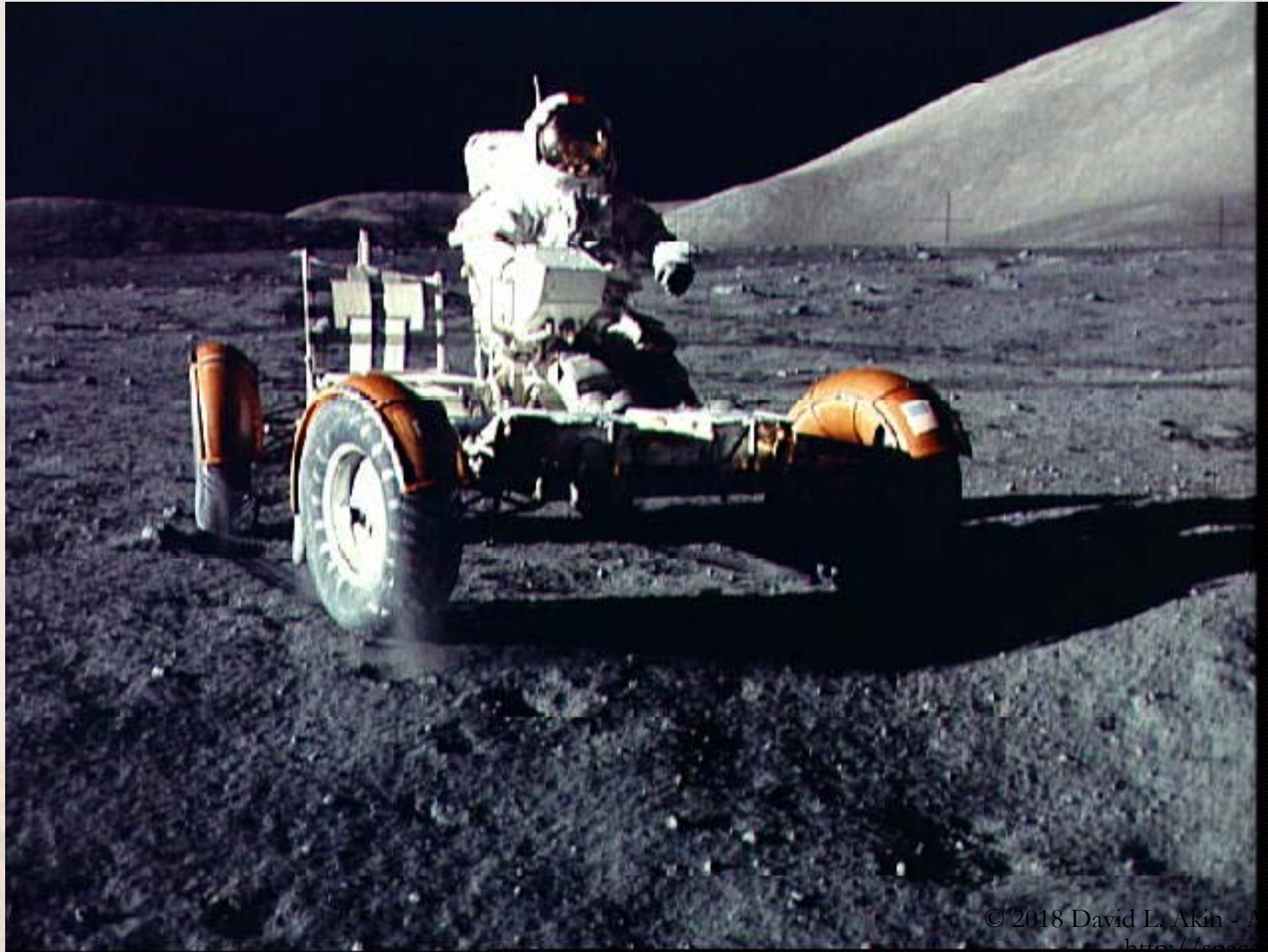


# Case Study: Lunar Roving Vehicle



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Case Study – Lunar Roving Vehicle  
ENAE 788X - Planetary Surface Robotics

# Concepts for Lunar Equipment Carriers

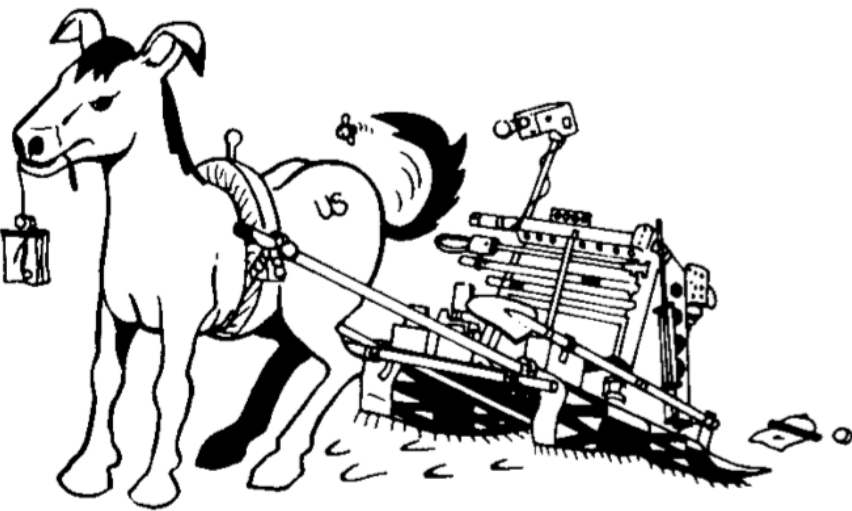


Figure 1. - The travois.

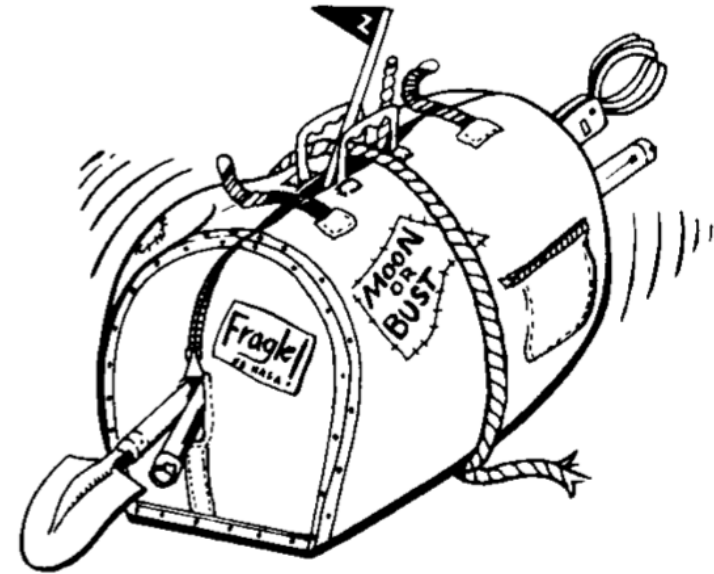


Figure 2. - The suitcase.



