Gaits and Locomotion

- Locomotion
- Metabolic energy
- Gaits
- Partial gravity simulation



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Hopping (Airless Flat Planet)

Constant velocity in horizontal direction produces

 $d = V_h t_{flt} = 2 \frac{V_h V_v}{q}$

 $V_h = V \cos \gamma; V_v = V \sin \gamma$



g

 $V_{v,h} V \qquad \text{Use } F = ma \text{ for vertical motion}$ $\dot{V}_{v} = -g \qquad h = V_{v}t - \frac{1}{2}gt^{2}$

 $t_{flt} = 2V_v/g$

 $d = 2\frac{V^2 \sin \gamma \cos \gamma}{I} = \frac{V^2}{I} \sin (2\gamma)$ g



Hopping (Airless Flat Planet)

 $h_{max} = V_v \frac{V_v}{g} - \frac{1}{2}g\left(\frac{V_v}{g}\right)^2 \qquad V_v = \frac{V}{\sqrt{2}}$

4g

 $h_{max} = \frac{V^2}{4\pi} = \frac{\sqrt{gd^2}}{4\pi} = \frac{d}{4\pi}$



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4g



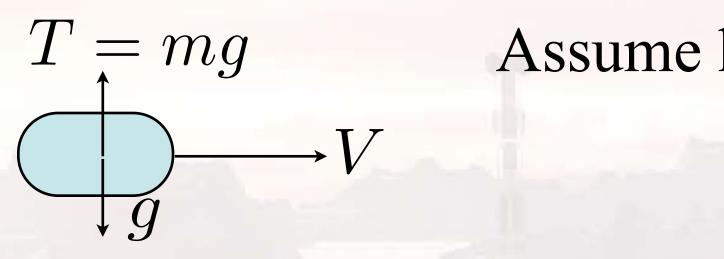
V_v, h V_v, h V Horizontal distance is maximized when $\sin(2\gamma) = 1$ $\gamma_{v_h, d}$ $\gamma_{opt} = \frac{\pi}{2} = 45^o$ $d_{max} = \frac{V^2}{g}$

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 $V = \sqrt{gd} \qquad \Delta V_{total} = 2V = 2\sqrt{gd}$



Propulsive Gliding (Airless Flat Planet)



(includes acceleration and deceleration)

 $t_{flt} = d/V$

Total ΔV becomes

 $\Delta V_{total} = \Delta V_v +$



Assume horizontal velocity is V

 $\Delta V_h = 2V$

$$\Delta V_v = gt_{flt} = \frac{gd}{V}$$

$$\Delta V_h = 2V + \frac{gd}{V}$$



Propulsive Gliding (Airless Flat Planet)

 $\frac{\partial}{\partial V} \left(2V + \frac{gd}{V} \right) = 0 \qquad \qquad 2 - \frac{gd}{V^2} = 0$

 $V_{opt} = \sqrt{\frac{gd}{2}}$



Want to choose V to minimize

 $\Delta V_{total} = 2\sqrt{\frac{gd}{2}} + gd\sqrt{\frac{2}{qd}} = 2\sqrt{2}\sqrt{gd}$



Nondimensional Forms

Define
$$\nu \equiv \frac{V}{\sqrt{dg}}$$

 $\nu_{flat glide} = 2\sqrt{2}$ $\nu_{flat hop} = 2 \qquad \eta = \frac{1}{4}$



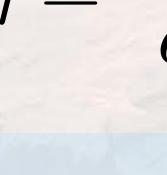
 $\rho \equiv \frac{d}{r} \qquad \eta \equiv \frac{h_{max}}{d}$ $\nu_{spherical glide} = 2\sqrt{2-\rho} \quad (0 \le \rho \le 1)$



Multiple Hops

- Assume n hops between origin and destination
- At each intermediate "touchdown", vv has to be reversed

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 $\Delta V_{total} = 2V + 2(n-1)V_v$ $t_{peak} = \frac{V_v}{g} \qquad \qquad t_{total} = 2nt_{peak} = 2n\frac{V_v}{g}$ $d = V_h t_{total} = \frac{2n}{q} V_h V_v \quad V_v = \sqrt{2gh_{max}} \quad \nu_v = \sqrt{\frac{2\eta}{n}}$ $\nu \equiv \frac{V}{\sqrt{dg}} \qquad \eta \equiv \frac{h_{max}}{d/n} \quad V_h = \frac{dg}{2nV_v} \quad \nu_h = \frac{1}{2}\sqrt{\frac{1}{2n\eta}}$



Multiple Hop Analysis $\Delta \nu = 2\nu + 2(n-1)\nu_v$

 $\Delta \nu = 2\sqrt{\frac{2\eta}{n}} + \frac{2\eta}{n}$ $\frac{\partial \Delta \nu}{\partial \eta} = \left[\frac{1}{\sqrt{\frac{2\eta}{n} + \frac{1}{8n\eta}}} \left(\frac{2}{n} + \frac{1}{8n\eta} \right) \right]$

(In general, solve numerically)



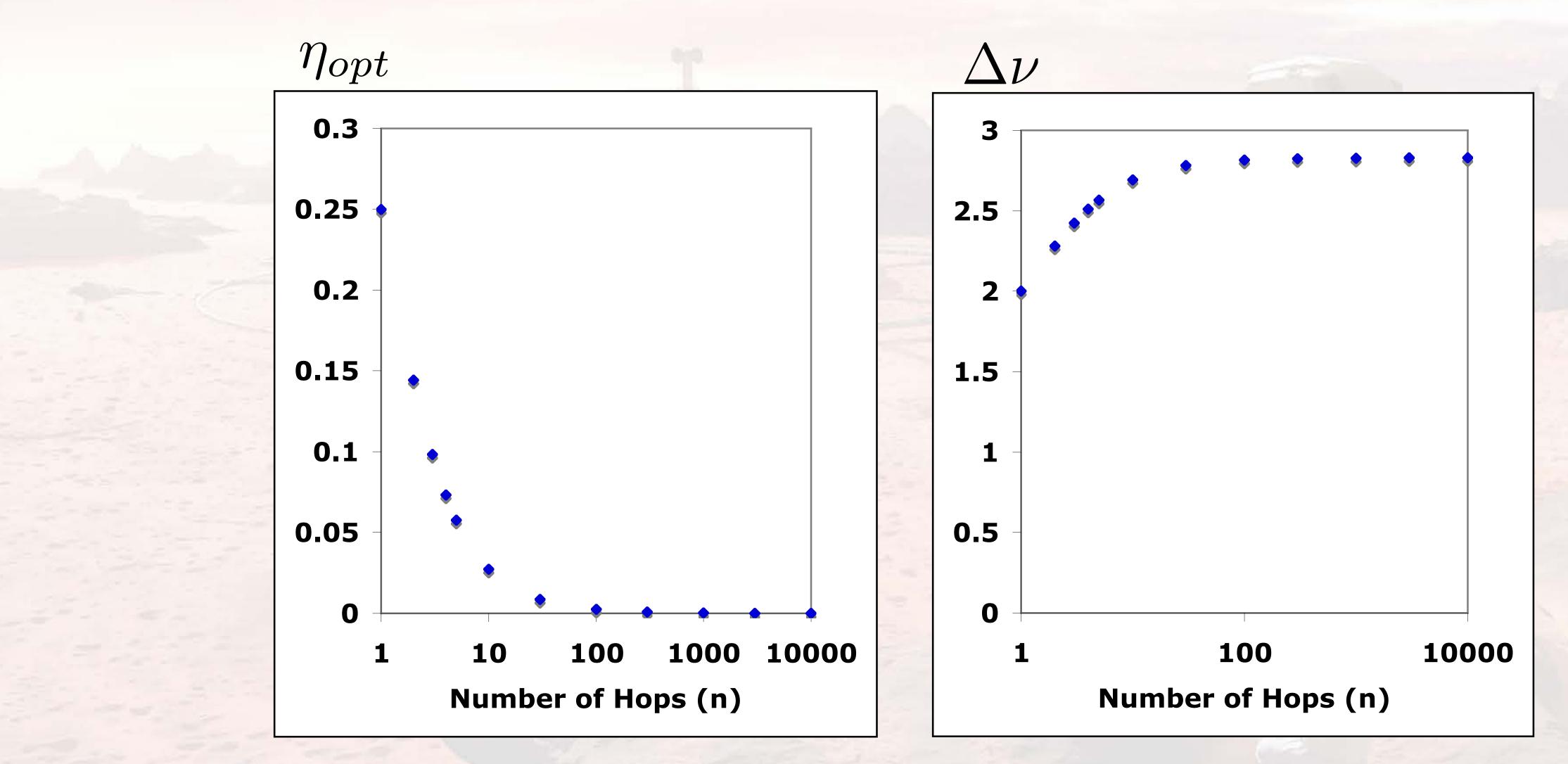
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$$\nu = 2\sqrt{\nu_v^2 + \nu_h^2 + 2(n-1)\nu_v}$$
$$\frac{1}{8n\eta} + 2(n-1)\sqrt{\frac{2\eta}{n}}$$
$$-\frac{1}{8n\eta^2}\bigg)\bigg] + (n-1)\sqrt{\frac{2}{n\eta}} = 0$$

Analytically messy, but note that for $n = 1 \Rightarrow \eta_{opt} = \frac{1}{\Lambda}$



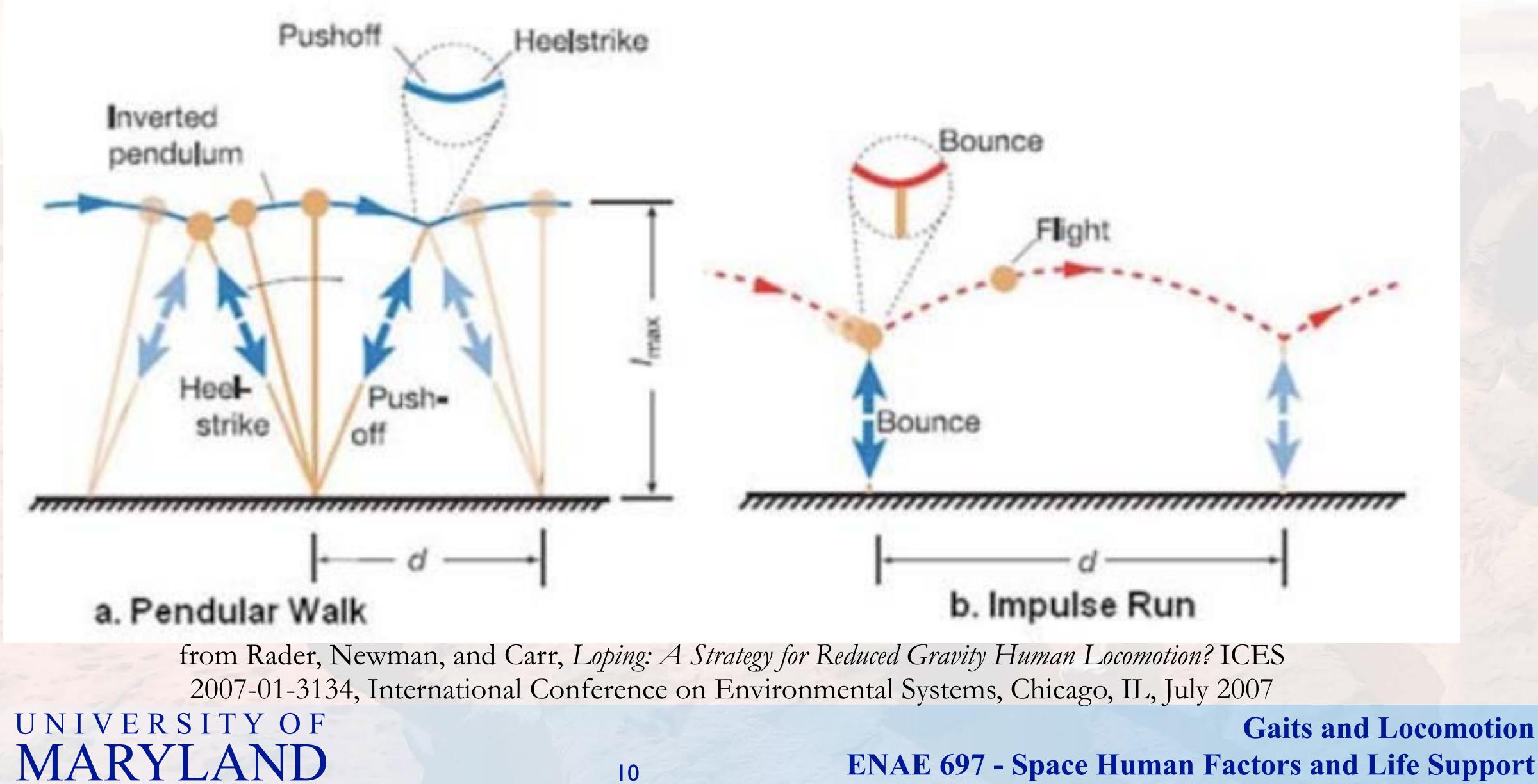
Optimal Solutions for Multiple Hops



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Walking and Running Gaits



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Implications of Pendulum Motion

