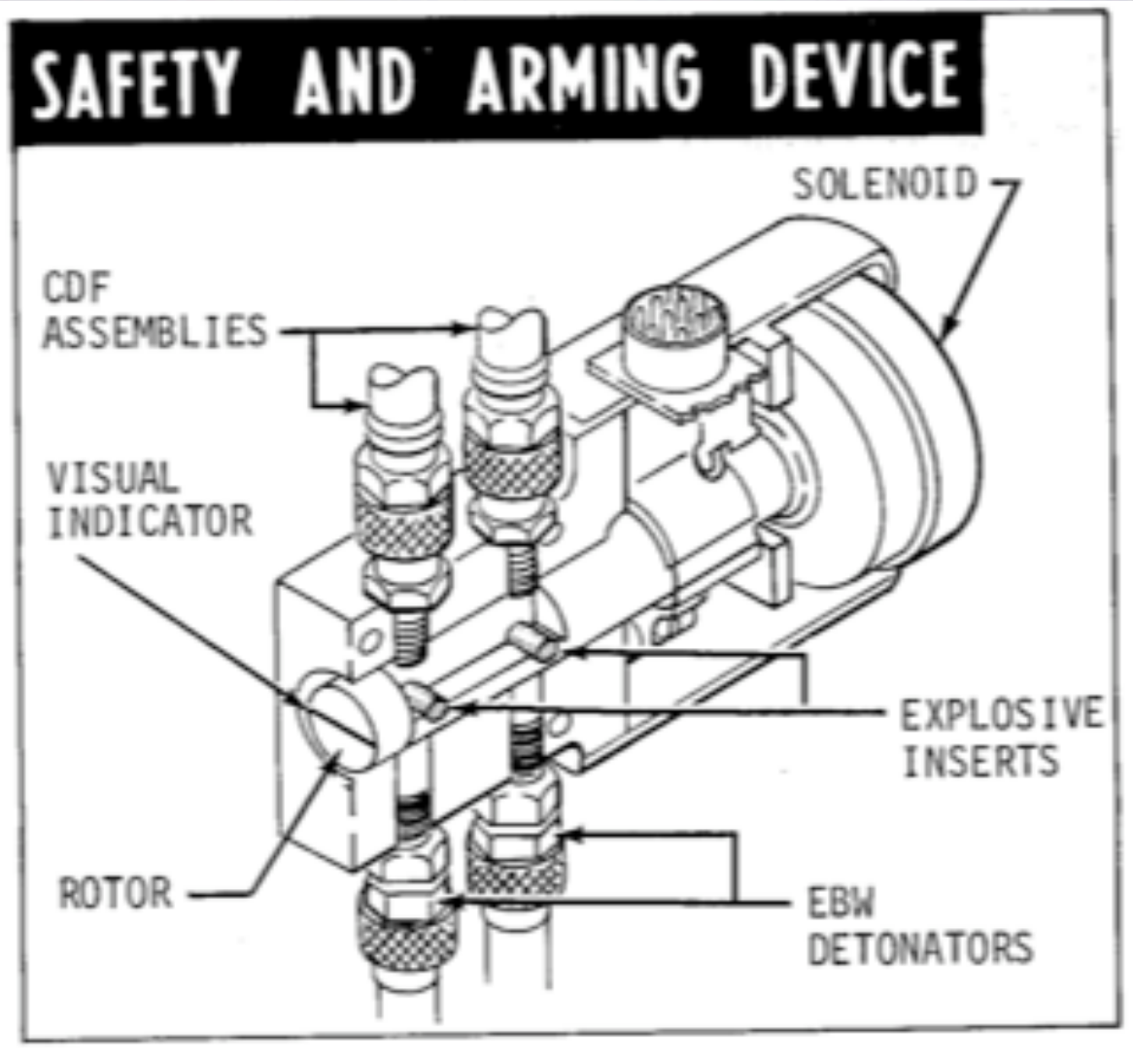


Range Safety System (S-IC/S-II)