# Concepts for Lunar Equipment Carriers

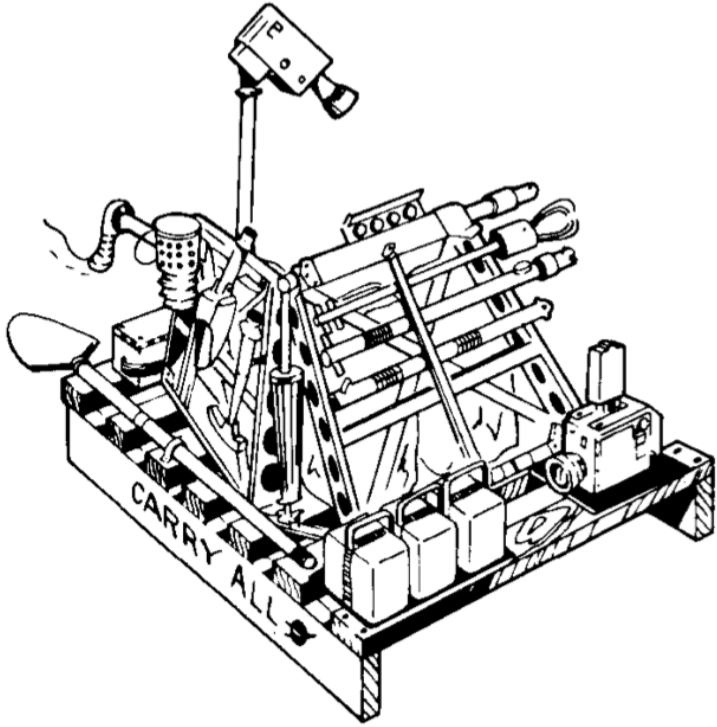


Figure 3. - The pallet.

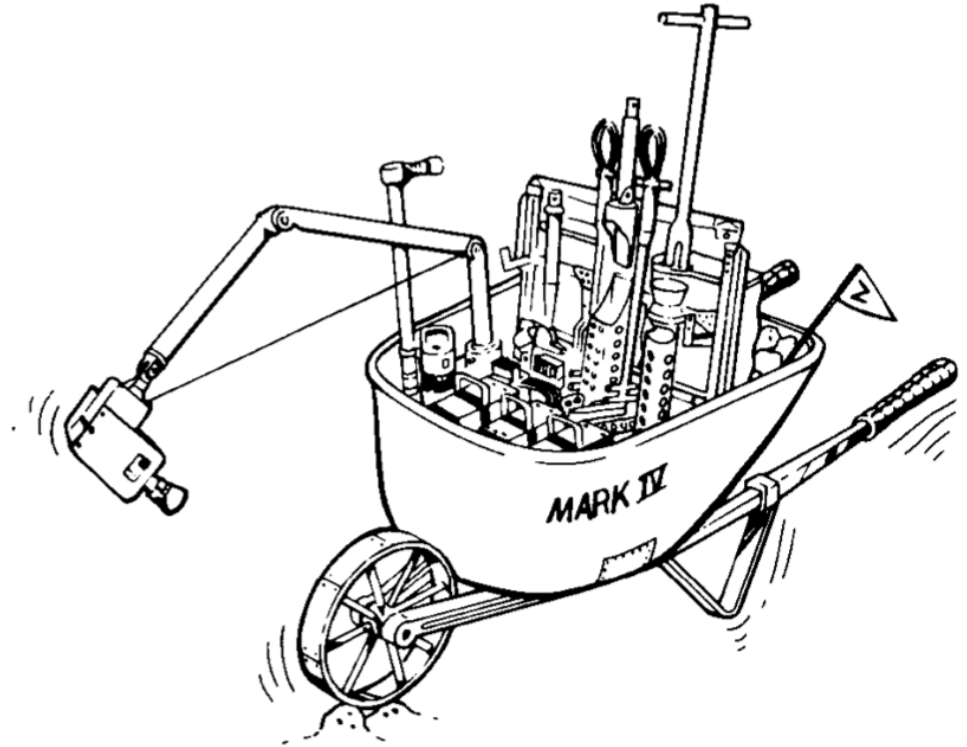


Figure 4. - The wheelbarrow.



# Concepts for Lunar Equipment Carriers

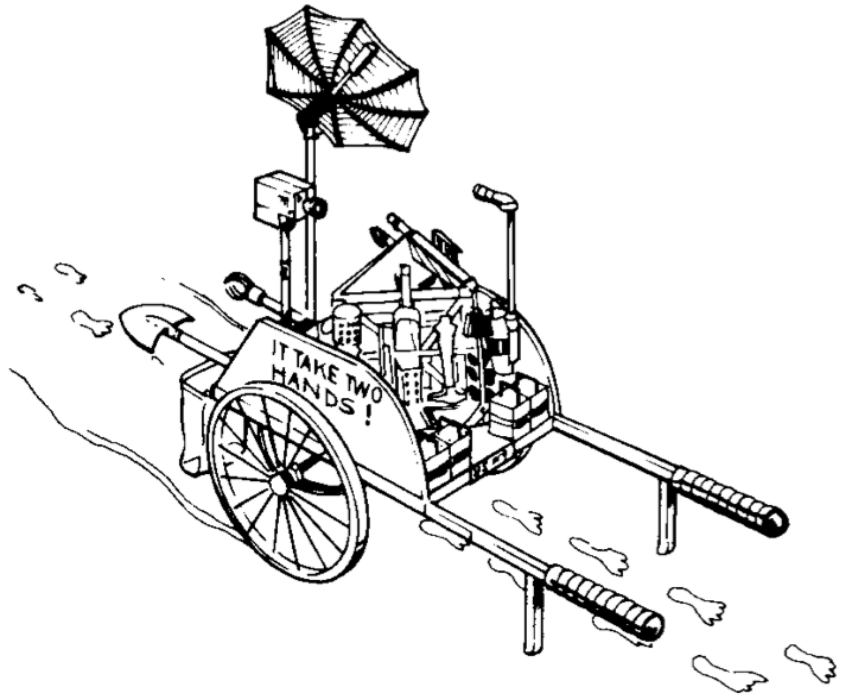


Figure 5. - The cart.