 $P \approx 2\pi \sqrt{\frac{L}{q}}$

 $I = mL^2$

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• Basic period of a pendulum

Inertias of point and distributed masses

• For 0.75 m leg length, UNIVERSITY OF MARYLAND

 $I = \frac{1}{3}mL^2$

$P_{walking} \approx 1.0 \text{ sec (on Earth)}$



Walking Froude Number

- Froude number is ratio between inertial and gravitational forces
- planing motion)
- Walking Froude number

• Feet leave the ground at Fr=1 • Walk-run transition typically occurs ~Fr=0.5

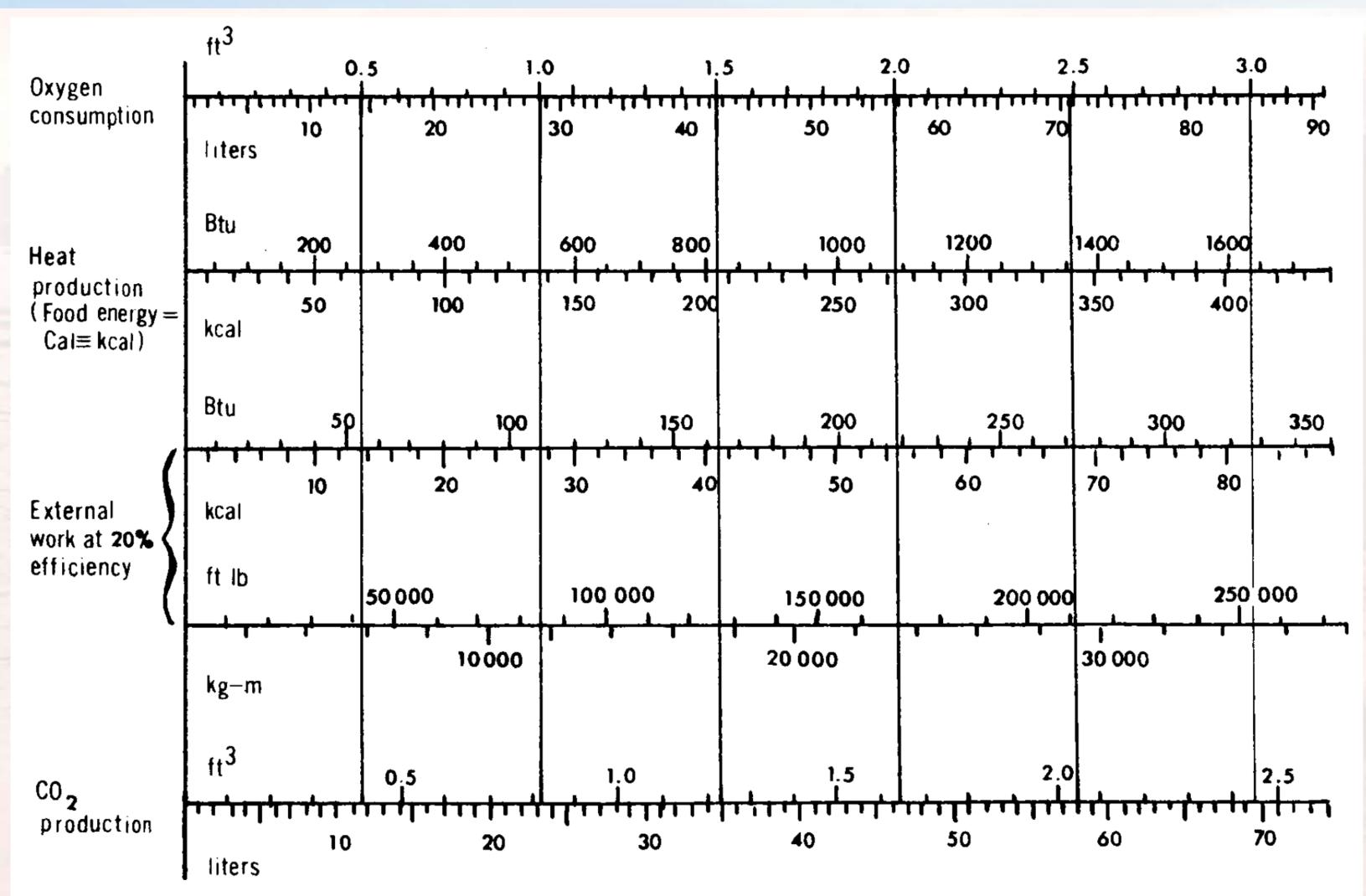


• Primary application is boat speed (transition between displacement and

 $Fr = \frac{V^2}{gL} = \frac{a_{centripetal}}{g}$



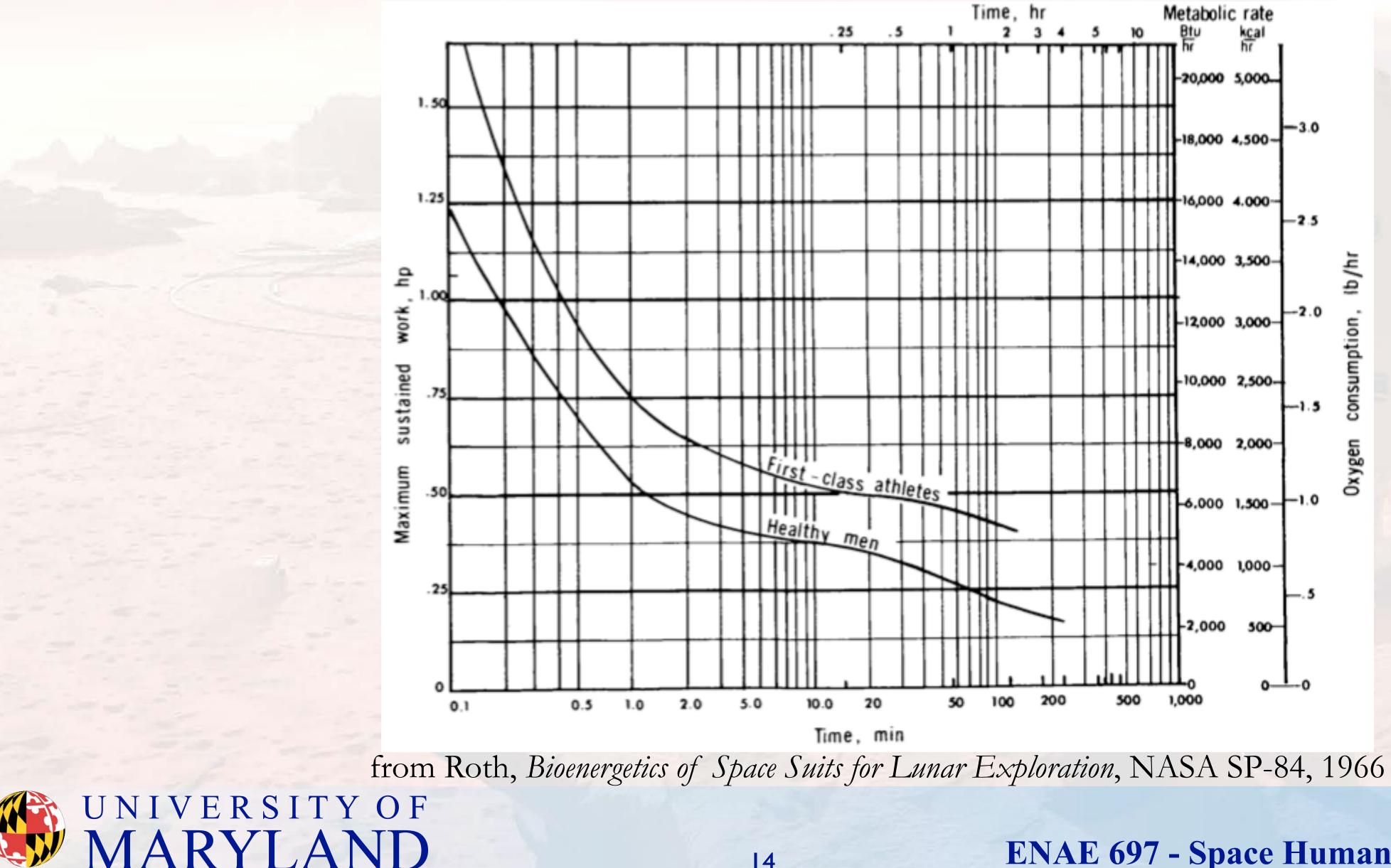
Metabolic Costs of Physical Exertion



from Roth, Bioenergetics of Space Suits for Lunar Exploration, NASA SP-84, 1966 UNIVERSITY OF MARYLAND **Gaits and Locomotion ENAE 697 - Space Human Factors and Life Support** 13