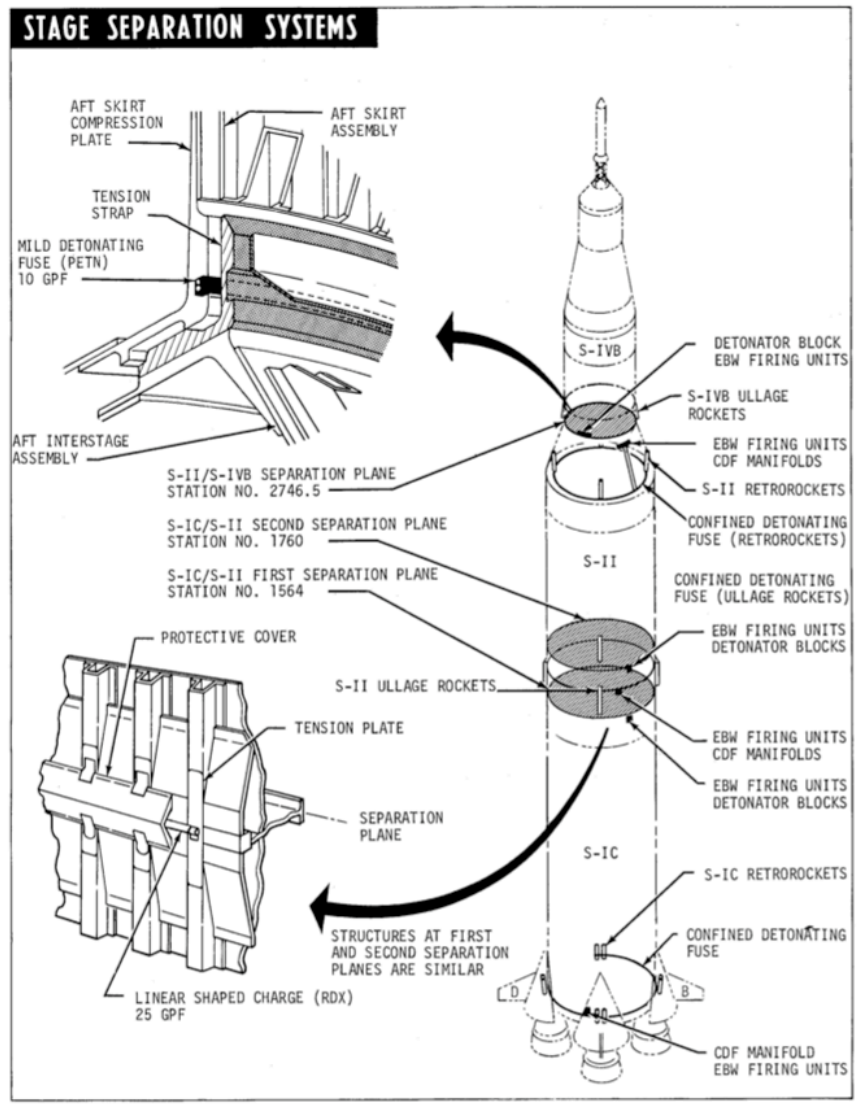


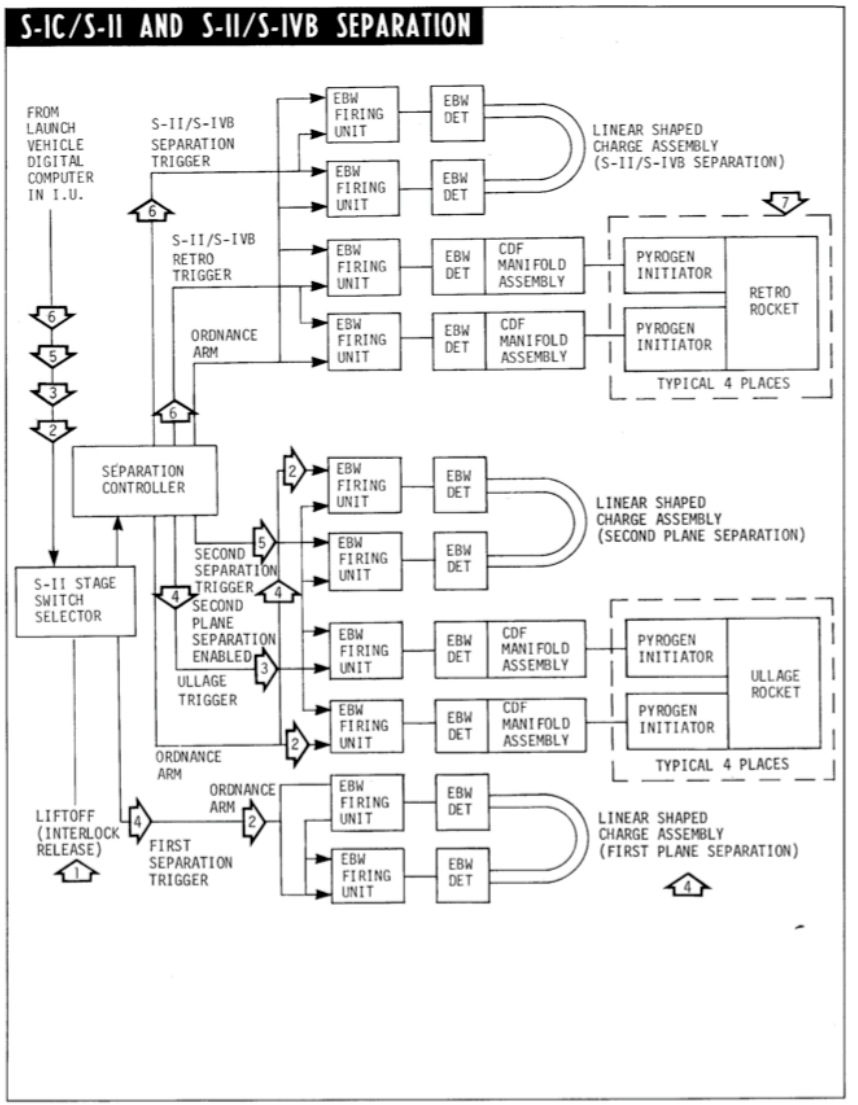


Safe and Arm Switch

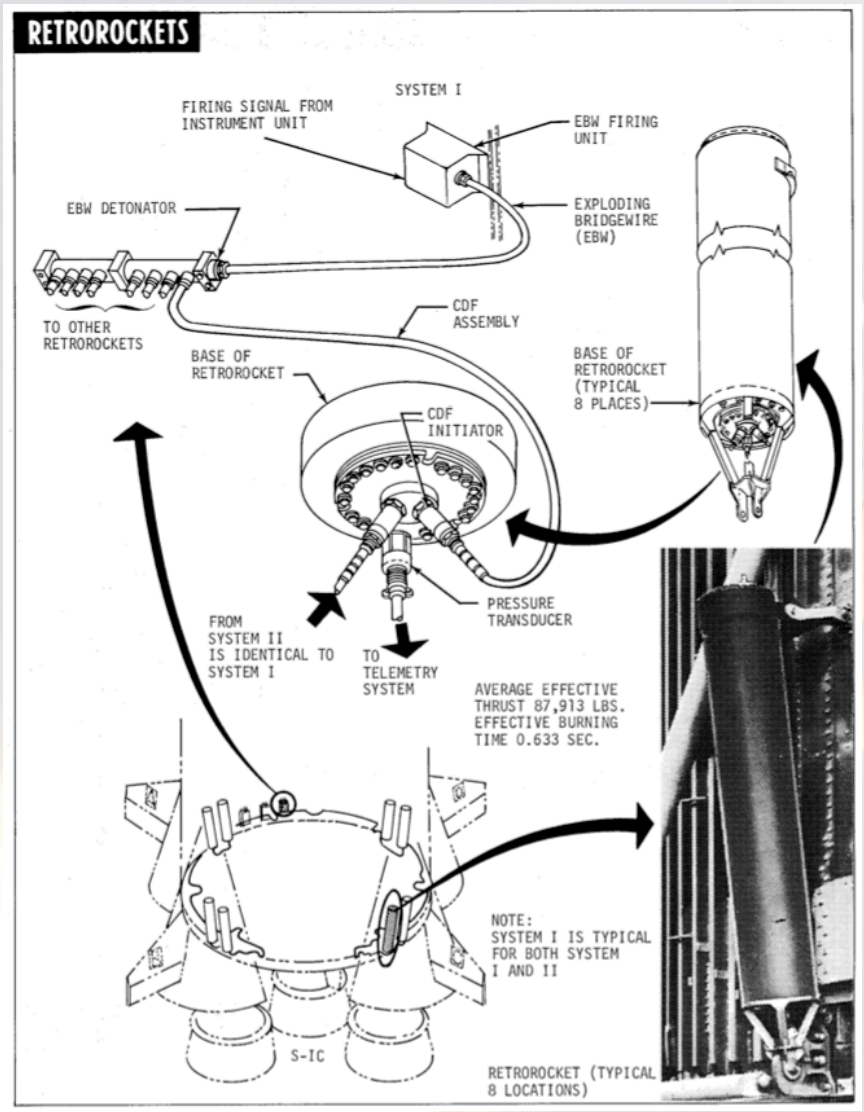


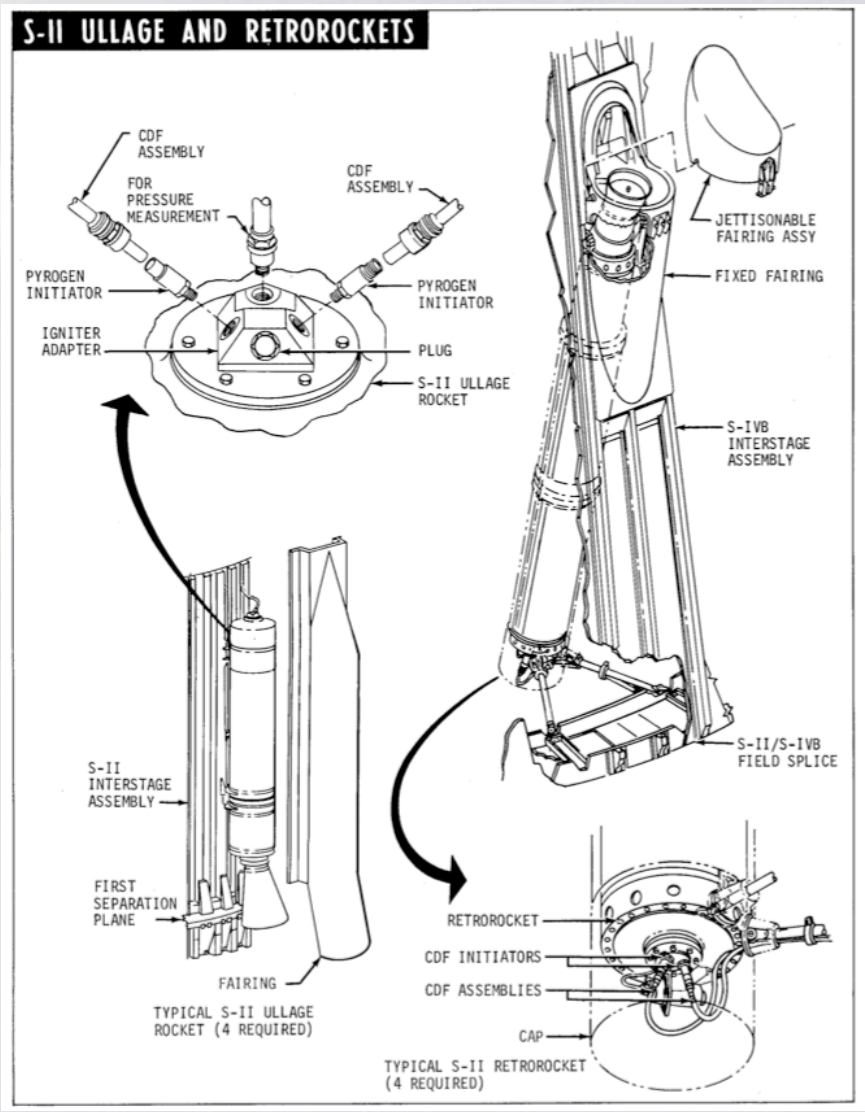
Stage Separation



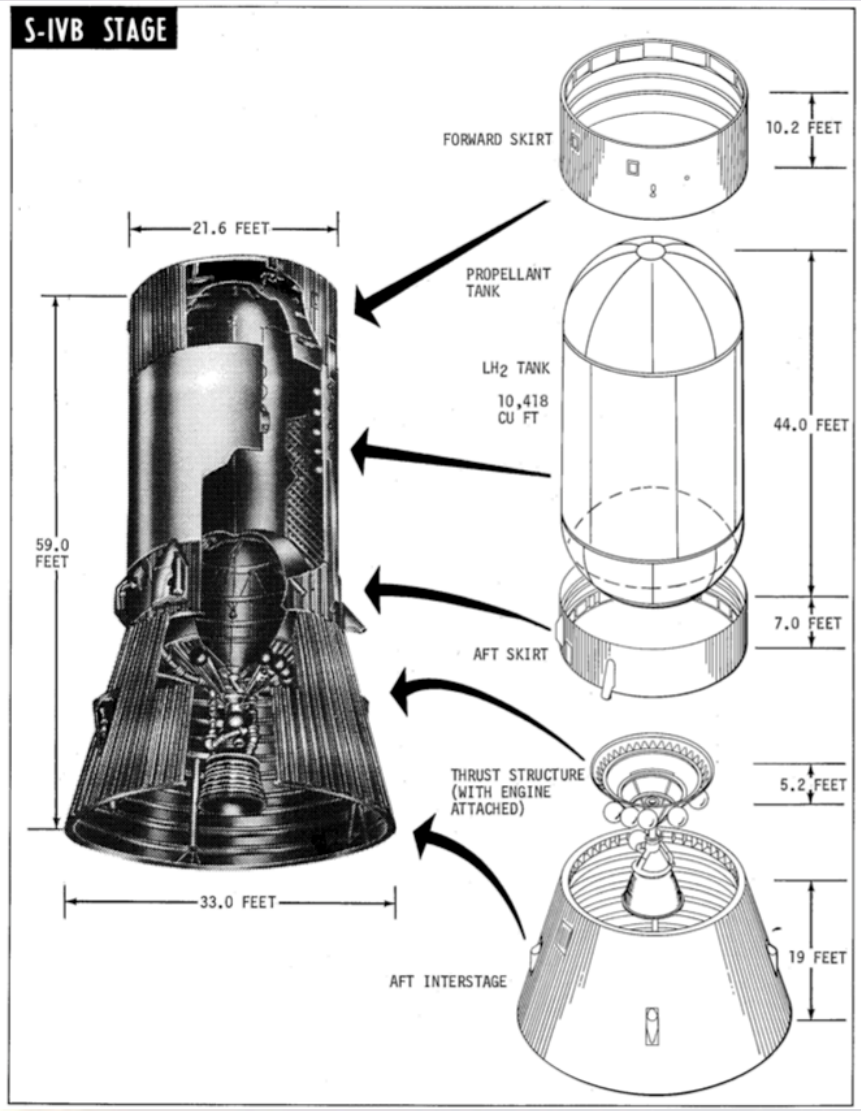


Retro and Ullage Rockets (Stage Separation)