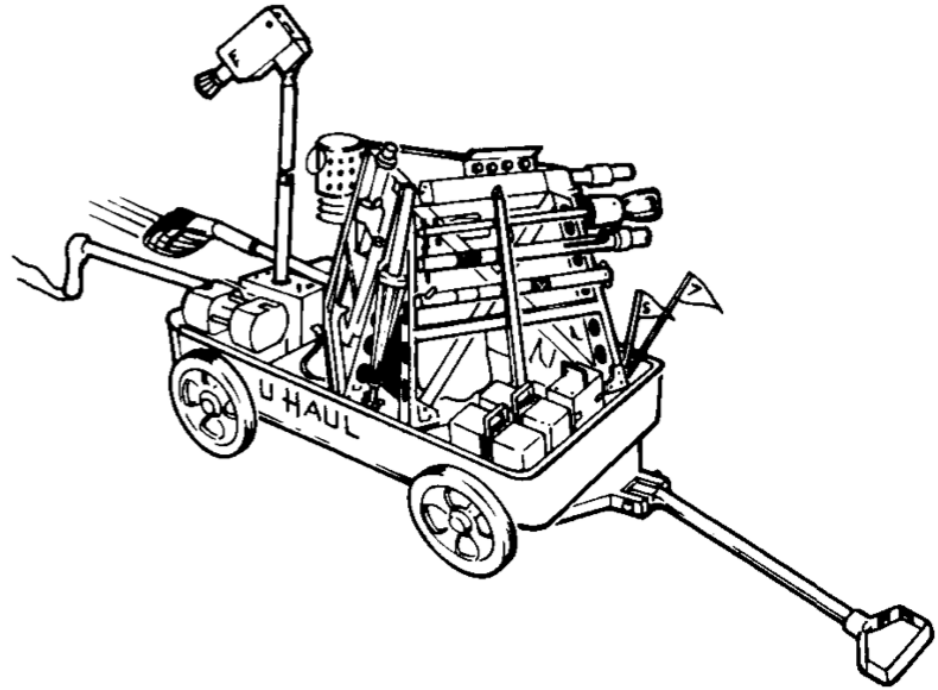
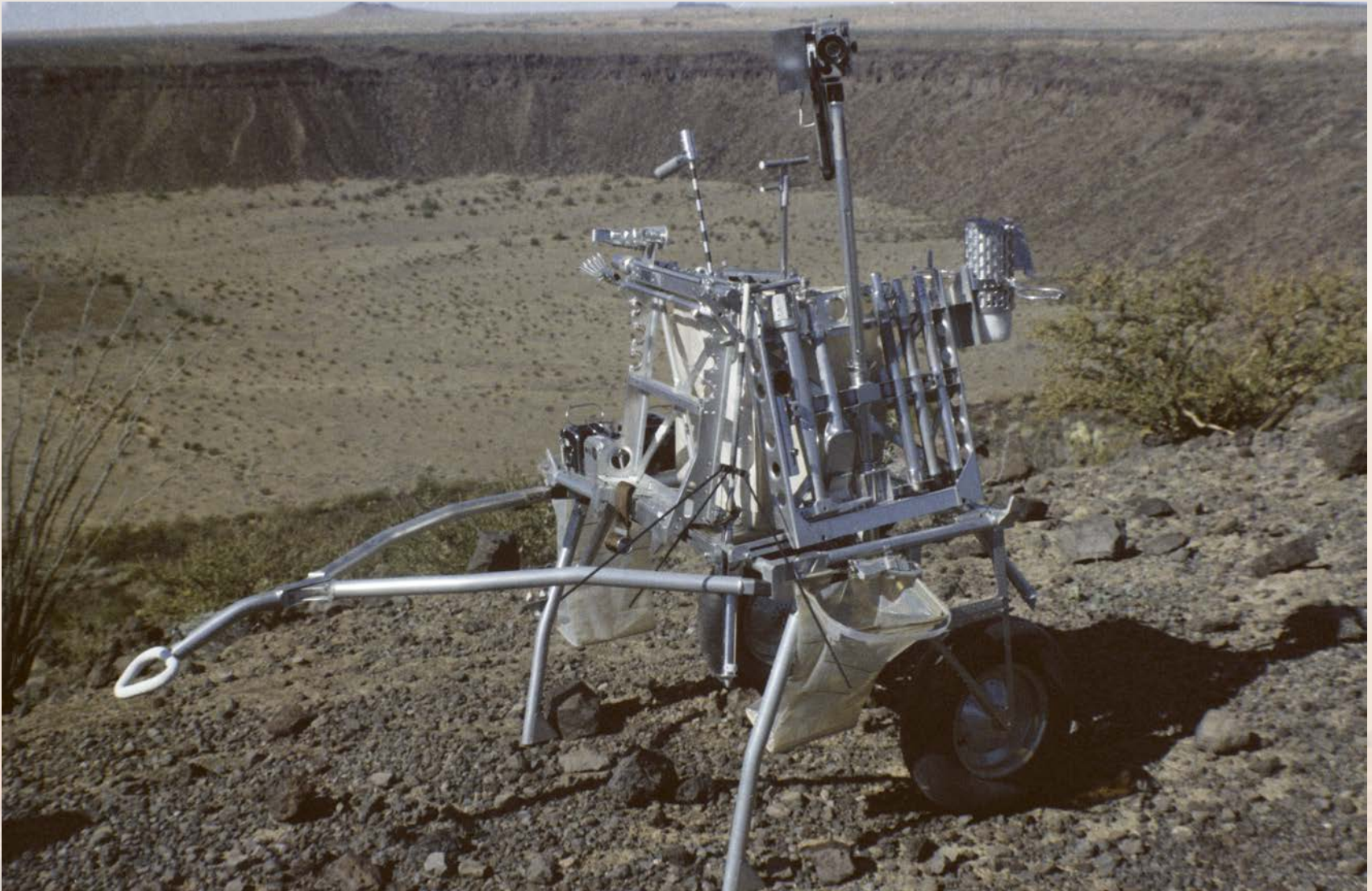


Figure 6. - The wagon.