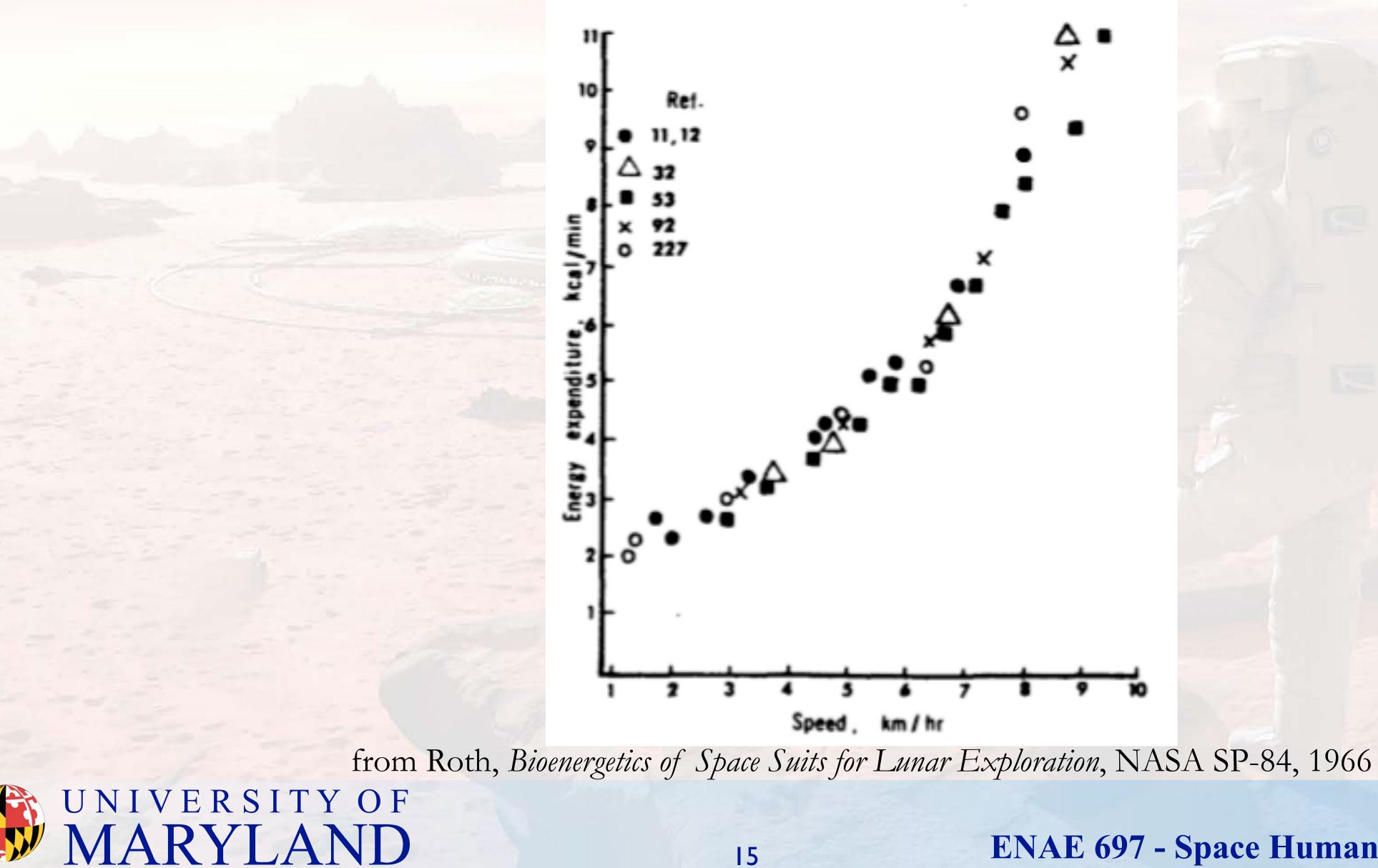


Maximum Sustained Work Output



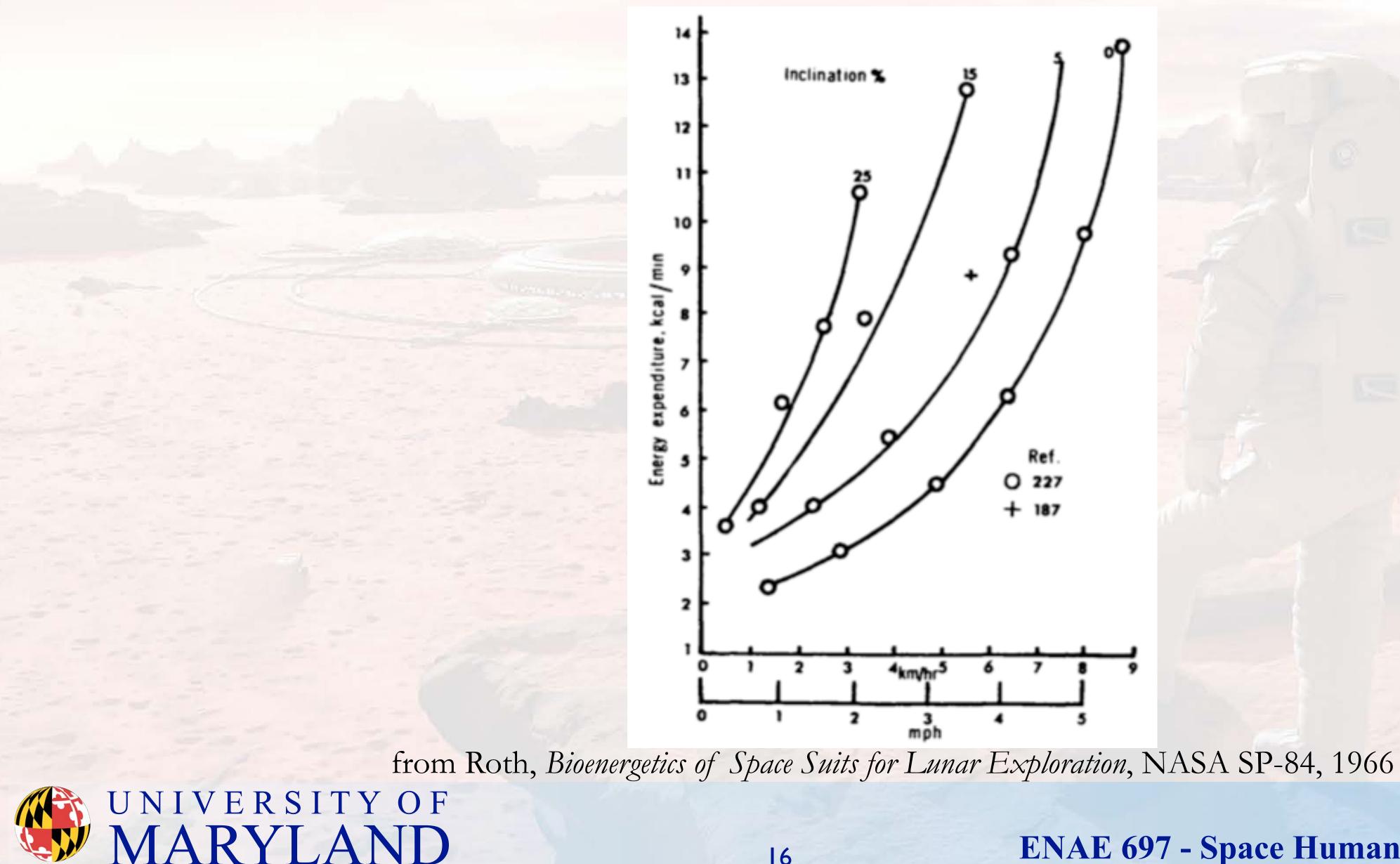


Energy Expenditure in Walking





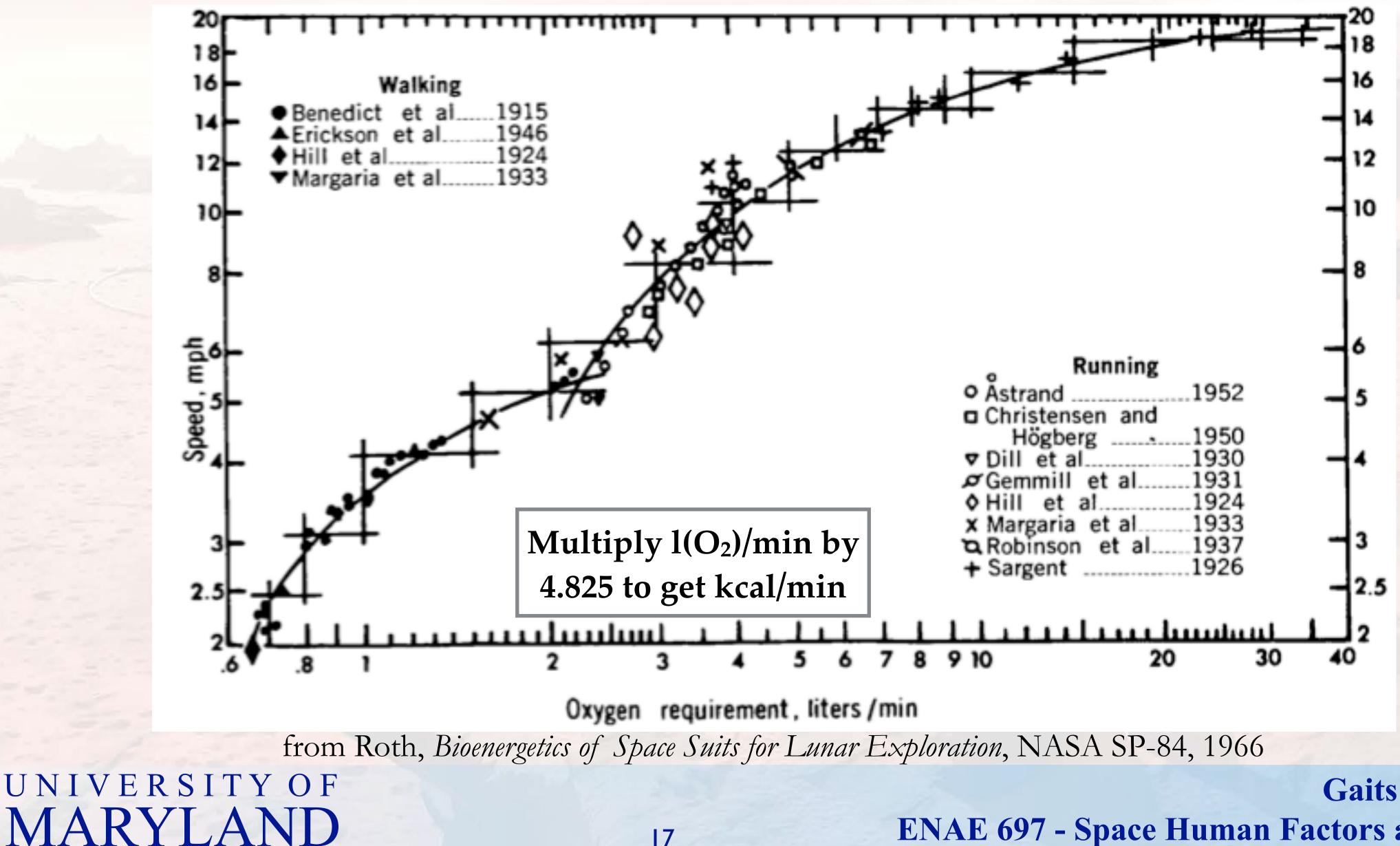
O2 Requirement to Walk and Run



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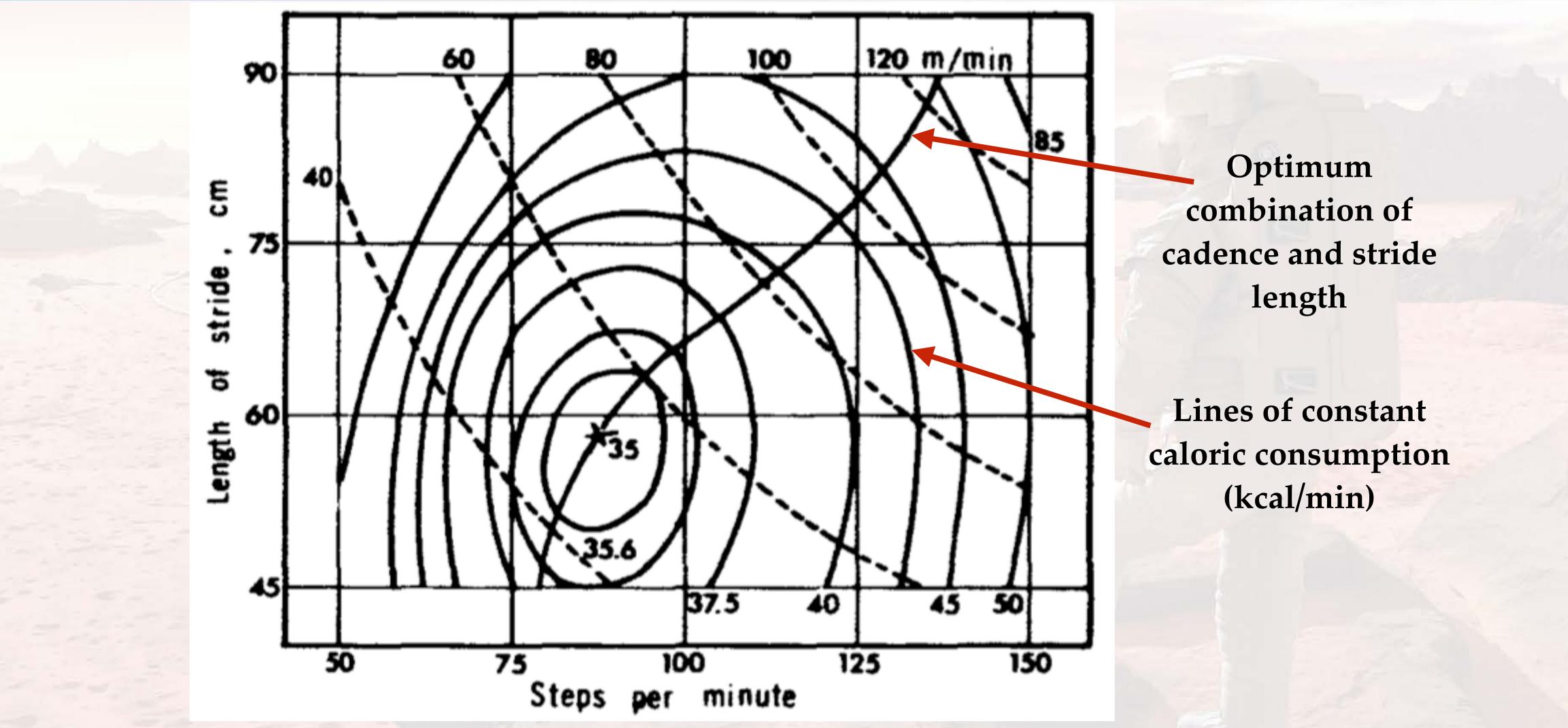