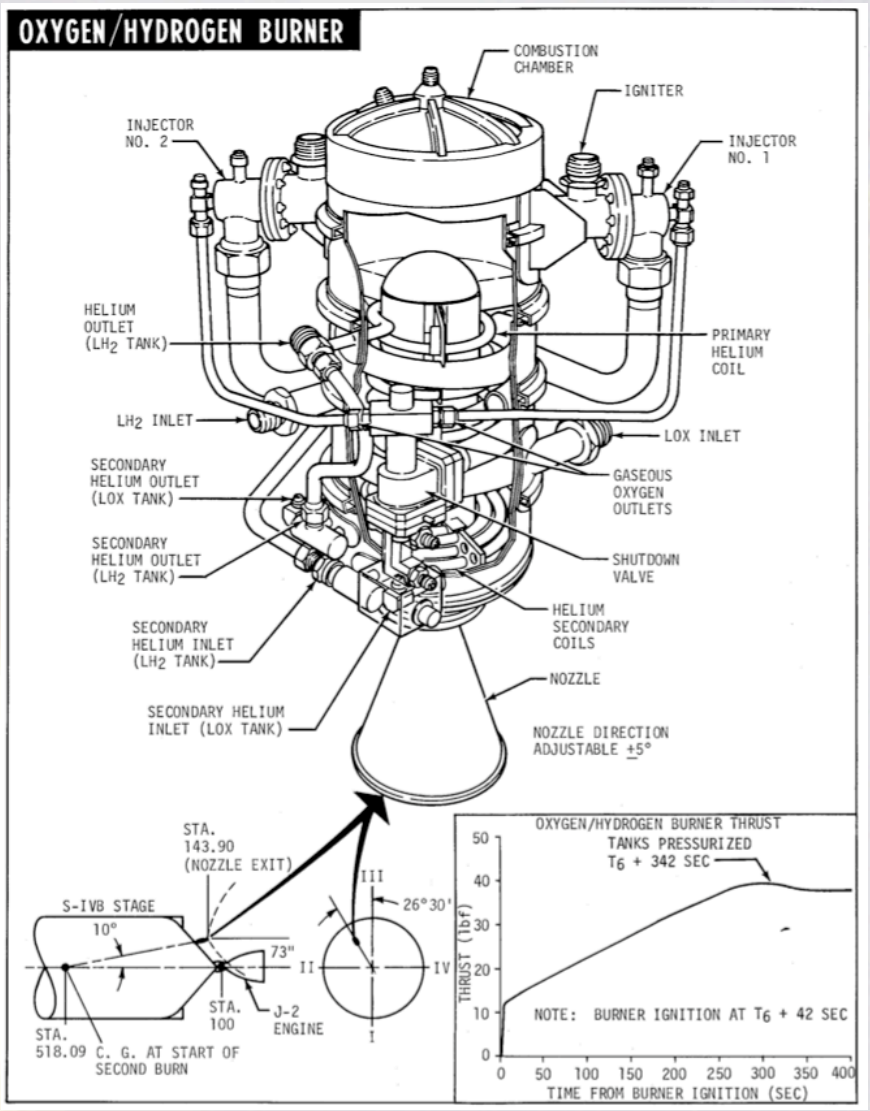




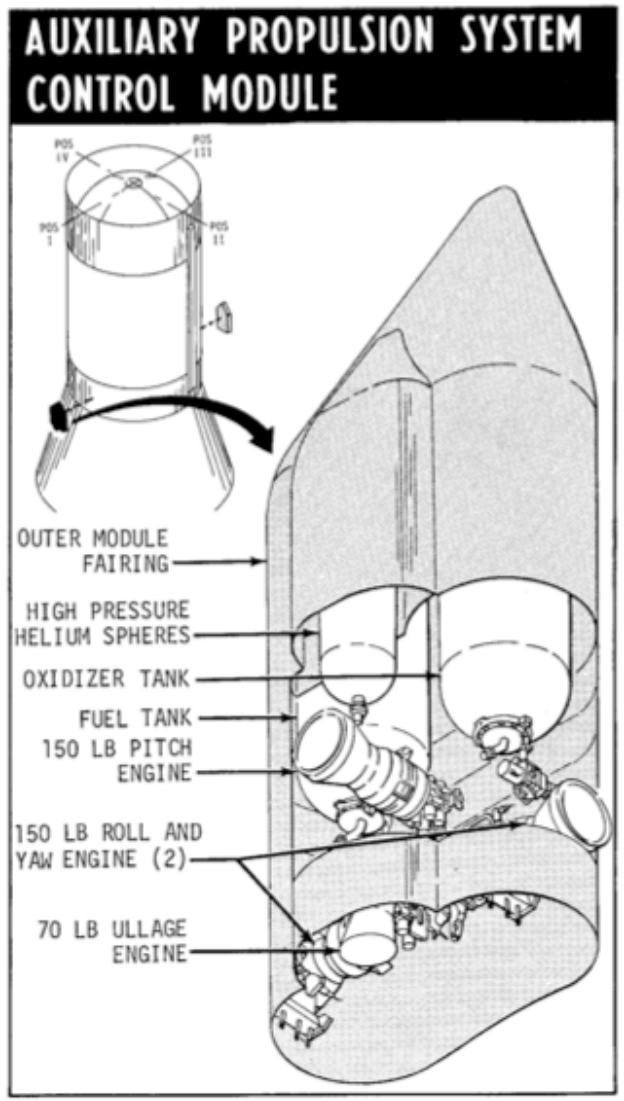
S-IVB Configuration

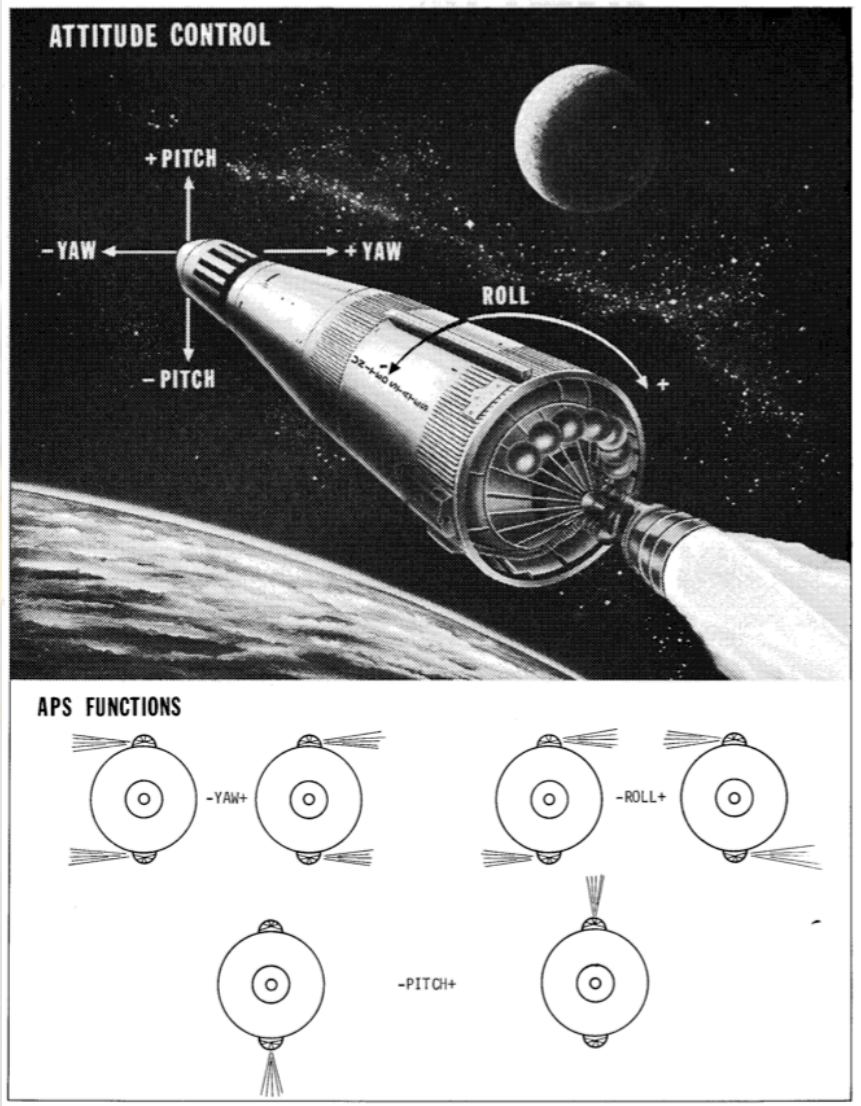


O2/H2 Burner (GHe Heater)