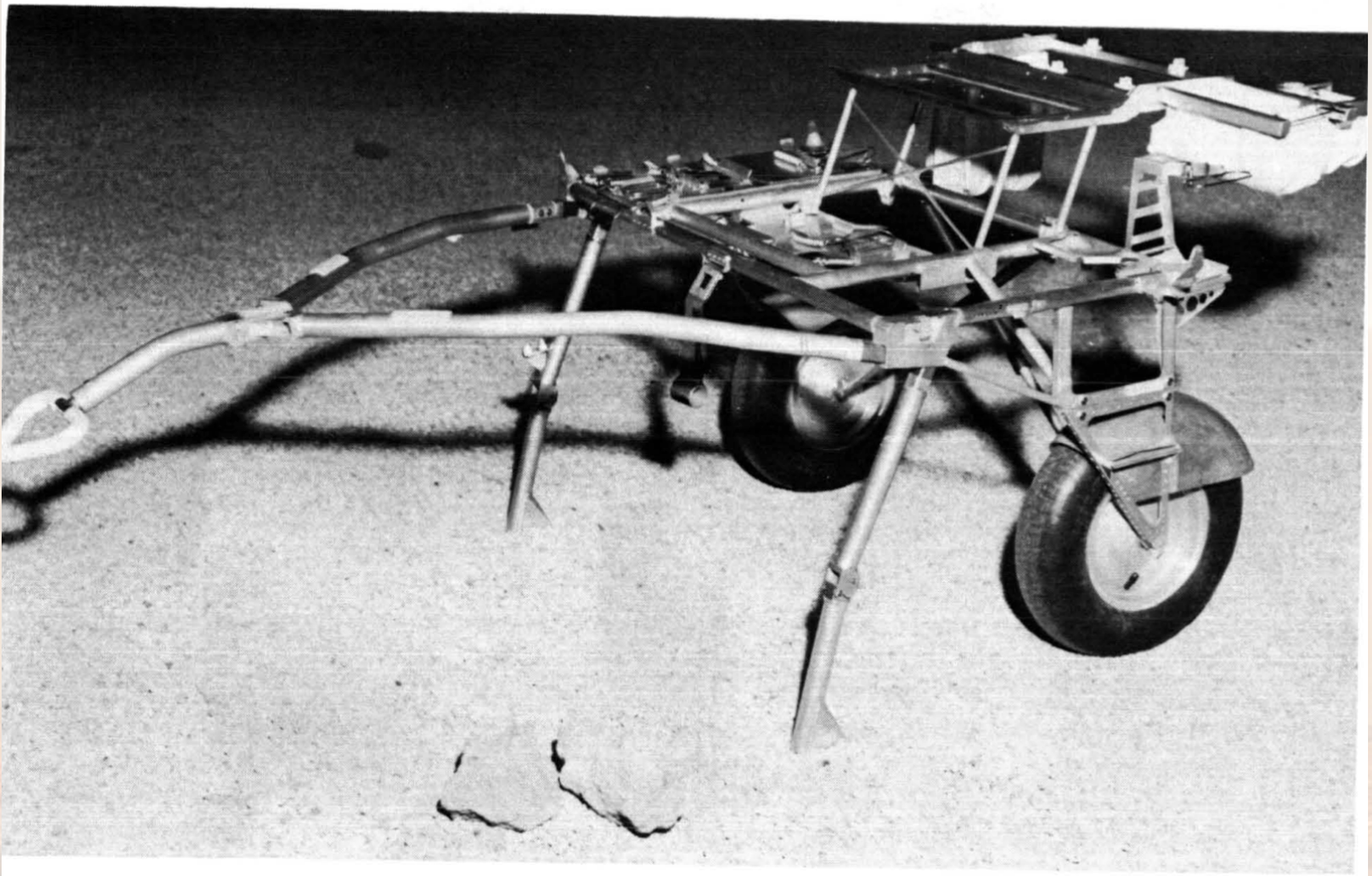


# Modular Equipment Transporter



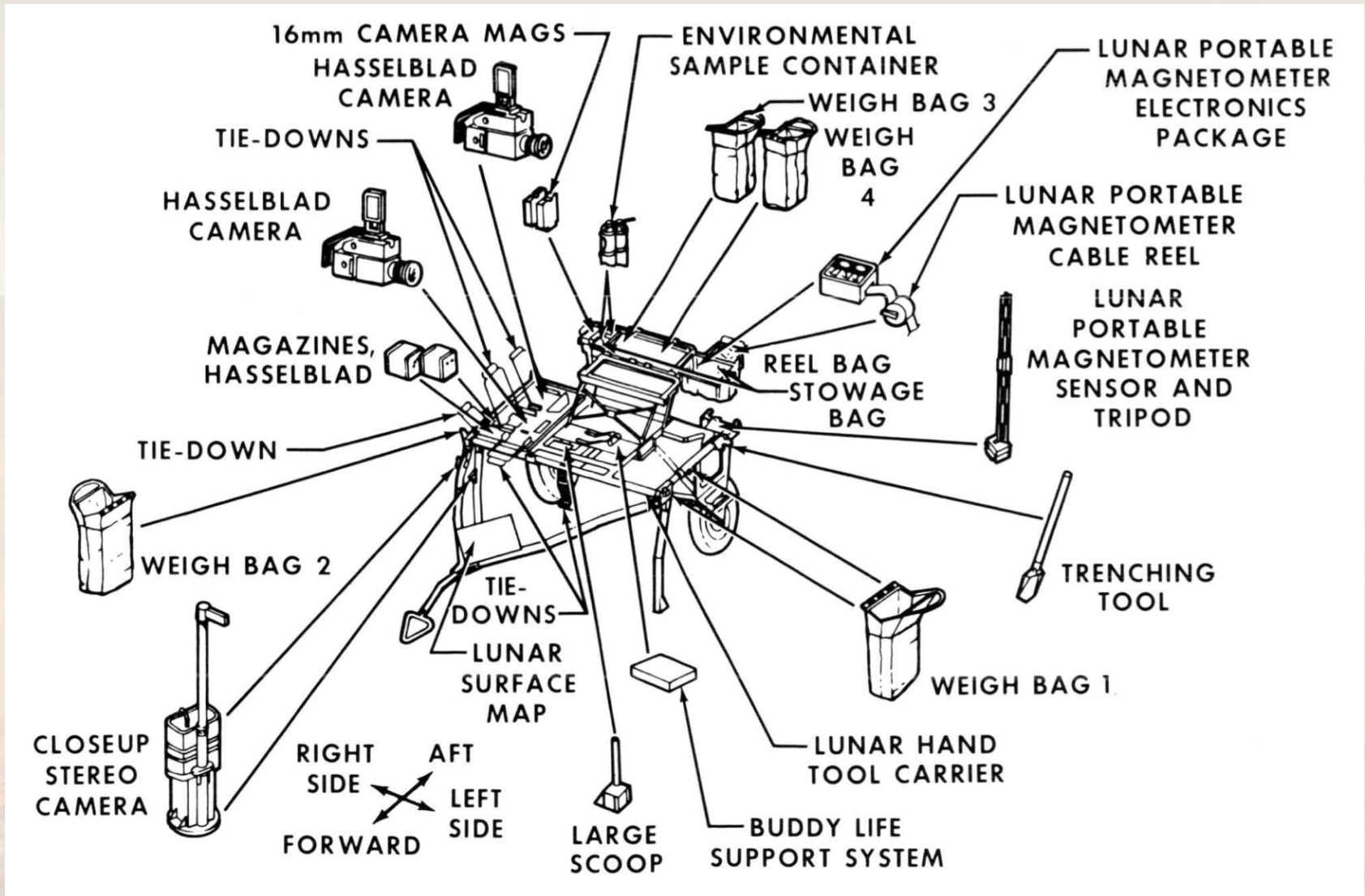
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# MET Structure



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# MET Equipment Load (Apollo 14)