O₂ Requirement to Walk and Run





Effect of Stride and Cadence

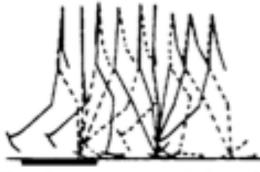


from Roth, Bioenergetics of Space Suits for Lunar Exploration, NASA SP-84, 1966





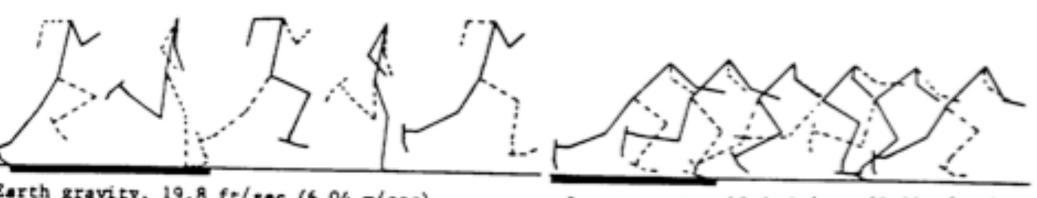
Body Inclination in Variable Gravity



Earth gravity, 4.0 ft/sec (1.22 m/sec)



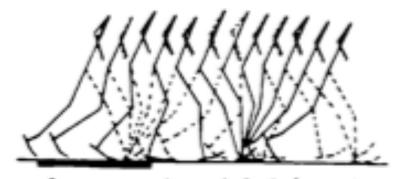
Earth gravity, 10.0 ft/sec (3.01 m/sec)



Earth gravity, 19.8 ft/sec (6.04 m/sec)

from Partial Gravity Habitat Study, Sasakawa International Center for Space Architecture, University of Houston, 1989





Lunar gravity, 4.1 ft/sec (1.25 m/sec)

(a) Walk,

Lunar gravity, 10.5 ft/sec (3.20 m/sec)

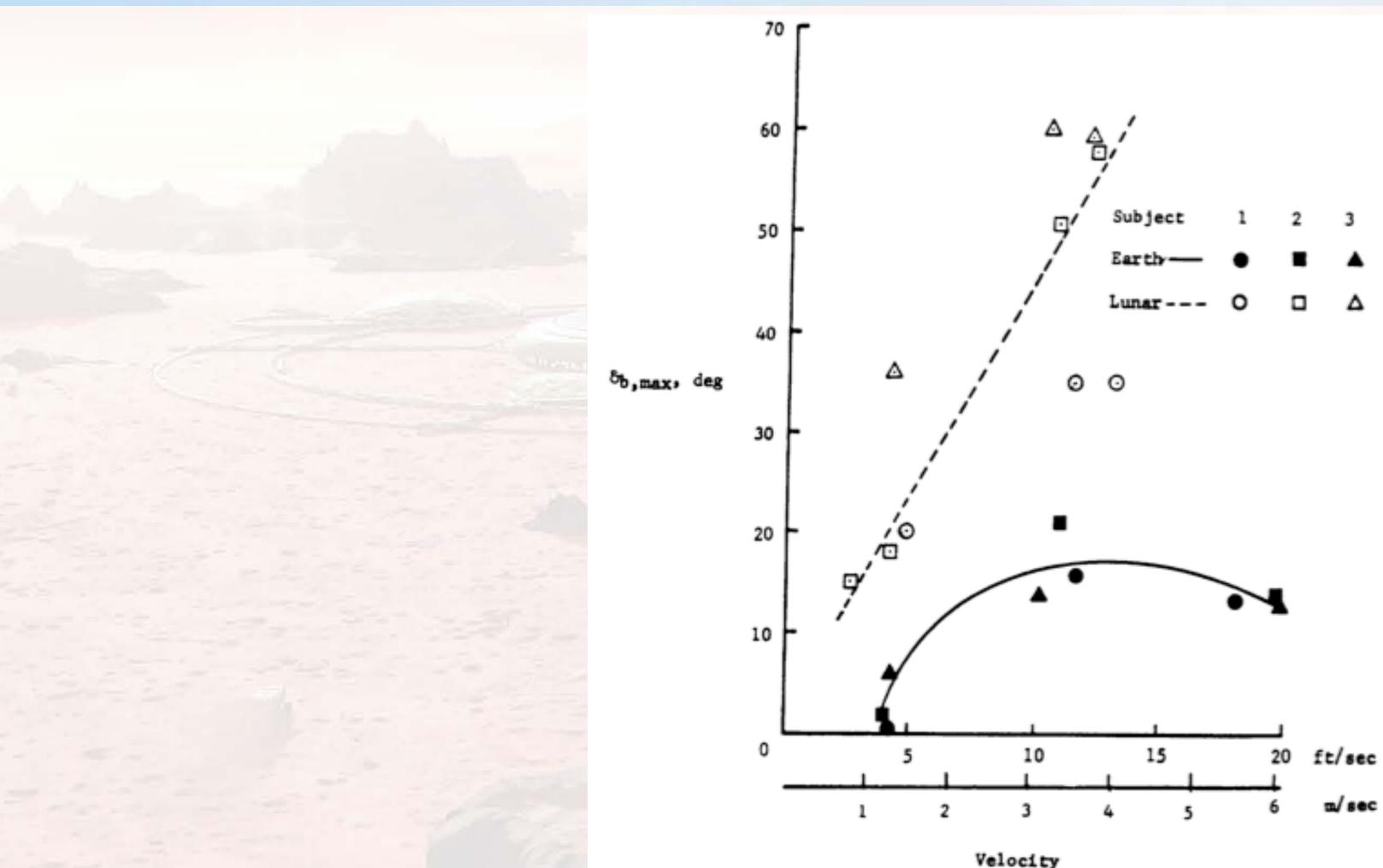
(b) Loss

Lunar gravity, 13.1 ft/sec (3.99 m/sec)

(c) Sprint,



Body Inclination in Variable Gravity

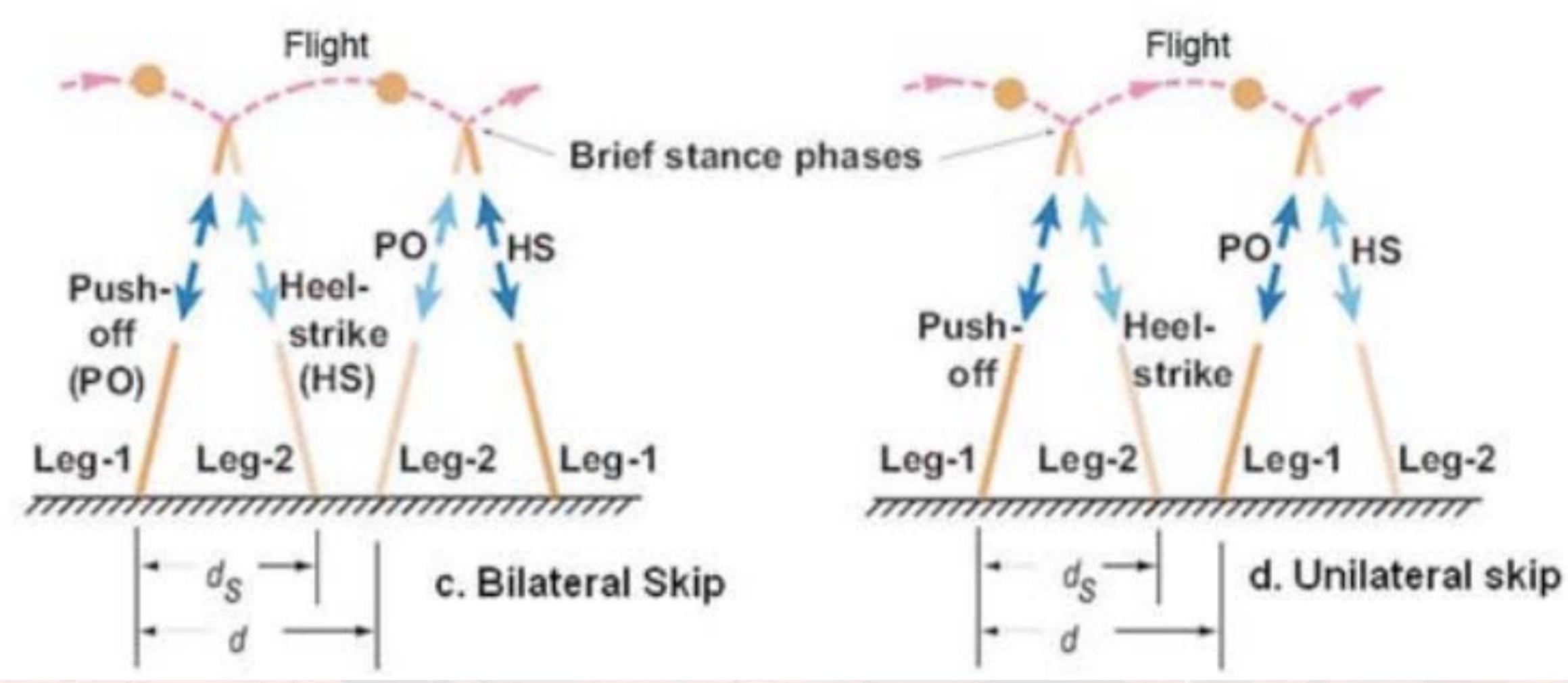


from Partial Gravity Habitat Study, Sasakawa International Center for Space Architecture, University of Houston, 1989