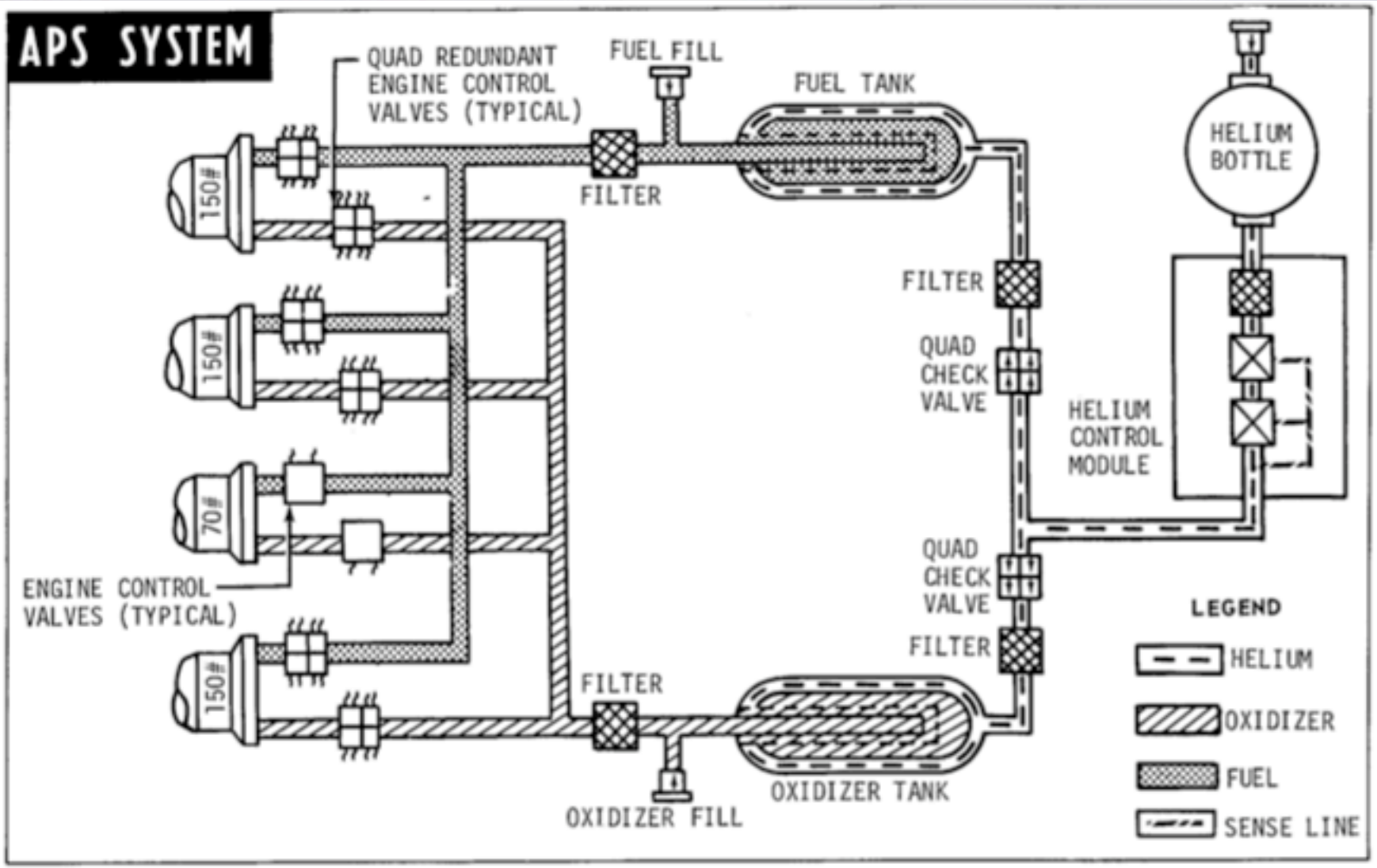


S-IVB Auxiliary Propulsion System



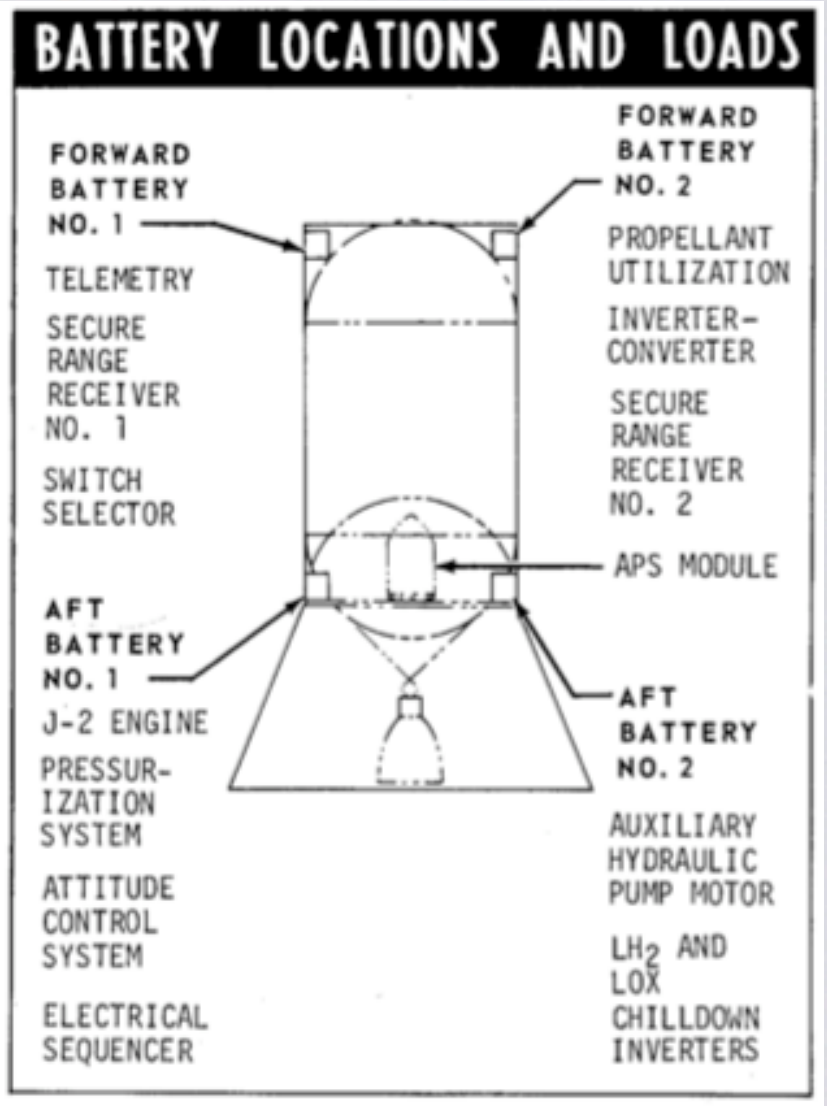


APS Plumbing and Control

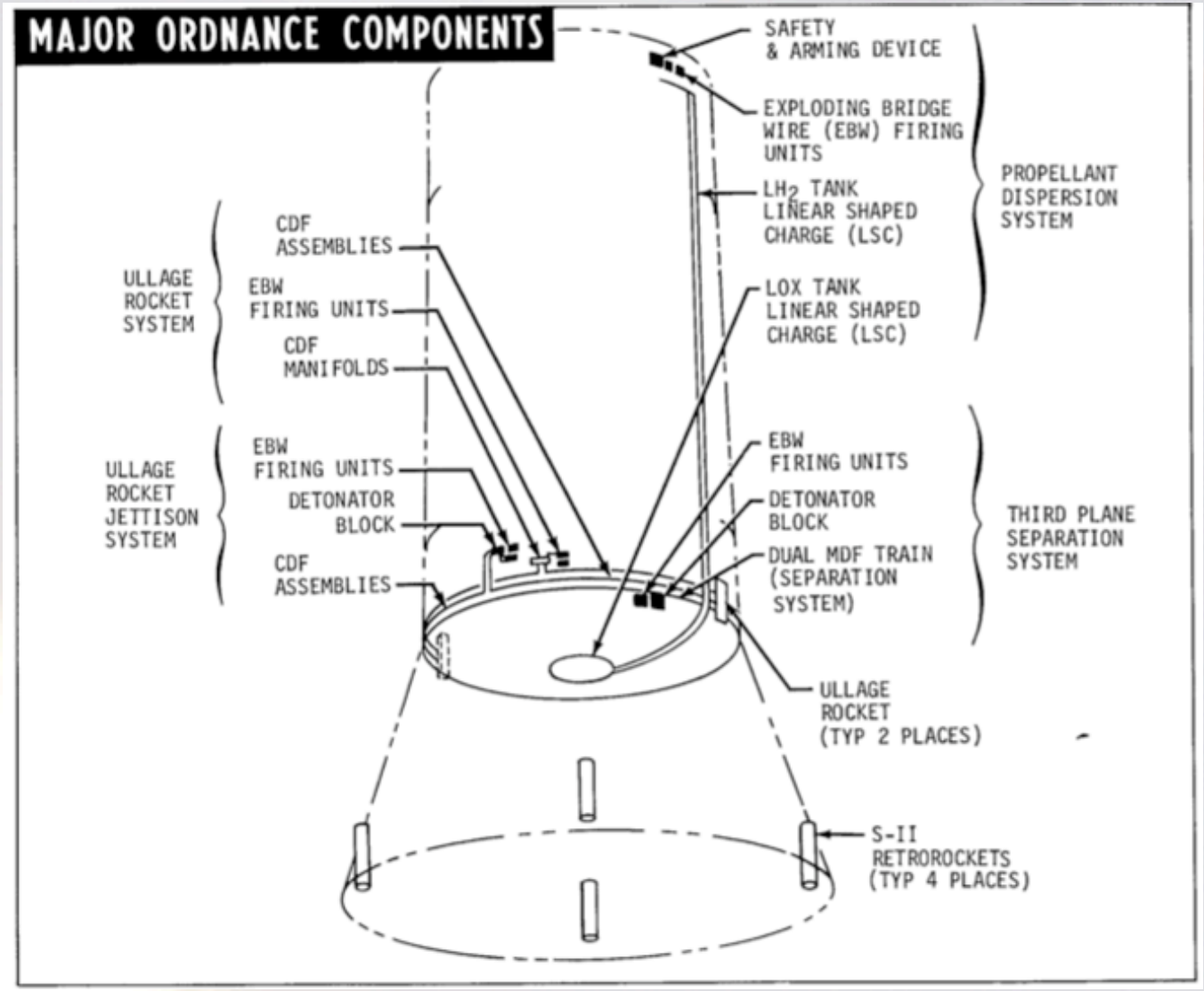


S-IVB Battery System

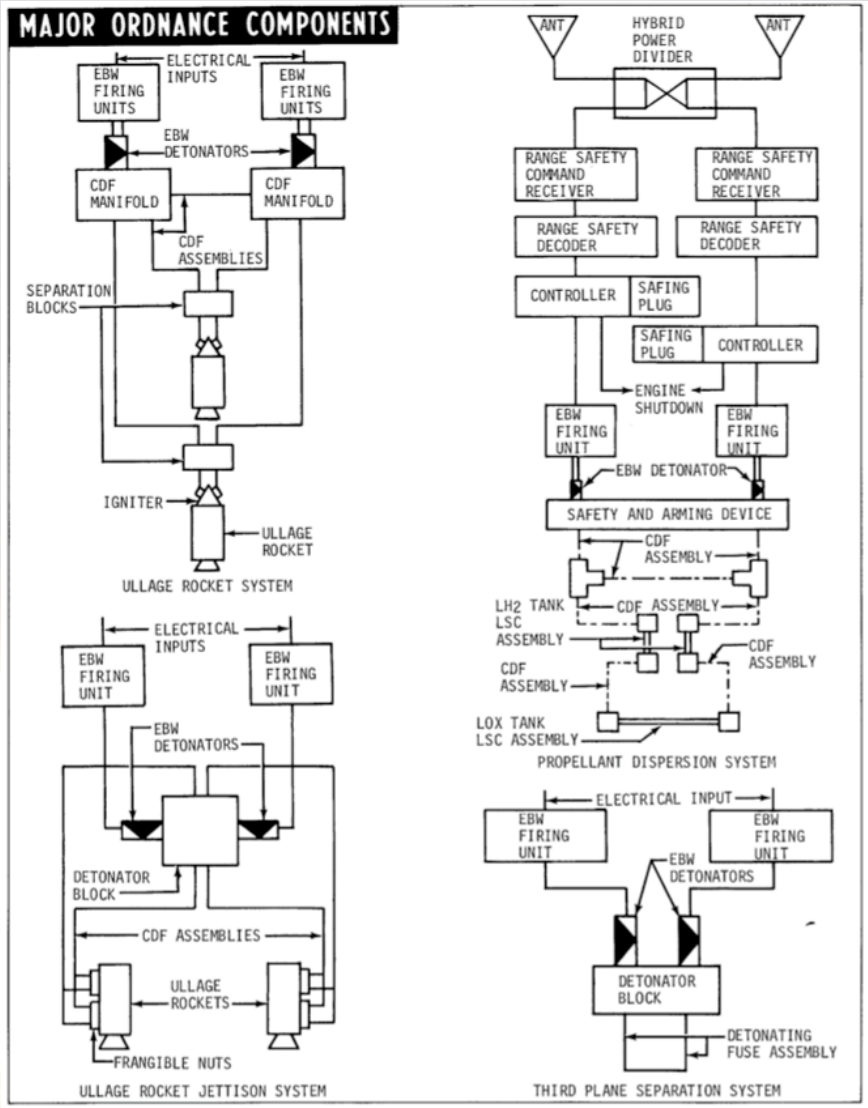
S-IVB BATTERY CHARACTERISTICS				
TYPE	Dry charge			
MATERIAL	Zinc/silver-oxide			
ELECTROLYTE	Potassium hydroxide (KOH) in pure water			
CELLS	20, with taps for selecting 18 or 19 to reduce output voltage as required			
NOMINAL VOLTAGE OUTPUT	1.5 vdc per cell 28 (±2) vdc per 18 to 20 cell group Aft Battery No. 2 is made up of two regular 28 (±2) vdc batteries and has an output of 56 (±4) vdc			
	FORWARD NO. 1	FORWARD NO. 2	AFT NO. 1	AFT NO. 2
CURRENT RATING	179 AH	12.2 AH	179 AH	49.6 AH
Gross Weight (Design target weight)	90 lbs	Two units: 20 lbs ea.	90 lbs	75 lbs