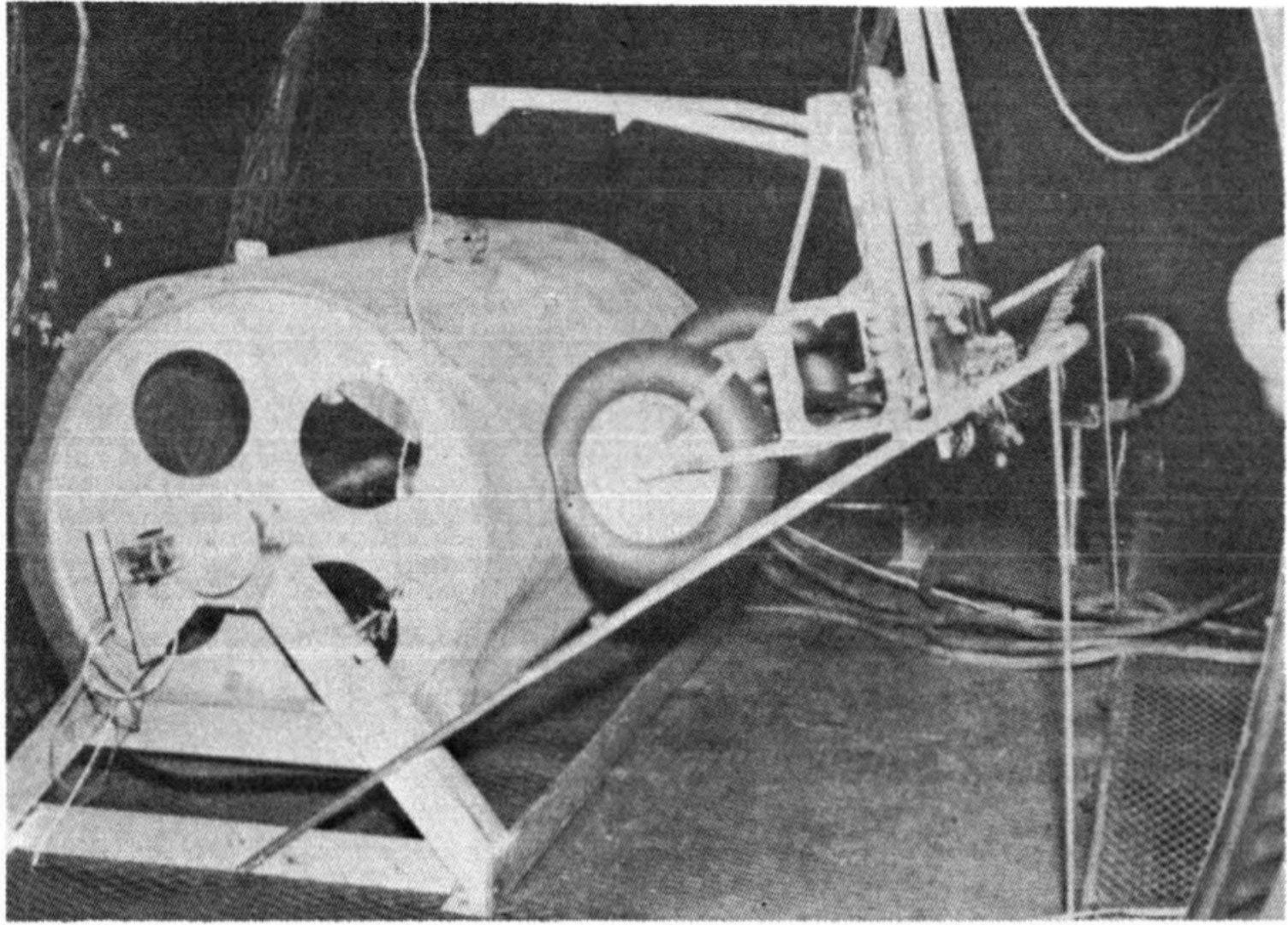
# Pneumatic Tire Specifications

Color . . . . .	Black
Size, width by height, cm (in.) . . . . .	10.16 by 40.64 (4 by 16)
Inflation pressure, N/m <sup>2</sup> (psia) . . . . .	10.34 × 10 <sup>3</sup> to 20.68 × 10 <sup>3</sup> (1.5 to 3)
Deflection under load, percent . . . . .	30
Allowable pressure loss:	
6 weeks in 101.34 × 10 <sup>3</sup> N/m <sup>2</sup> (14.7 psi) ambient and 2 weeks in vacuum, N/m <sup>2</sup> (psia) . . . . .	0.69 (0.1)
Abrasion and wear, meters (feet) of travel over simulated lunar surface . . . . .	6096 (20 000)
Outgassing, percent weight loss after baking in a vacuum chamber for 72 hr at 394.26° K (250° F) . . . . .	>4.3
Operating temperature environment, °K (°F) . . . . .	208.15 to 394.26 (-85 to 250)



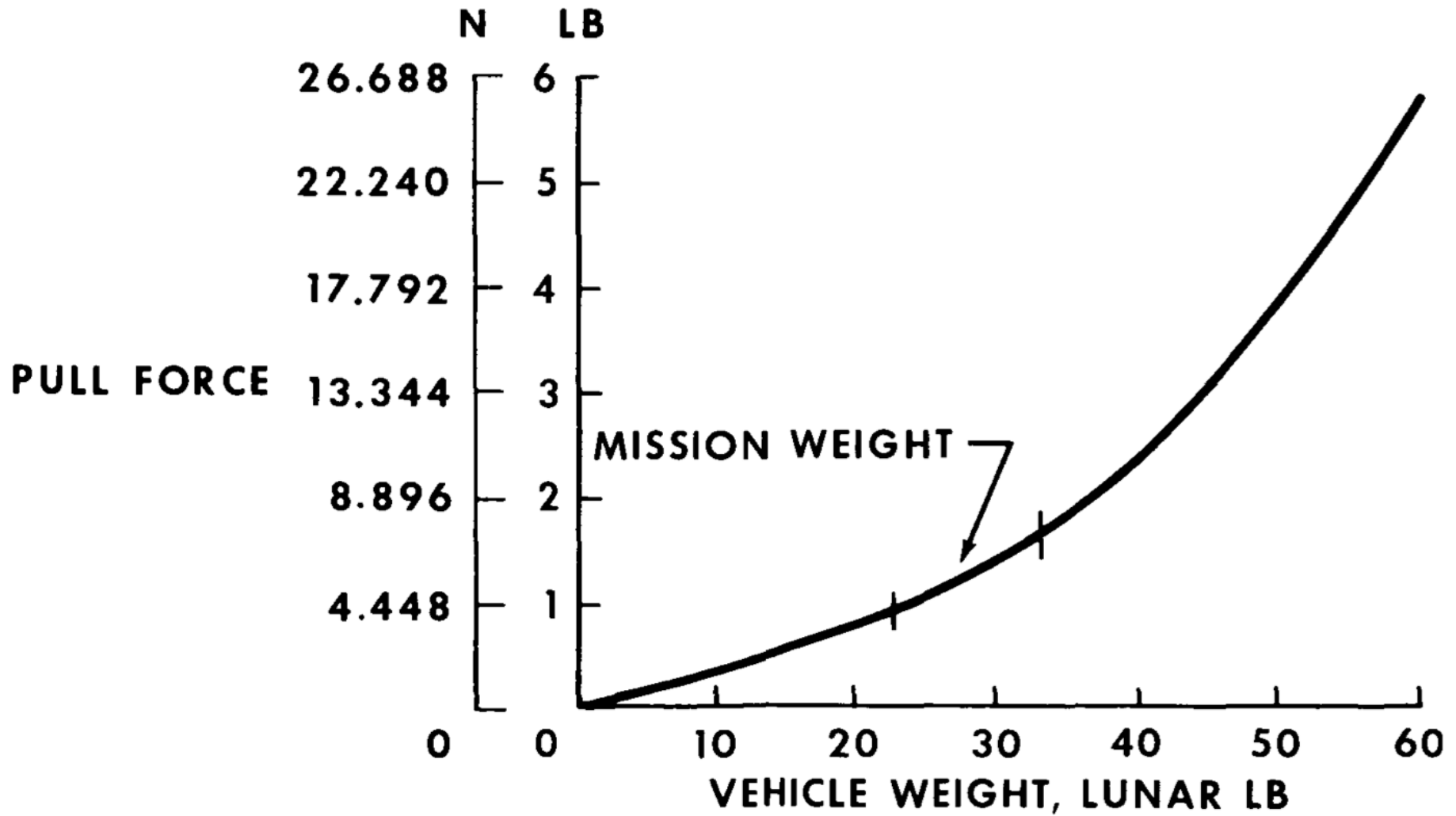


# Thermal Vacuum Testing of Wheels



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# Towing Force vs. Vehicle Weight (Moon)