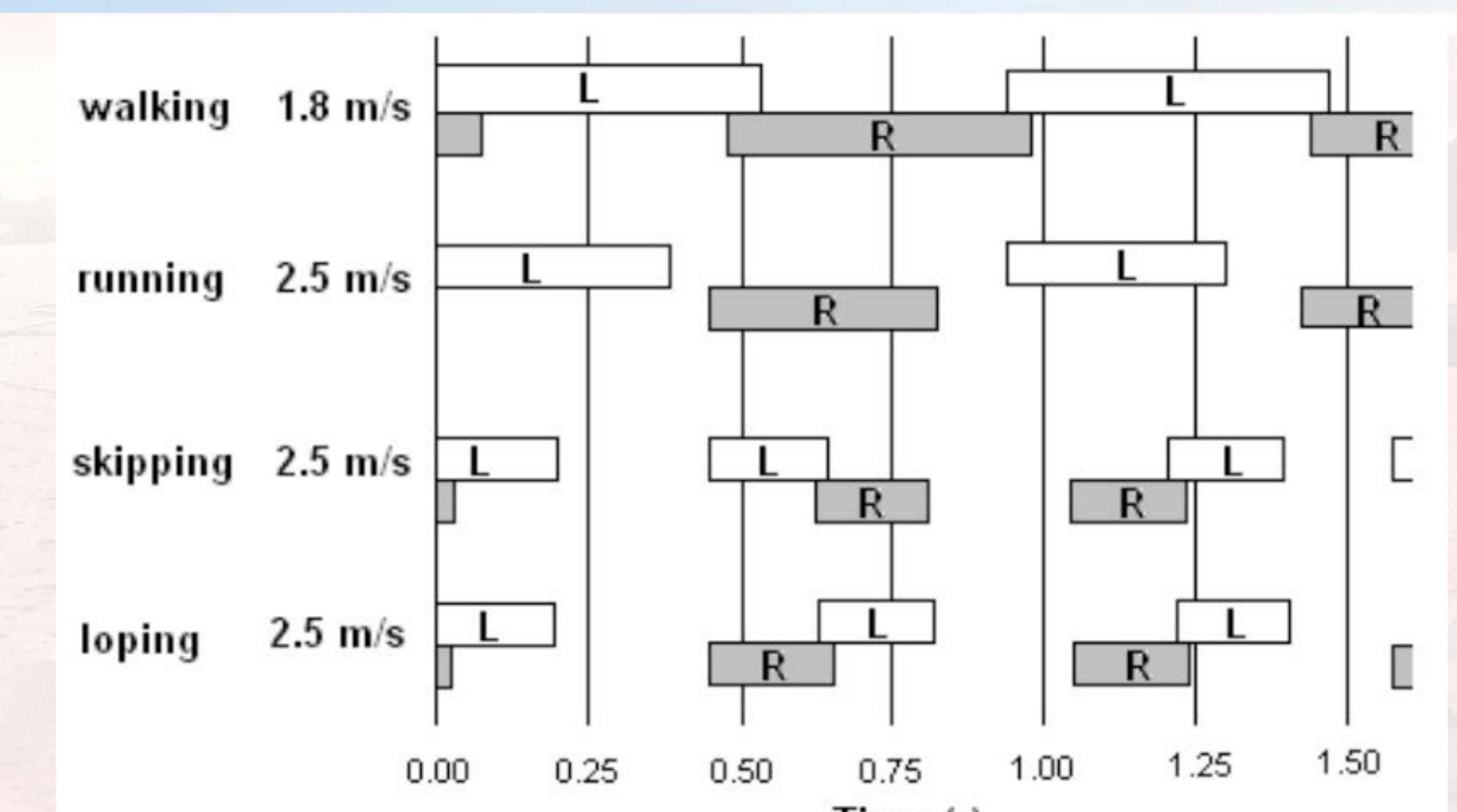
Skipping and Loping Gaits



from Rader, Newman, and Carr, Loping: A Strategy for Reduced Gravity Human Locomotion? ICES 2007-01-3134, International Conference on Environmental Systems, Chicago, IL, July 2007 UNIVERSITY OF MARYLAND 21 ENAE 697 - Space Human Factors and Life Support



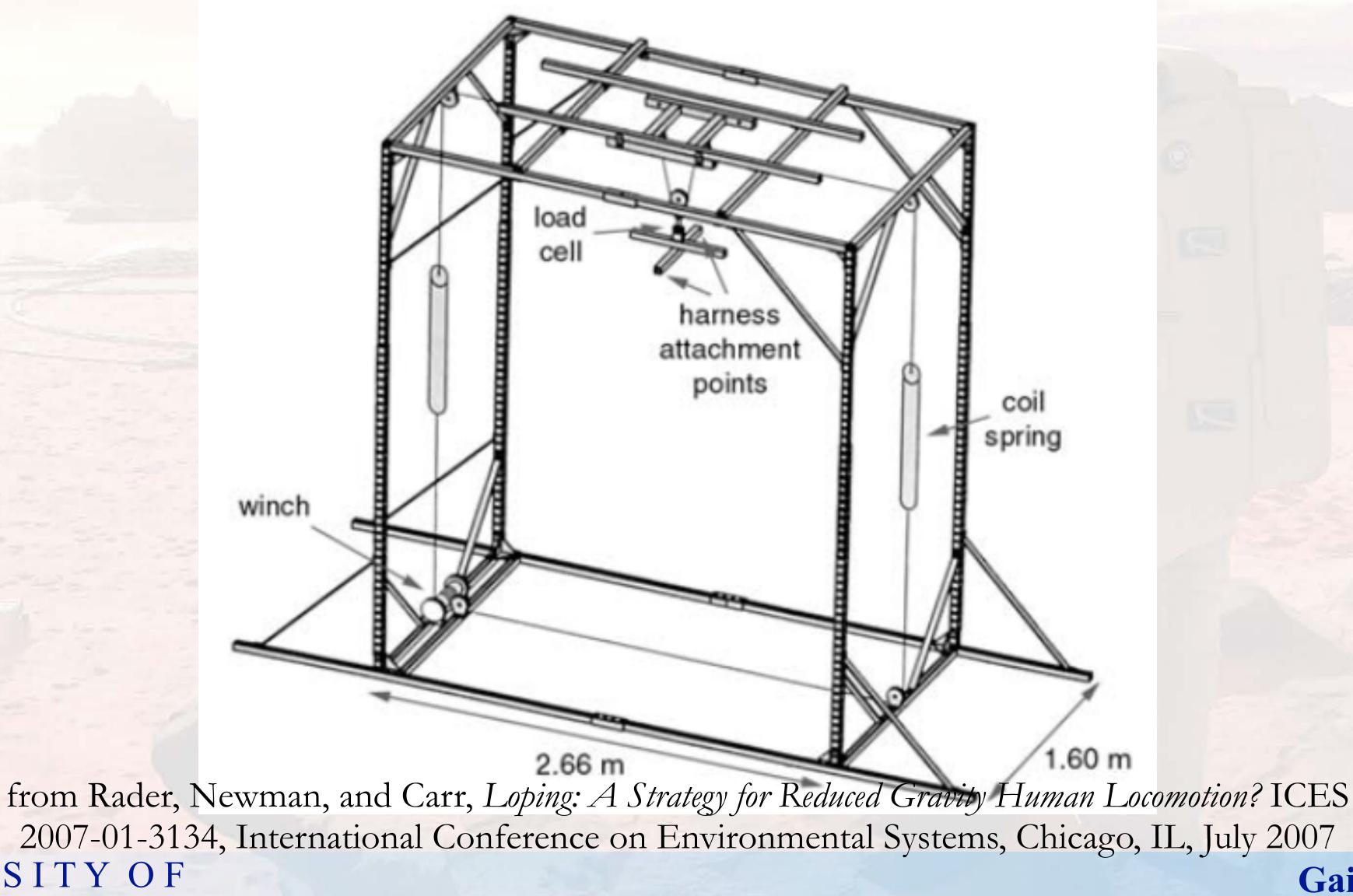
Gait Footfalls



from Rader, Newman, and Carr, Loping: A Strategy for Reduced Gravity Human Locomotion? ICES 2007-01-3134, International Conference on Environmental Systems, Chicago, IL, July 2007 UNIVERSITY OF MARYLAND 22 ENAE 697 - Space Human Factors and Life Support



MIT Gravity Offset System

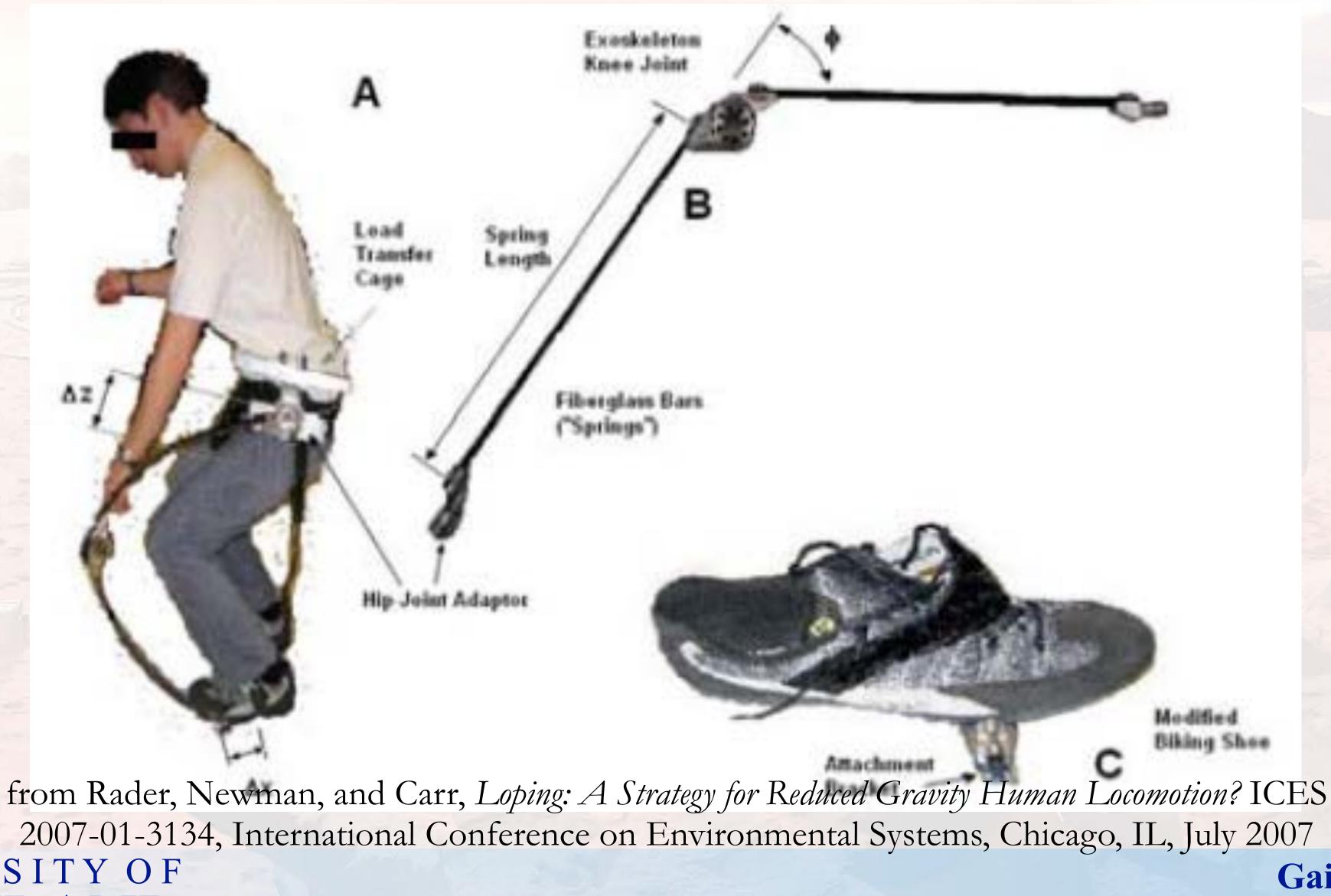


from Rader, Newman, and Carr, *Loping:* 2007-01-3134, International Conference UNIVERSITY OF MARYLAND 23