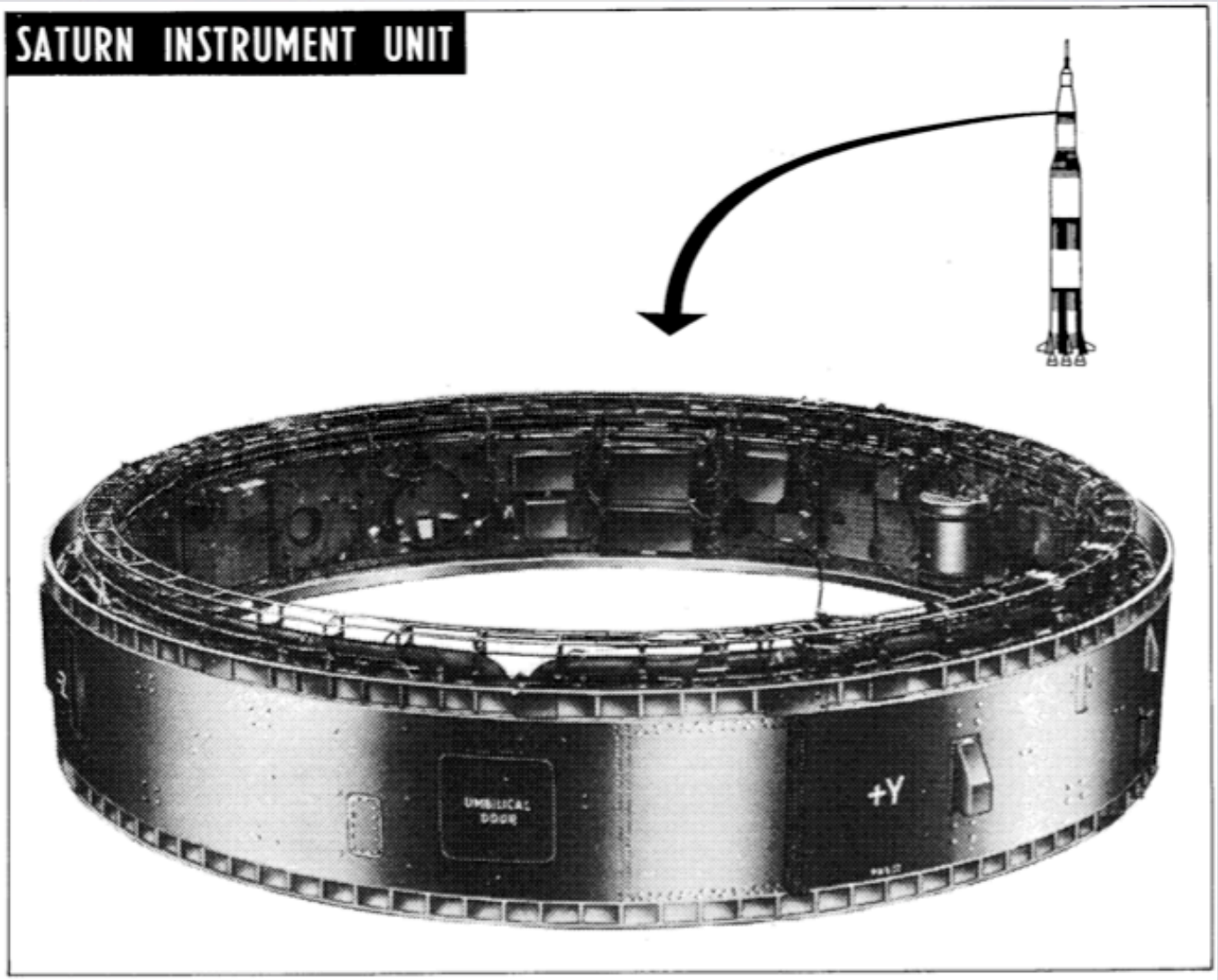
S-IVB Pyrotechnics



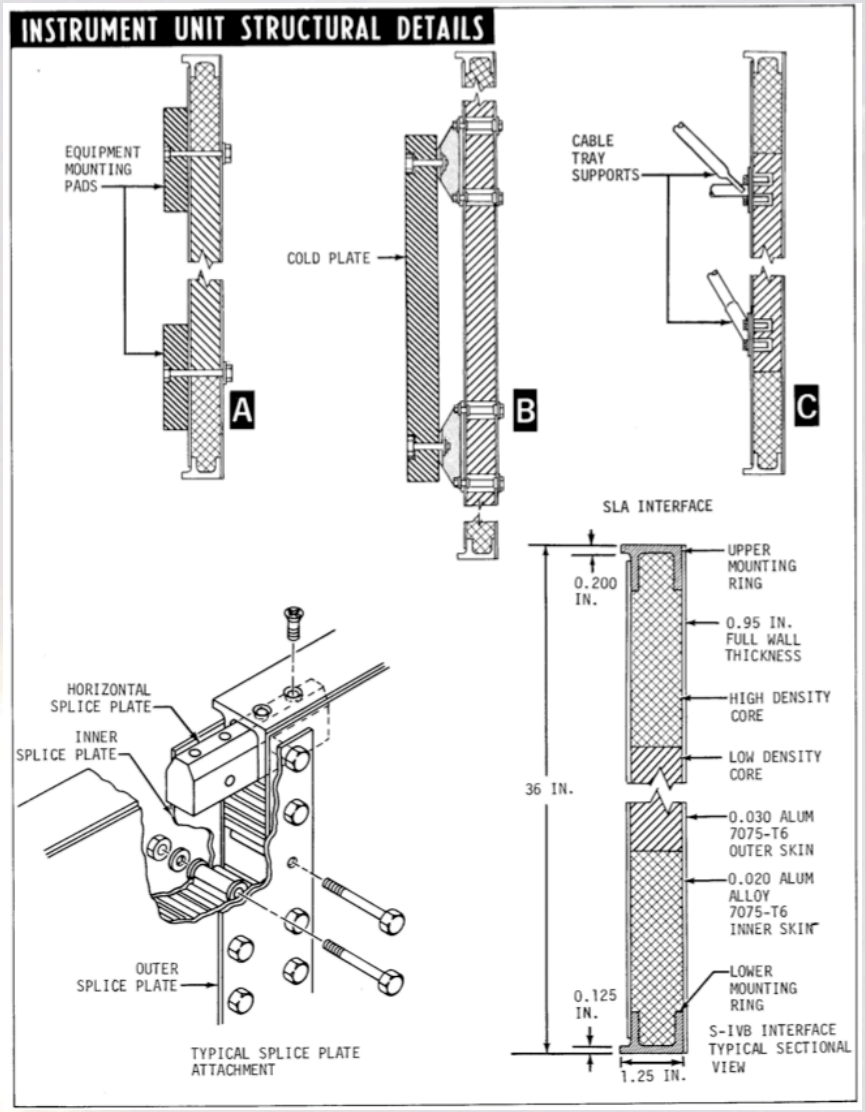
S-IVB Ordinance Control



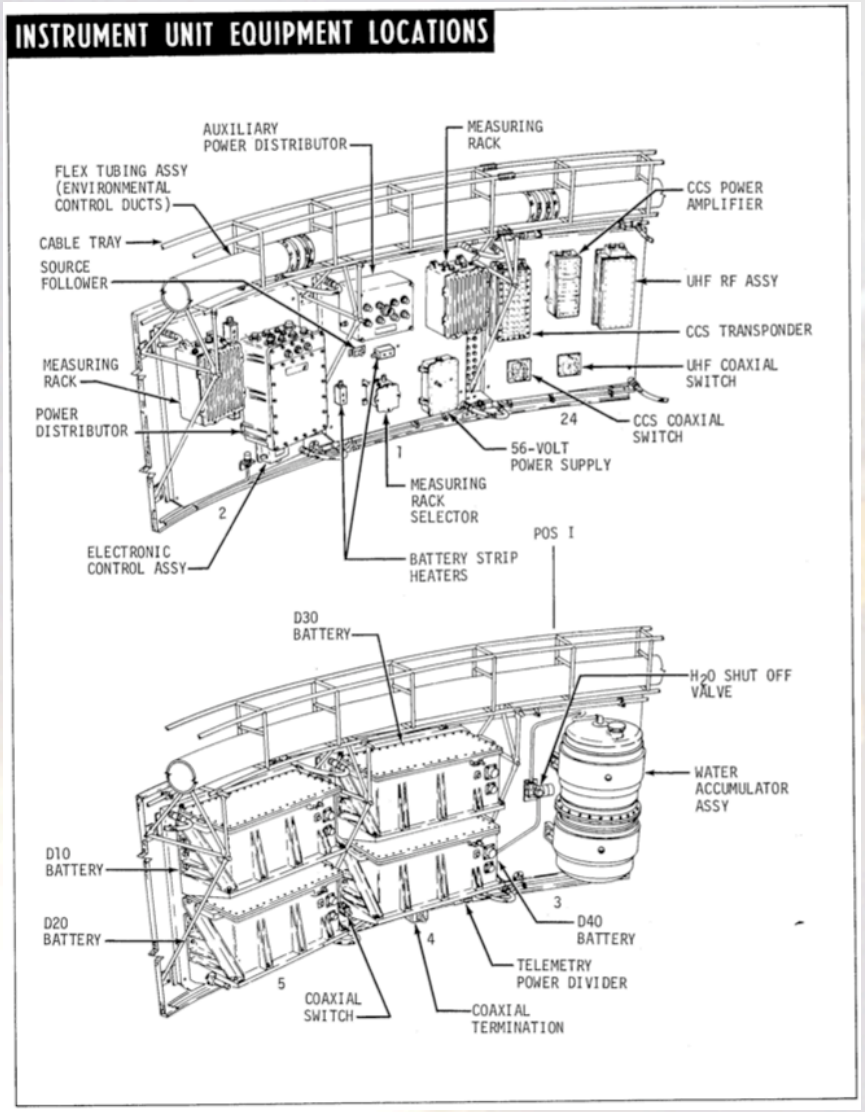
Saturn Instrument Unit