# Designing the Lunar Rover



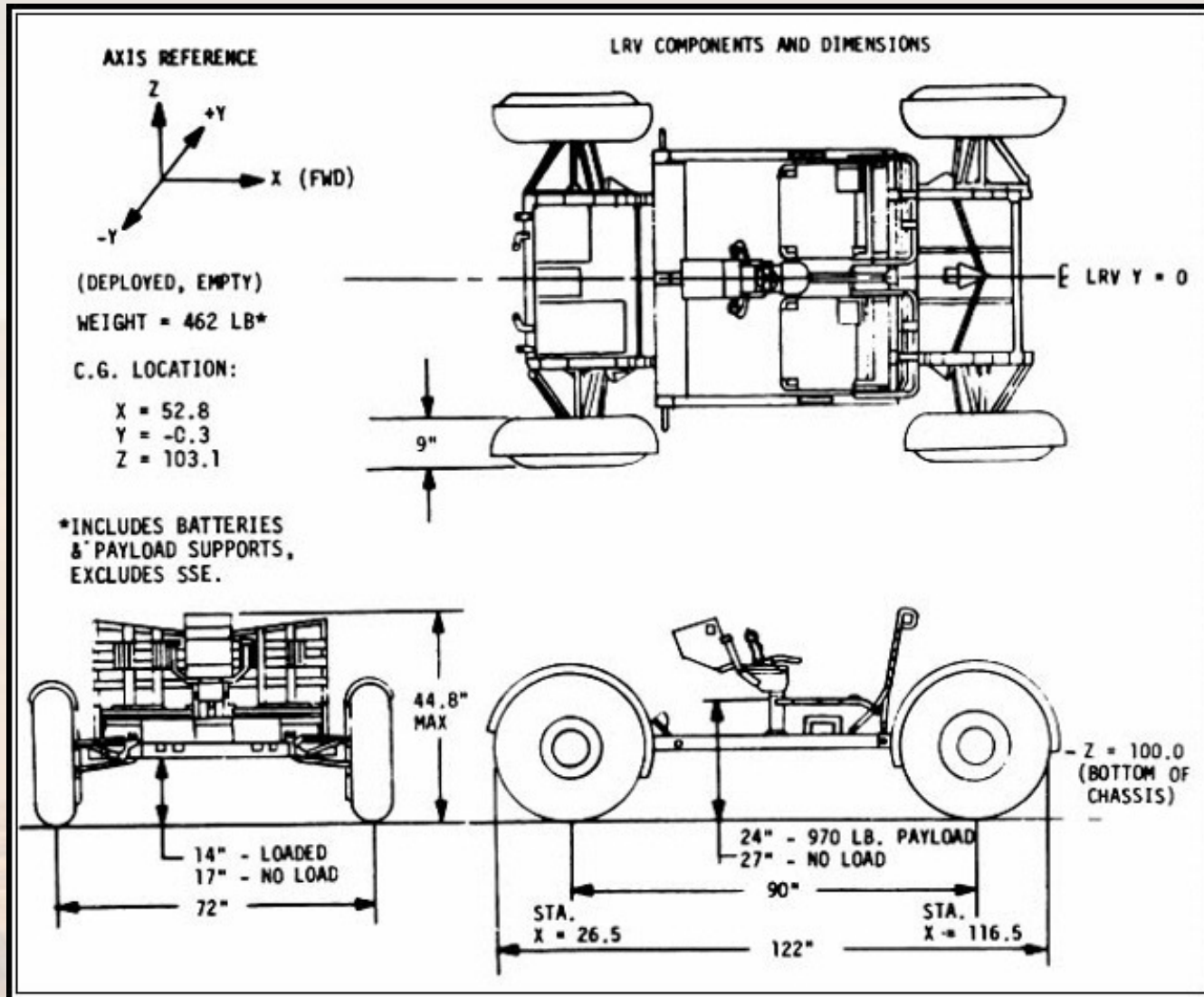
Excerpts from “Lunar Rover” from A&E series “Moon Machines” (available on YouTube at <https://www.youtube.com/watch?v=5aDSYTMqyQw> )



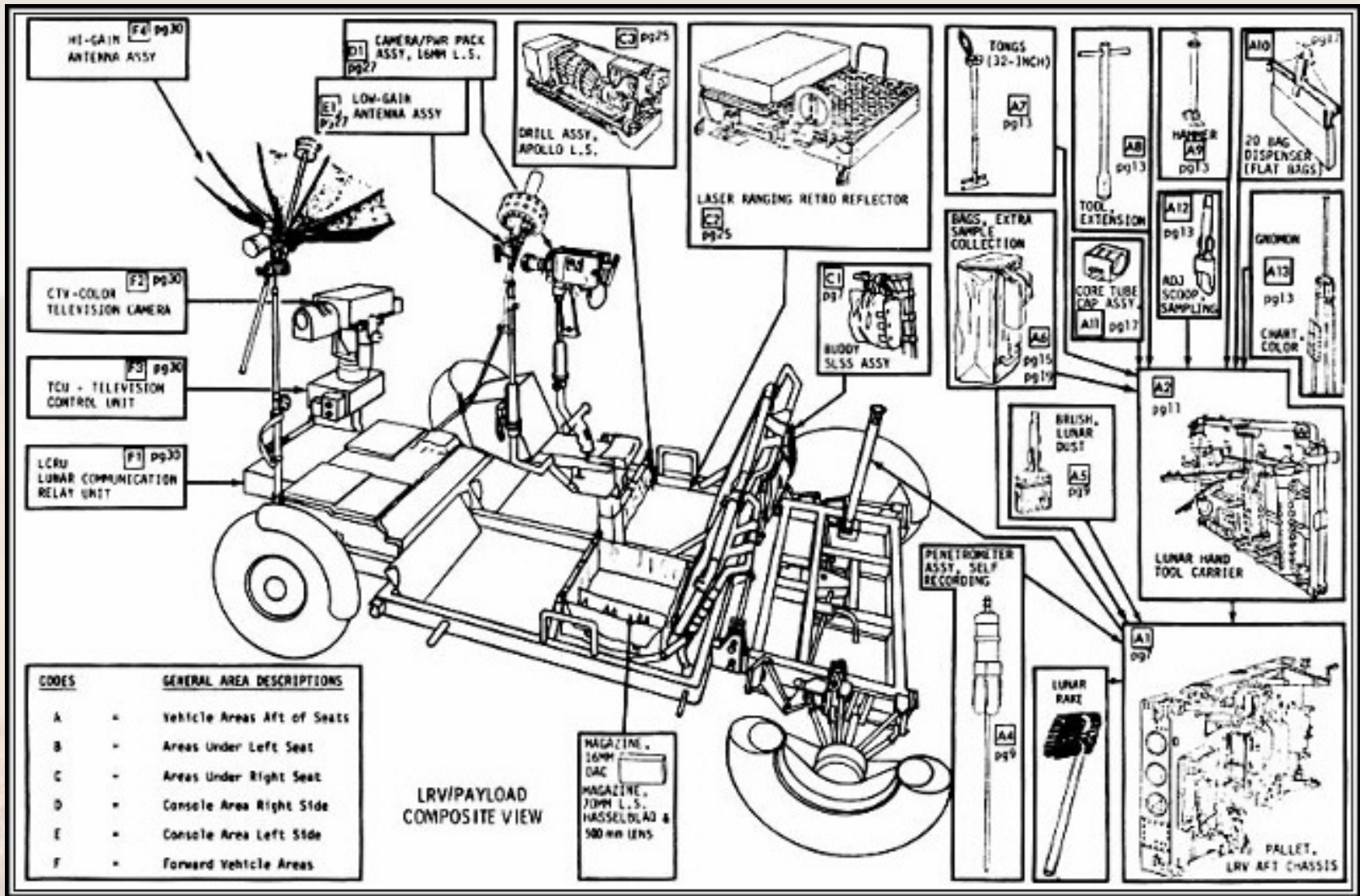
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**Case Study – Lunar Roving Vehicle**  
**ENAE 788X - Planetary Surface Robotics**