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MIT Exoskeletal Suit Joint Simulator



from Rader, Newman, and Carr, *Loping: A* 2007-01-3134, International Conference UNIVERSITYOF MARYLAND 24

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JSC Walkback Tests (2006-2007)

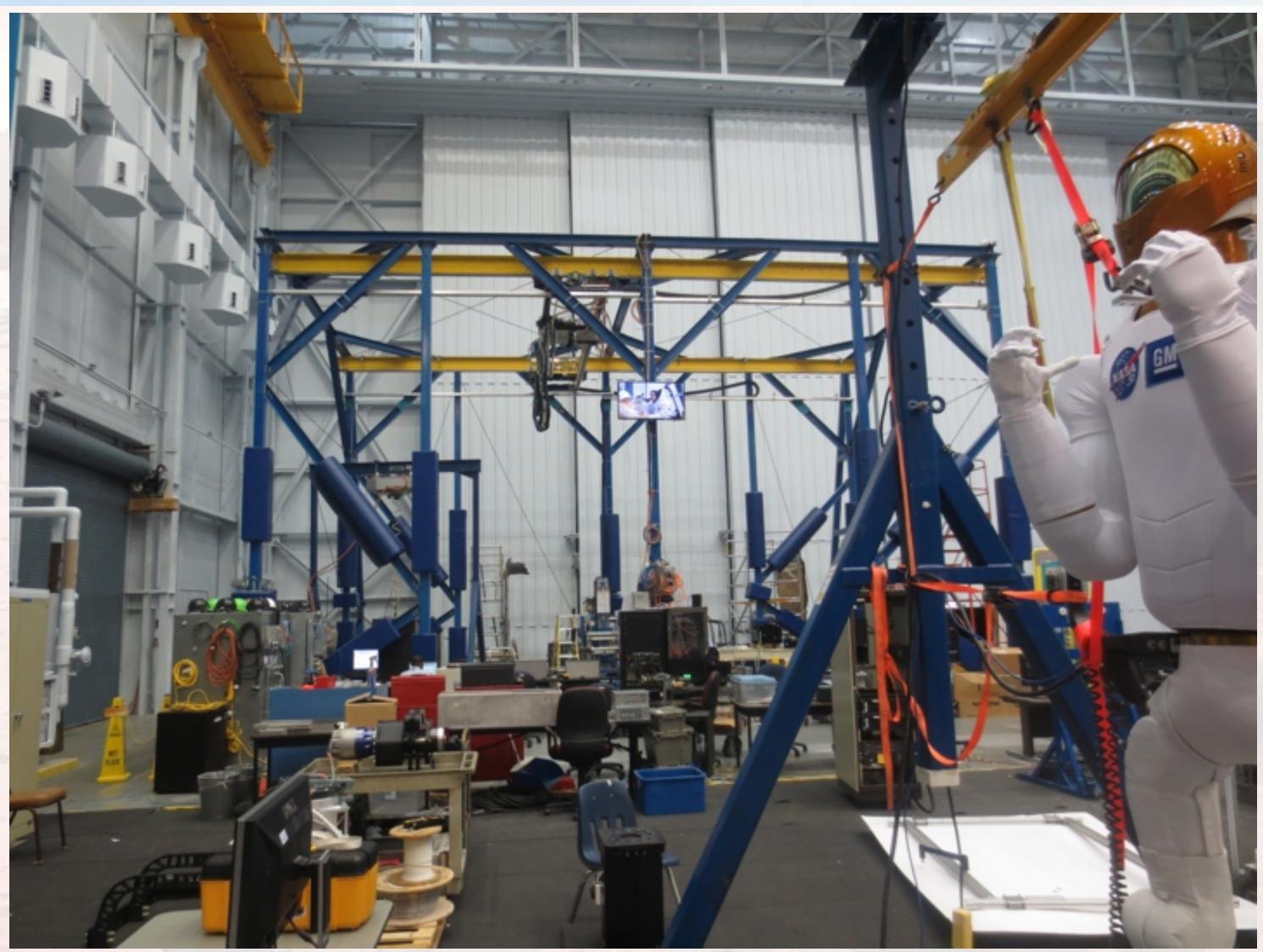
- lunar gravity
- Extra-large treadmill
- Pogo (pneumatic suspension) gravity offset device
- Mk. III suit
- Vicon motion tracking system
- Six test subjects (astronauts)
- Measured VO2, RPE, MCH



• Testing to verify metabolic cost of 10 km walkback from failed rover in



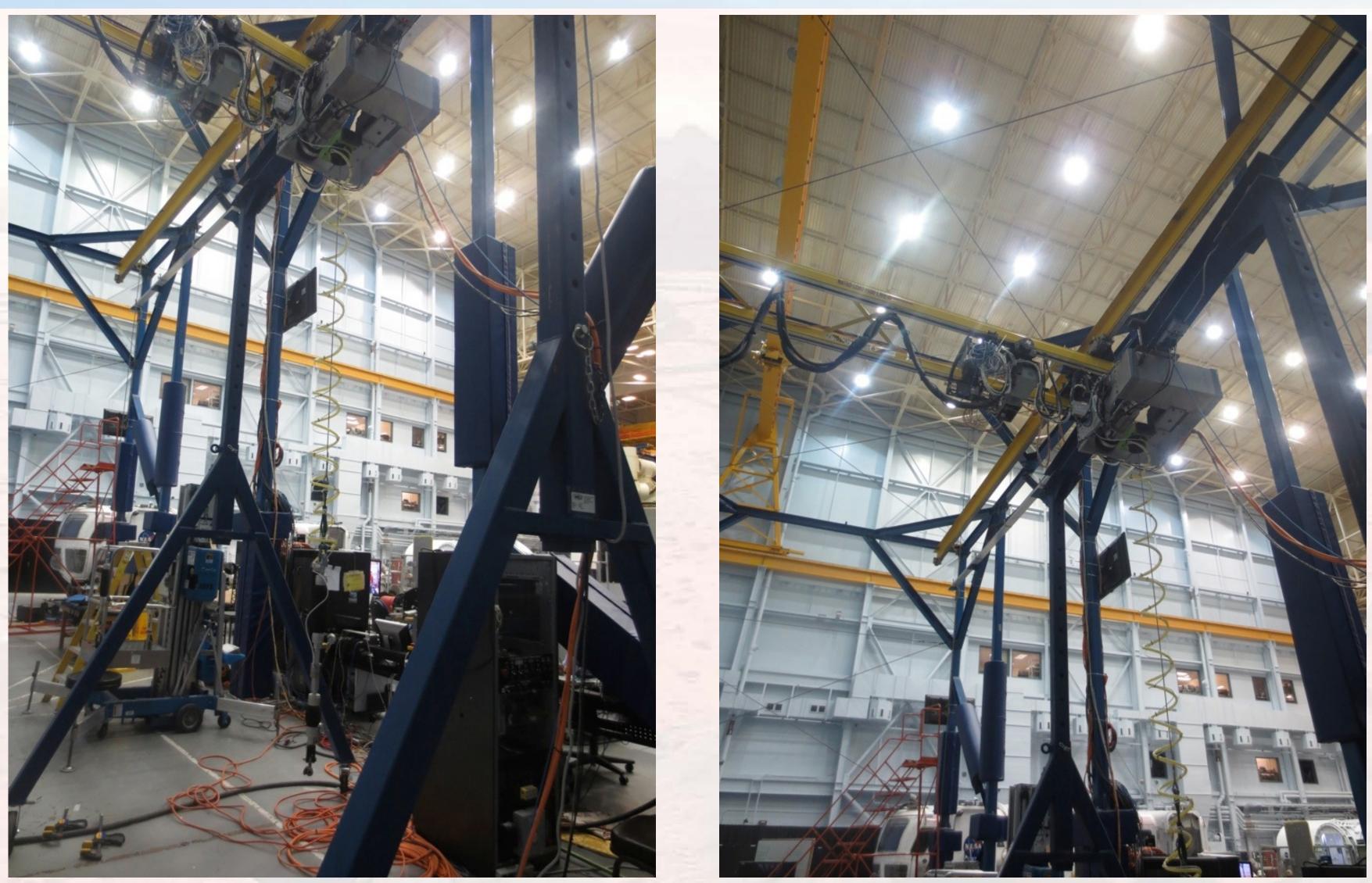
JSC ARGOS Gravity Offset System







ARGOS Suspension and Tracks







ARGOS Suspension System



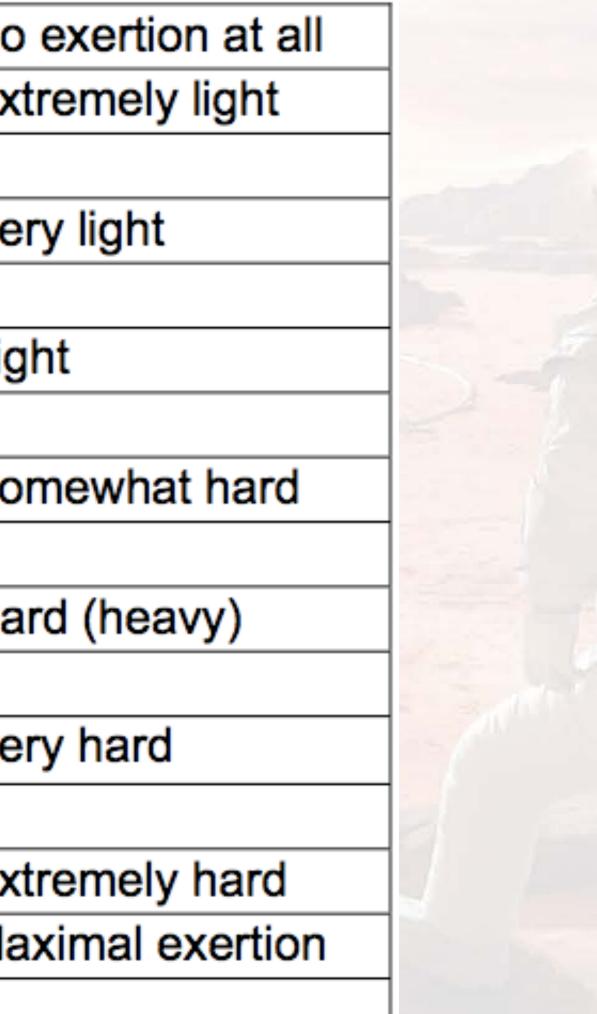




Borg Rating of Perceived Exertion (RPE)