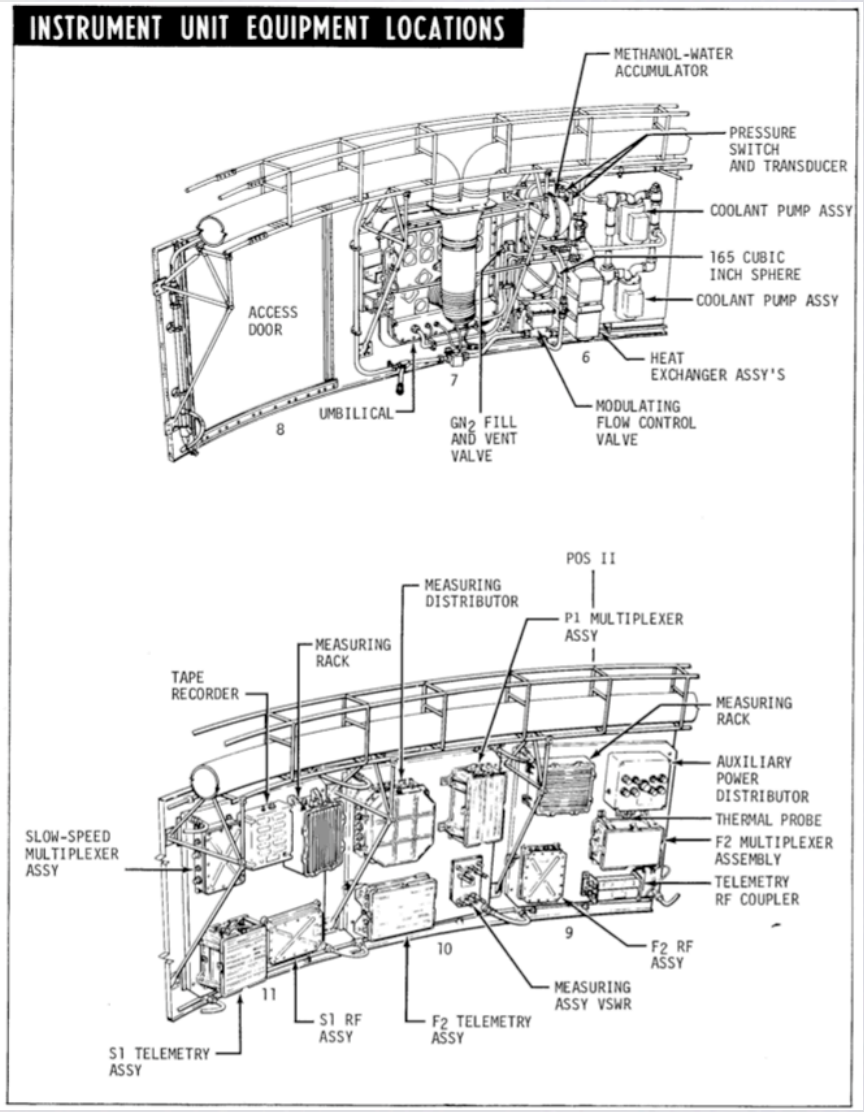


IU Structural Details

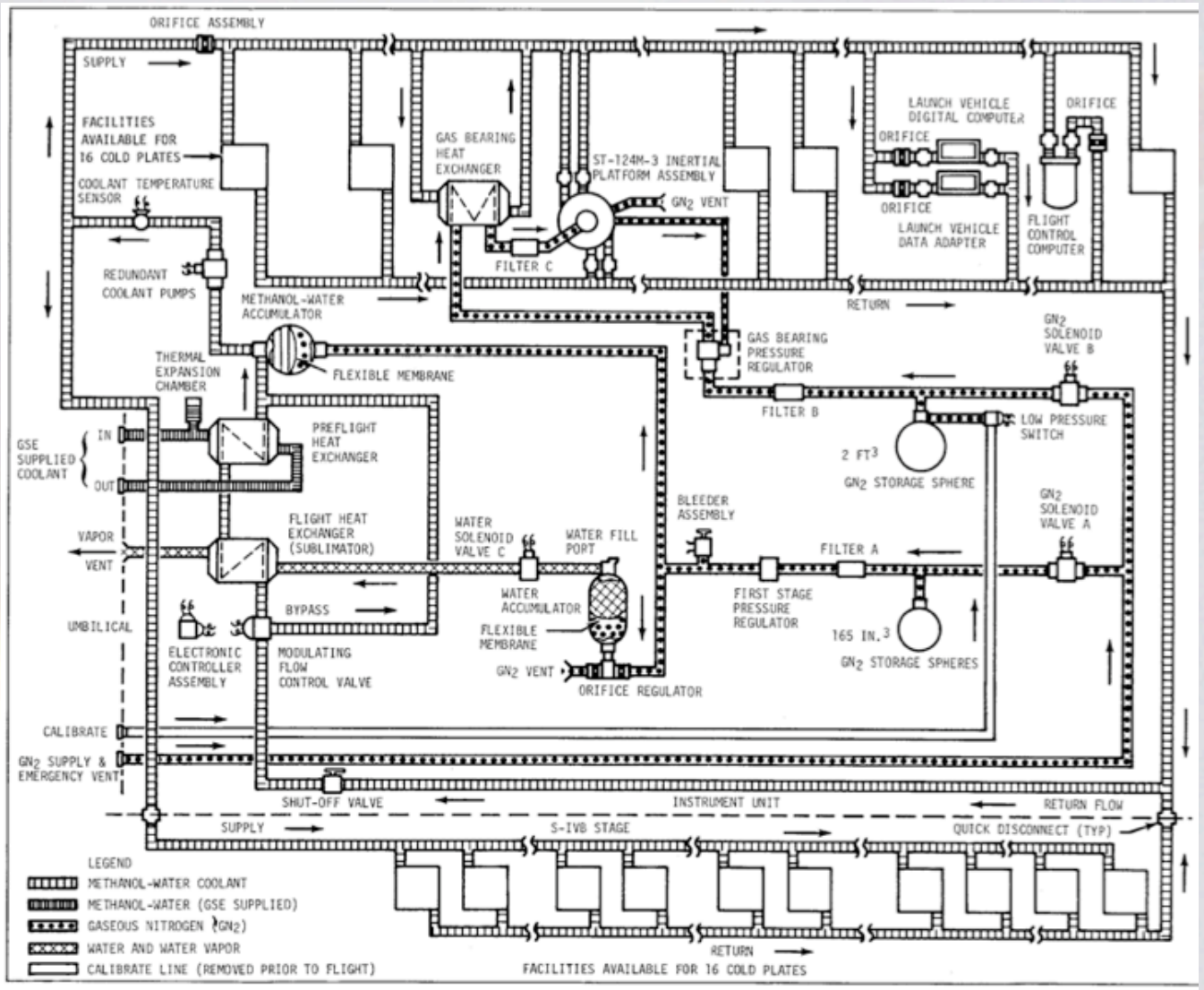


IU Equipment Locations

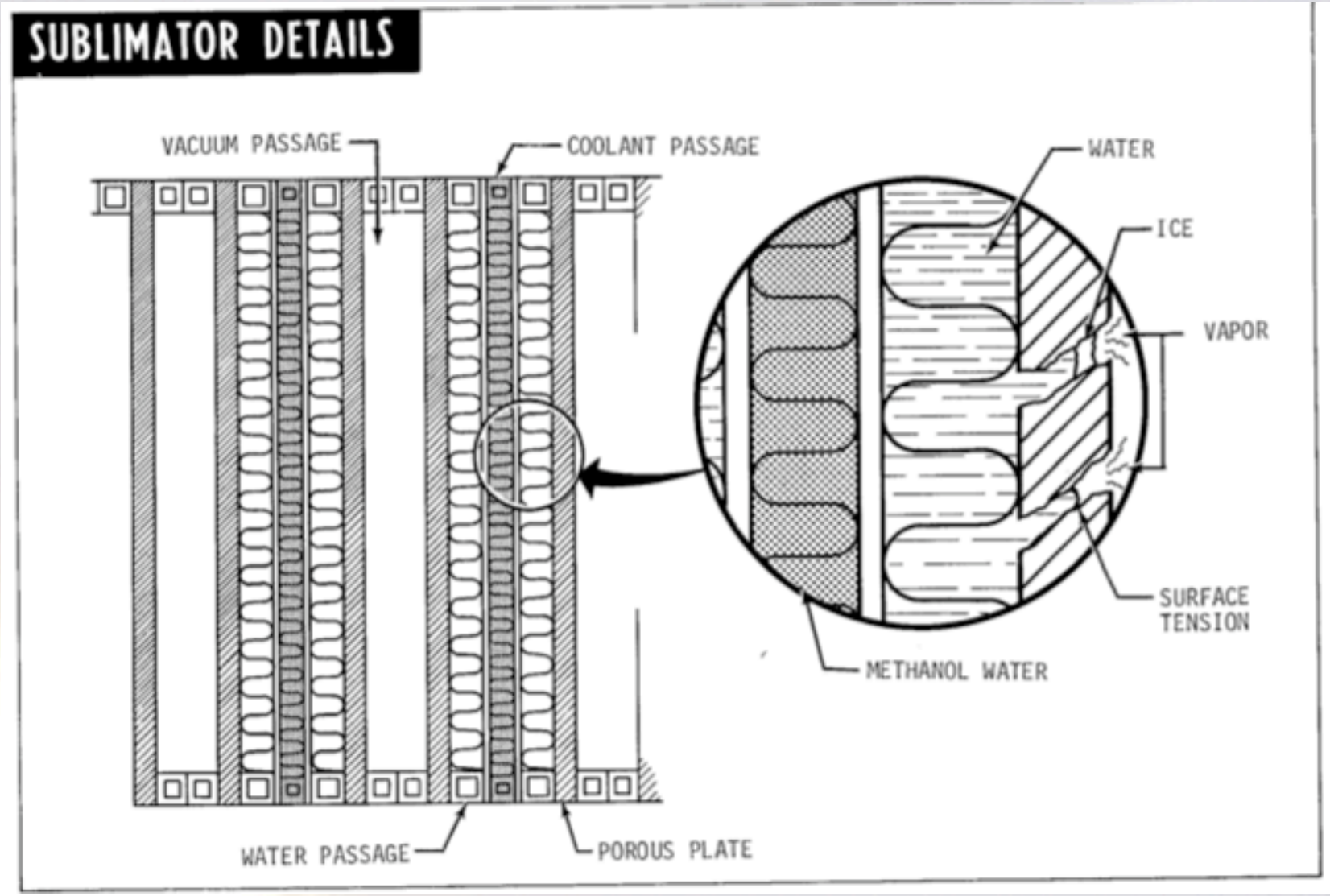