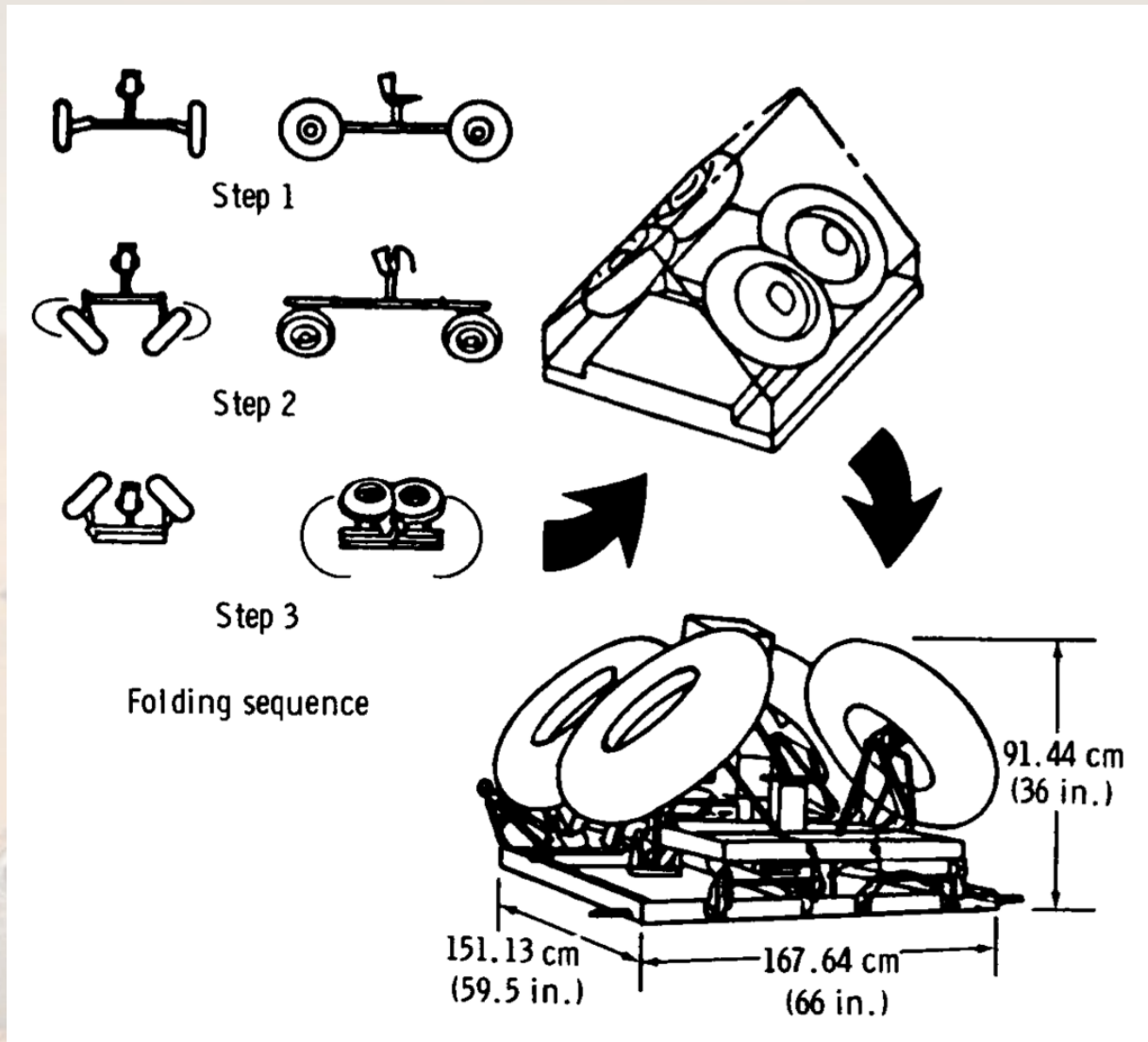
# LRV Three-View and Dimensions



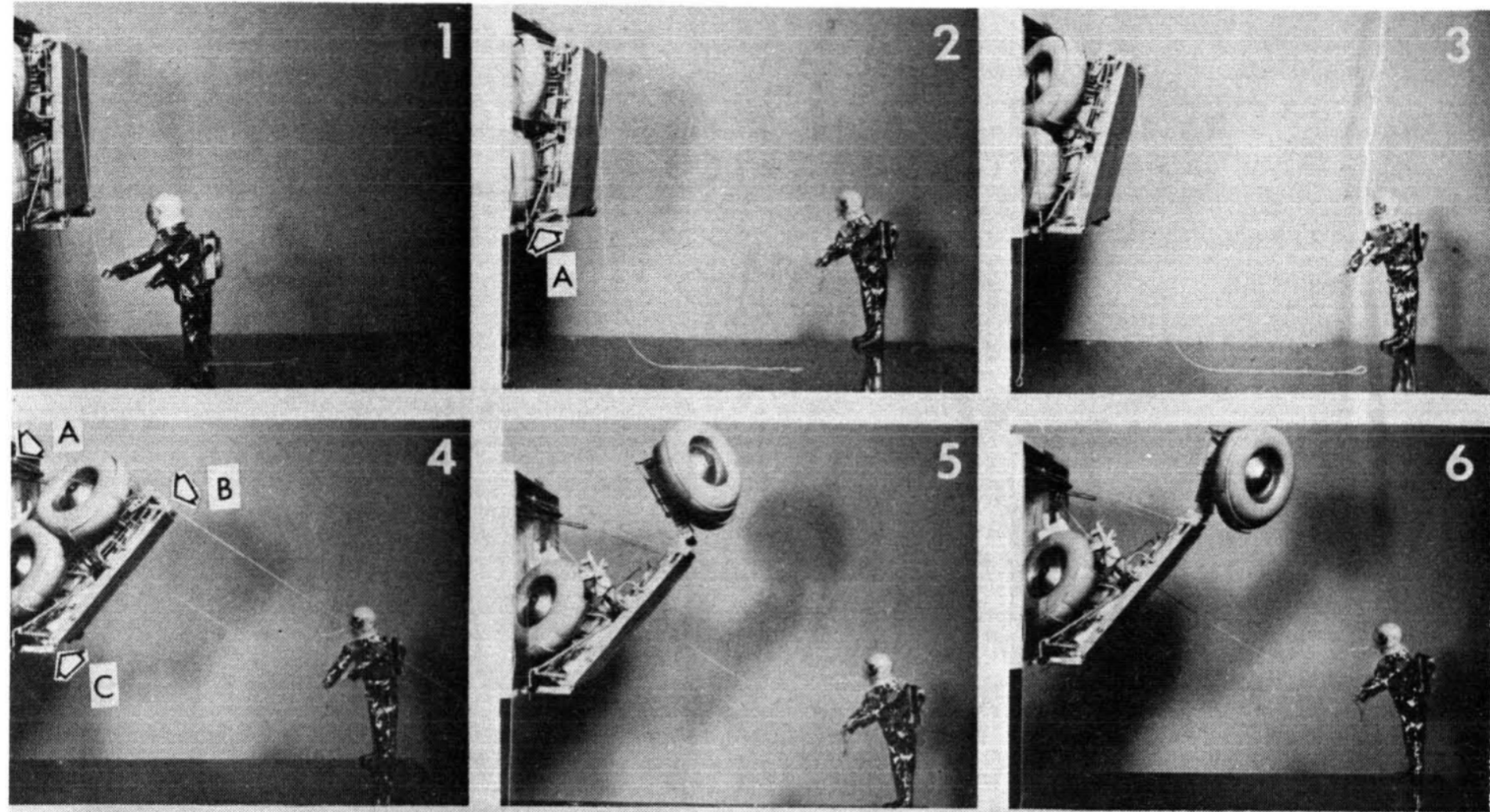
# LRV Payload Components



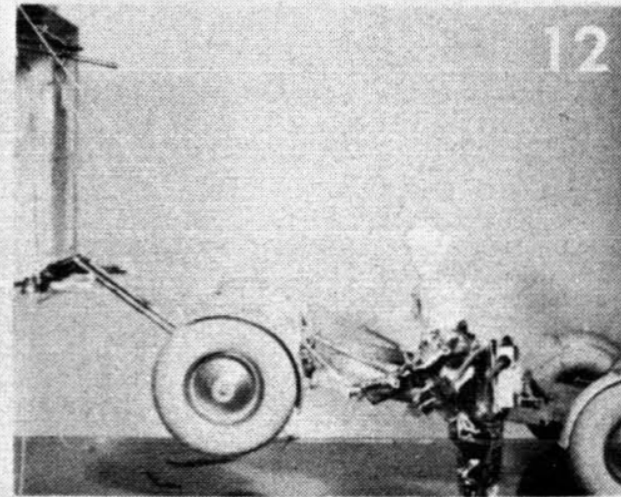