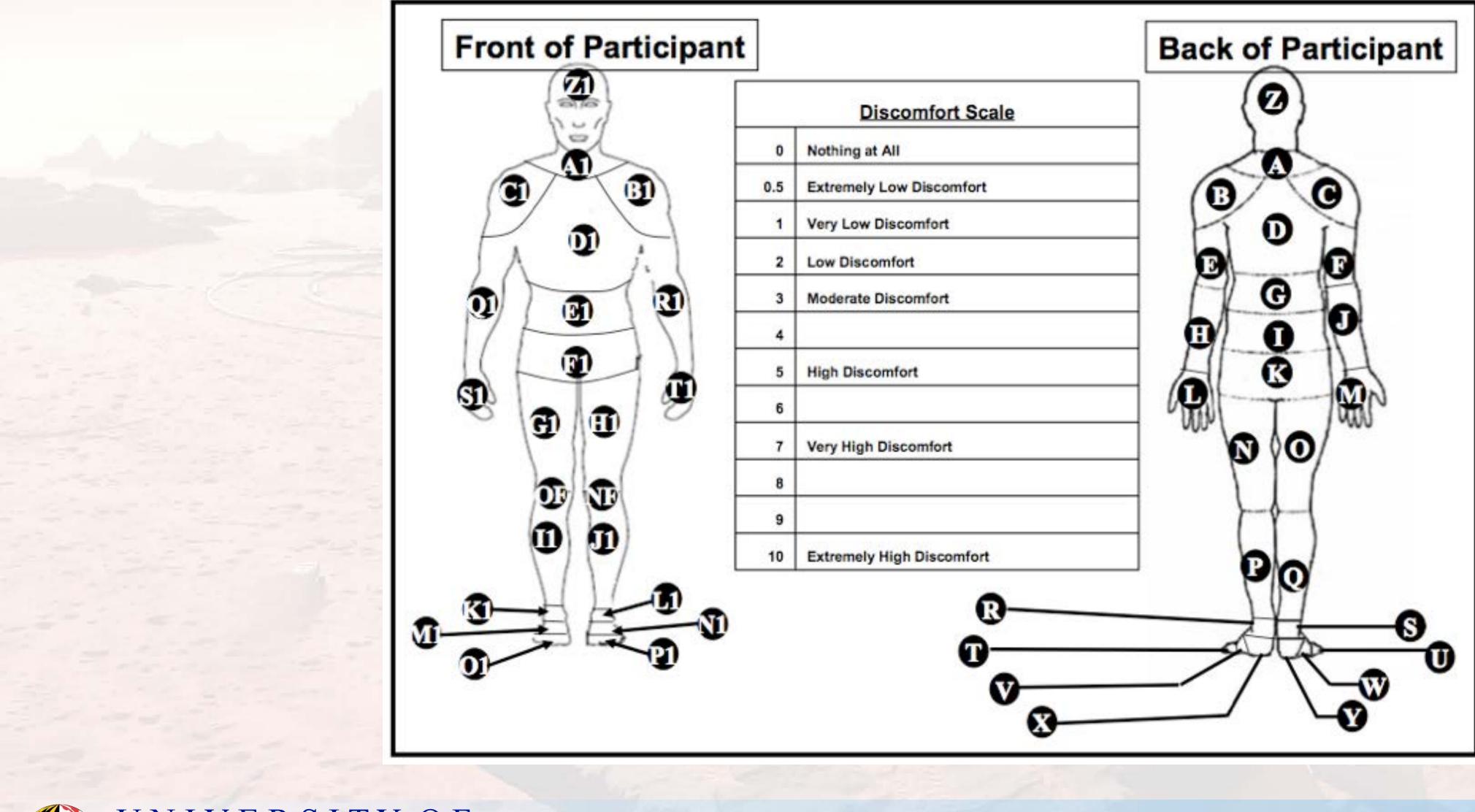
	6	No
	7	Ext
	8	
	9	Ver
	10	
	11	Lig
	12	
	13	Sor
	14	
	15	Har
	16	
	17	Ver
	18	
	19	Ext
	20	Ma
		1
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Corlett & Bishop Discomfort Scale







Speeds Used for Testing

I and the	
Jsed	TO
	lsed

Stage	Speed	
1	X minus 1.1 mph	r
2	X minus 0.8 mph	v
3	X minus 0.5 mph	2 V
P	rs = x	
4	X plus 0.5 mph	/ r
5	X plus 1.5 mph	r r
6	X plus 2.5 mph	C



r the Energy-Velocity Tests:

Comments

Subtract 0.3 mph per stage; need smaller increments for walking

Subtract 0.5 mph to assure walking out of transition zone

No data collected in transition zone

Add 0.5 mph to assure running out of transition zone Add 1.0 mph to distinguish metabolic/biomechanical differences at running speeds