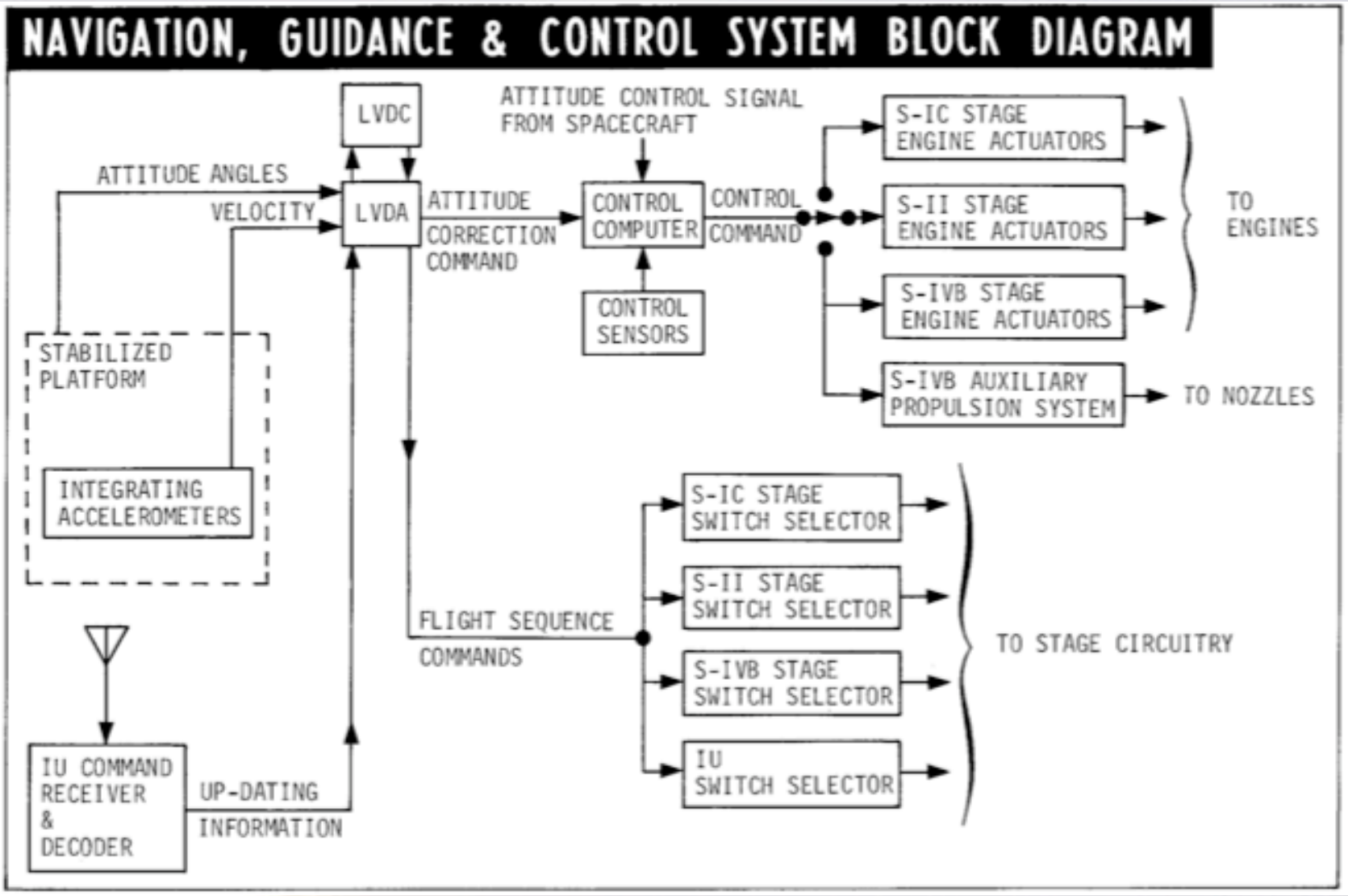
IU Thermal Conditioning System



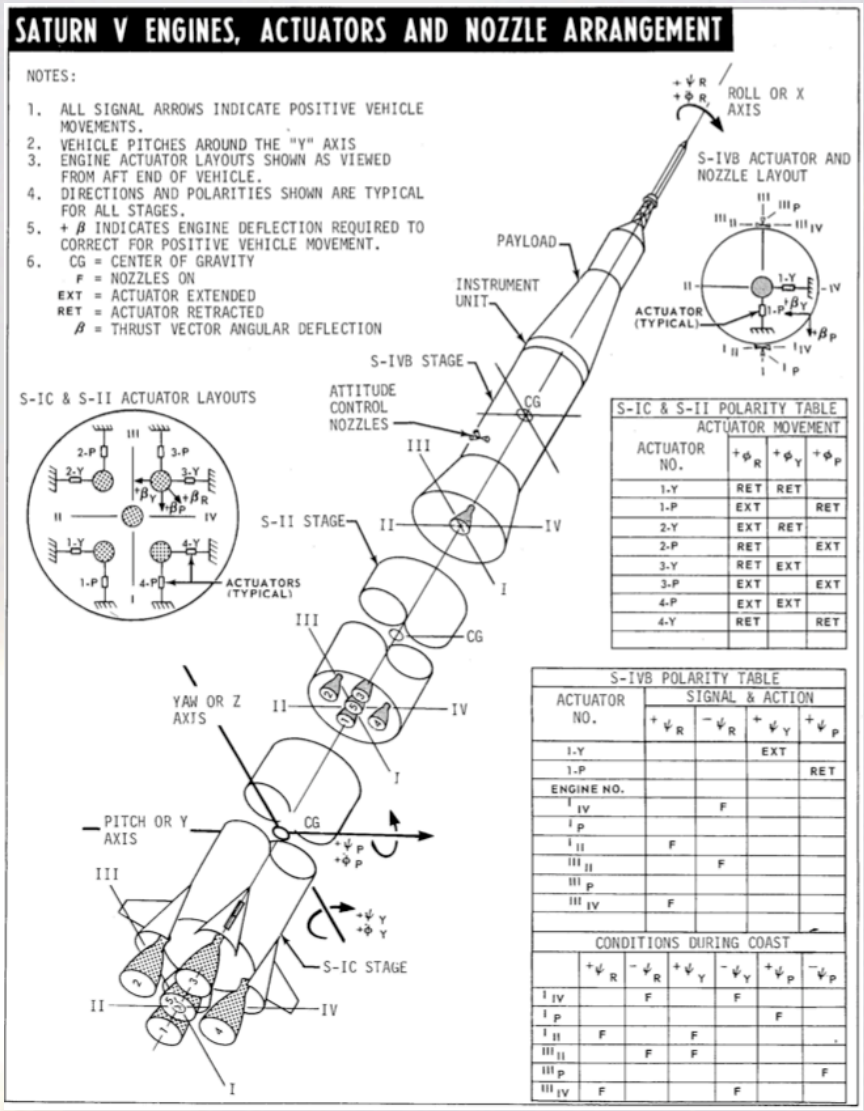
IU Thermal Conditioning Sublimator



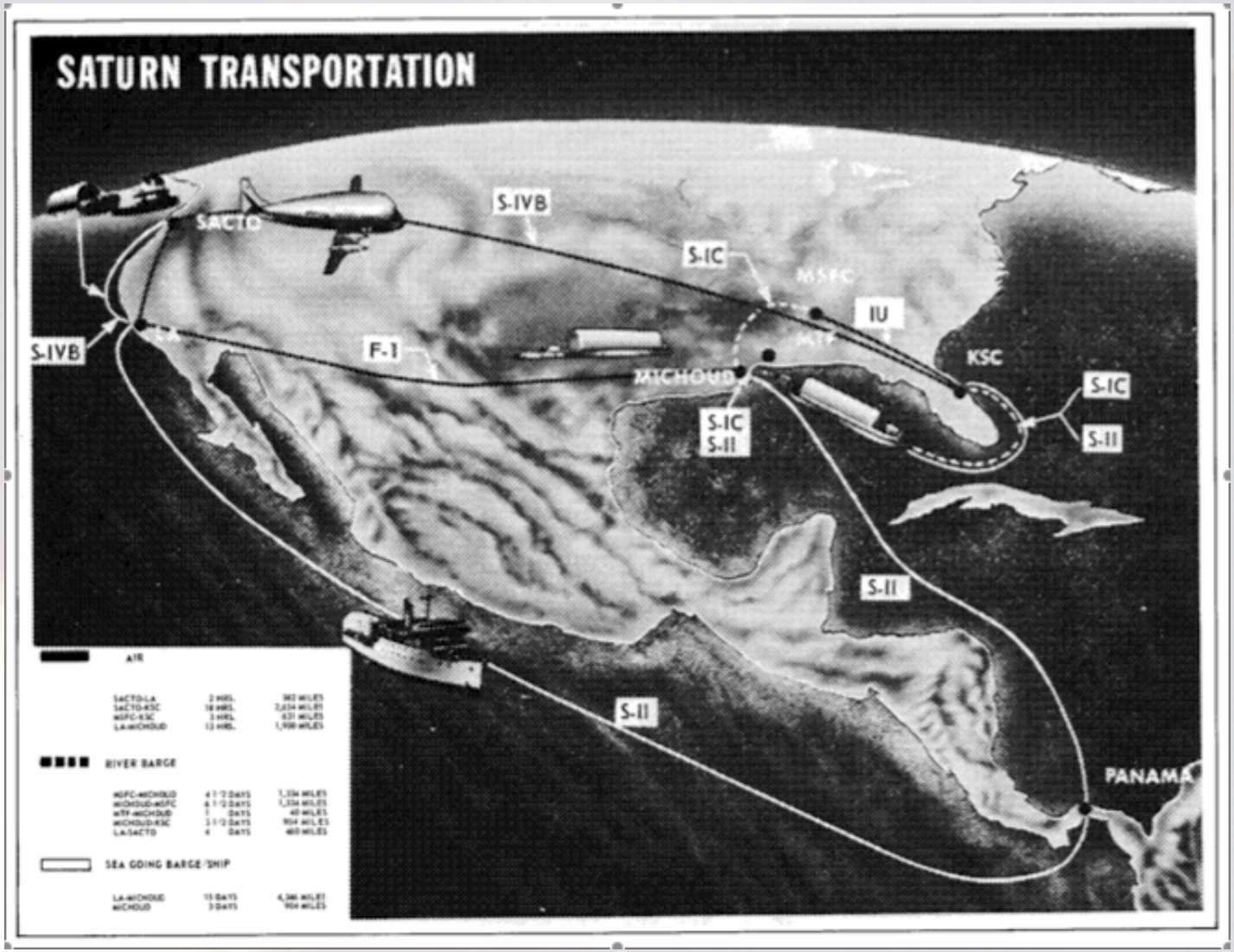