Shirt-Sleeve Suspension







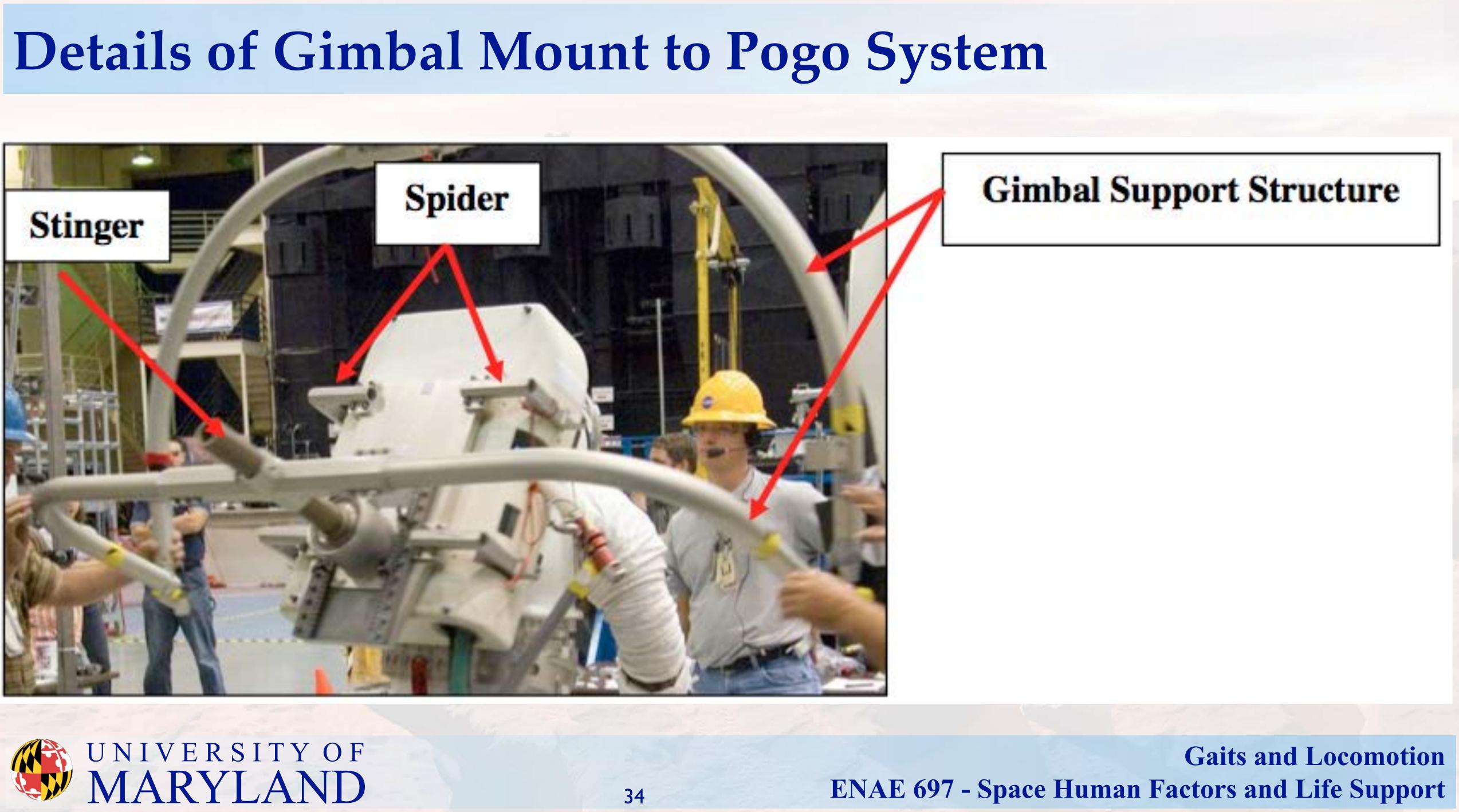
Suited Partial-Gravity Suspension









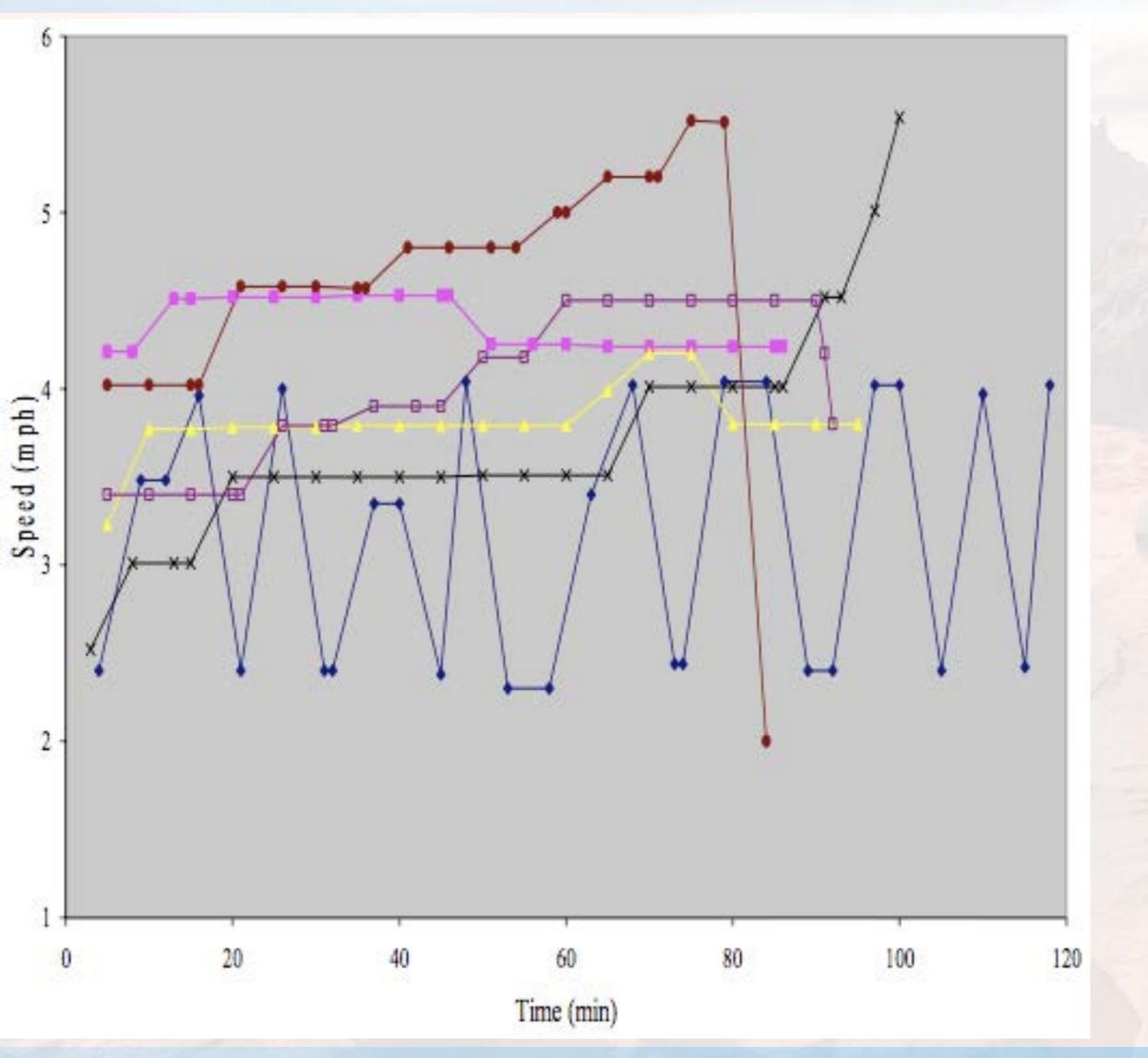




Test Subject Performing 10 km Traverse

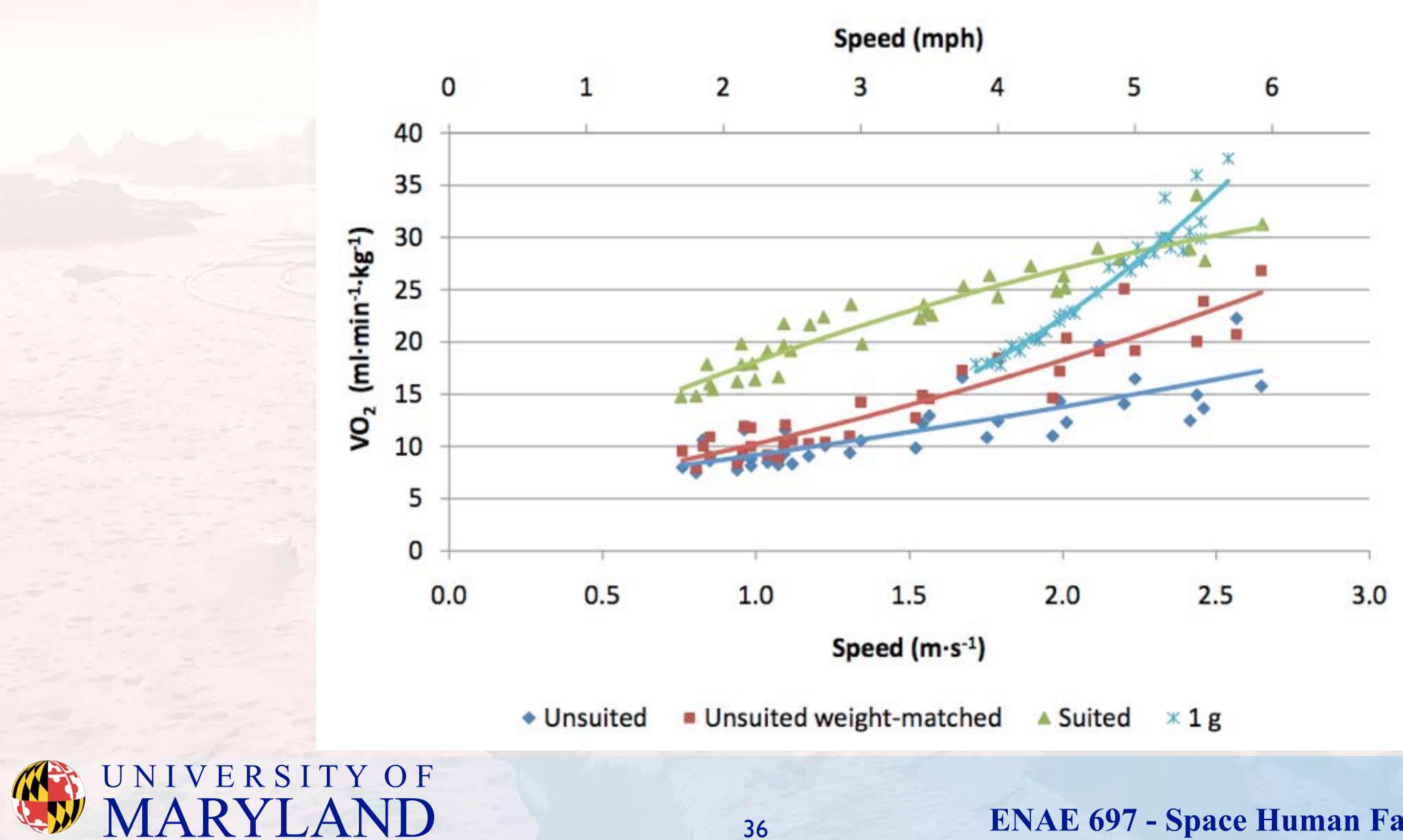






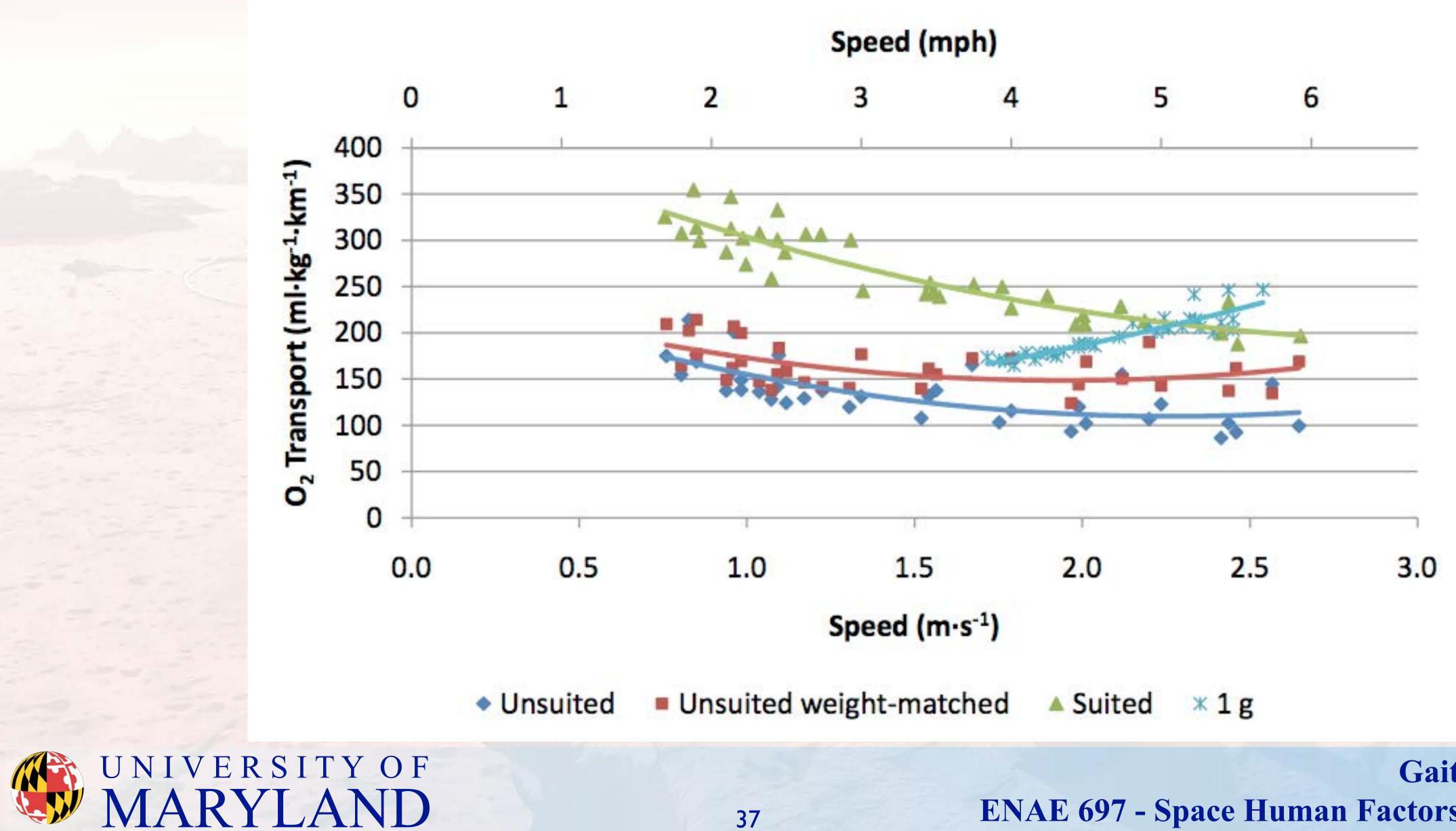


Metabolic Costs of Lunar Locomotion



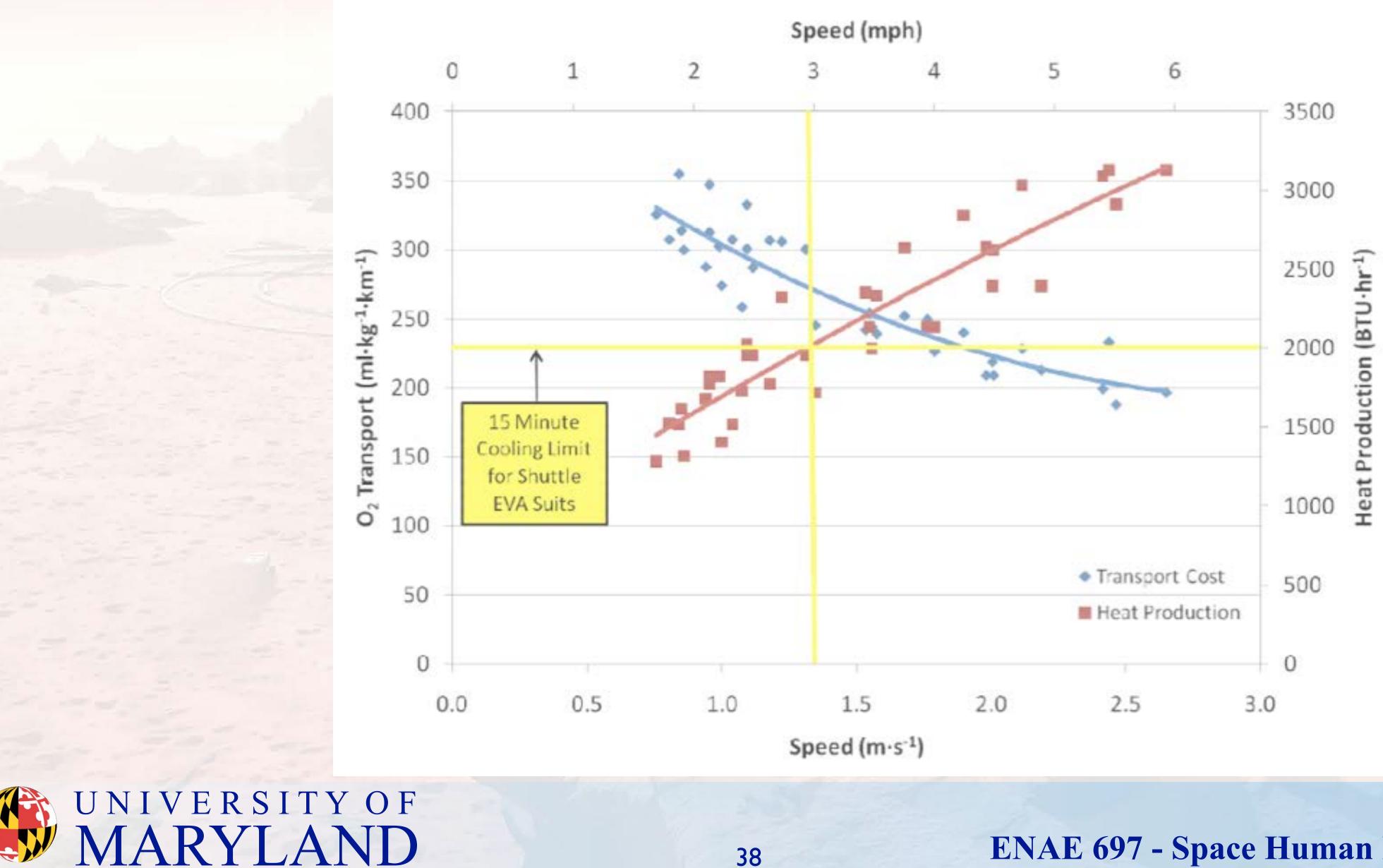


O2 Transport Cost of Lunar Locomotion





Cooling and Energy Use in Lunar Run



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References for This Lecture

- Conference on Environmental Systems, San Francisco, CA, July 2008
- Conference on Environmental Systems, Chicago, IL, July 2007
- Rader, Newman, and Carr, Loping: A Strategy for Reduced Gravity Human Locomotion? ICES



• Gernhardt, Norcross, and Vos, Integrated Suit Test 1 - A Study to Evaluate Effects of Suit Weight, Pressure, and Kinematics on Human Performance During Lunar Ambulation ICES 2008-01-1951, International

• Vos, Gernhardt, and Lee, The Walkback Test: A Study to Evaluate Suit and Life Support System Performance Requirements for a 10 Kilometer Lunar Traverse in a Planetary Suit ICES 2007-01-3133, International

2007-01-3134, International Conference on Environmental Systems, Chicago, IL, July 2007