GN&C System Block Diagram



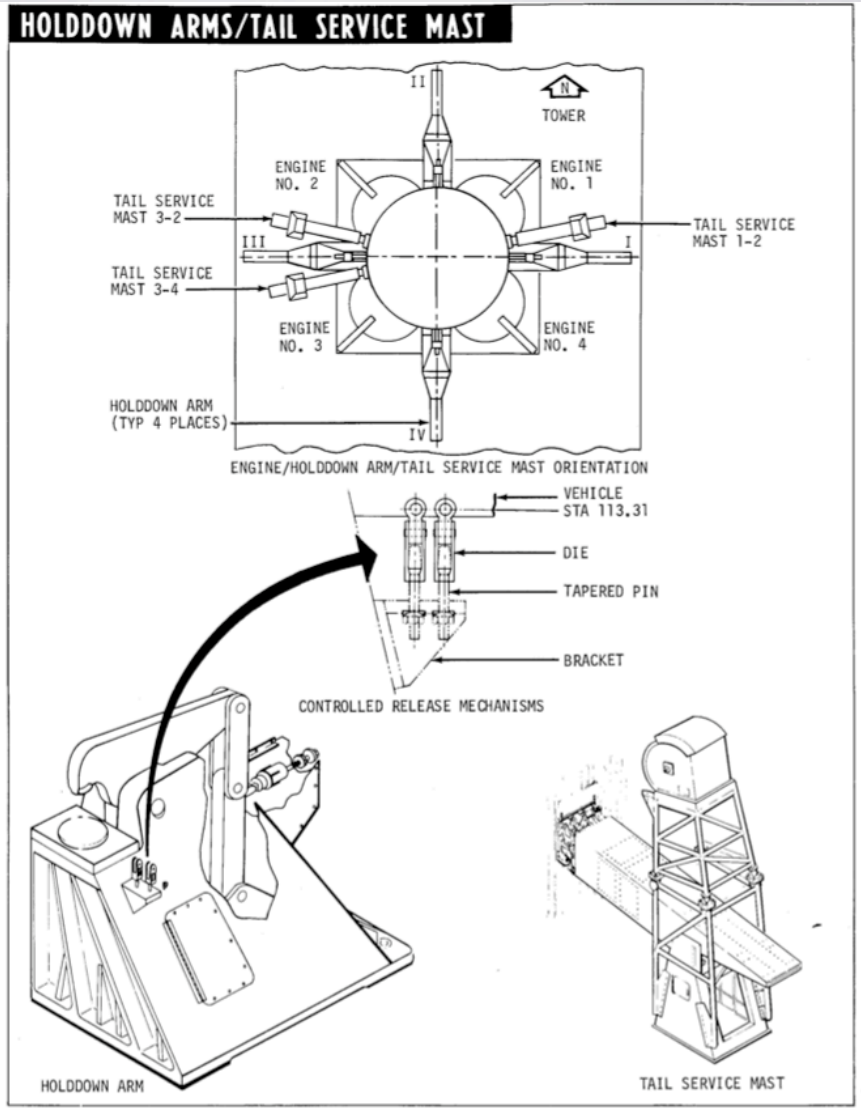
Vehicle Control Actuators



Ground Handling of Components



Launch Pad Interfaces



Emergency Pad Egress