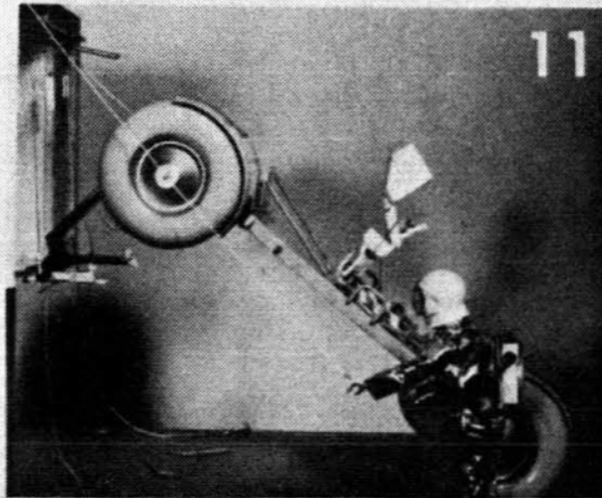
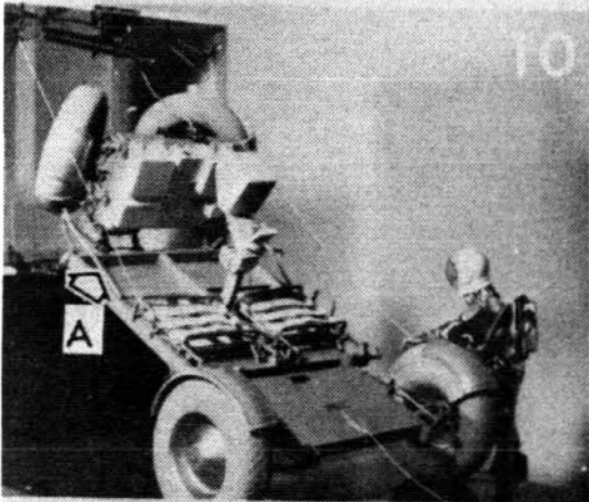
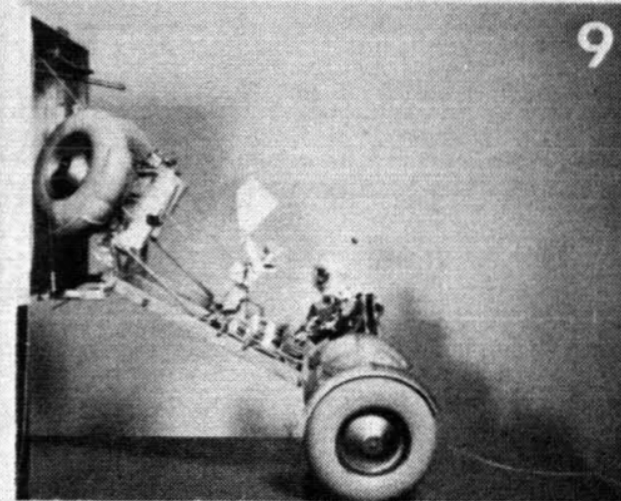
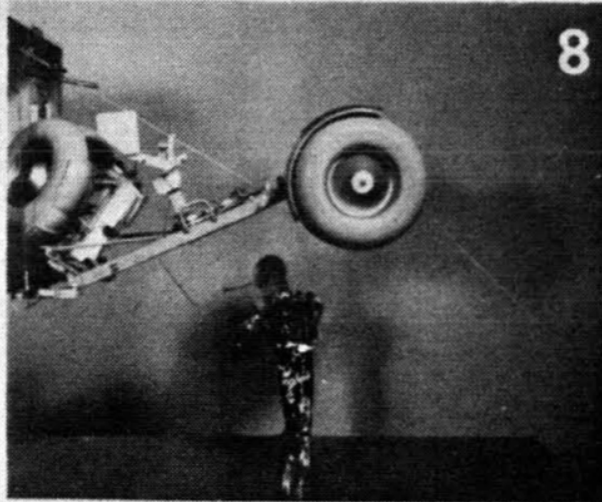
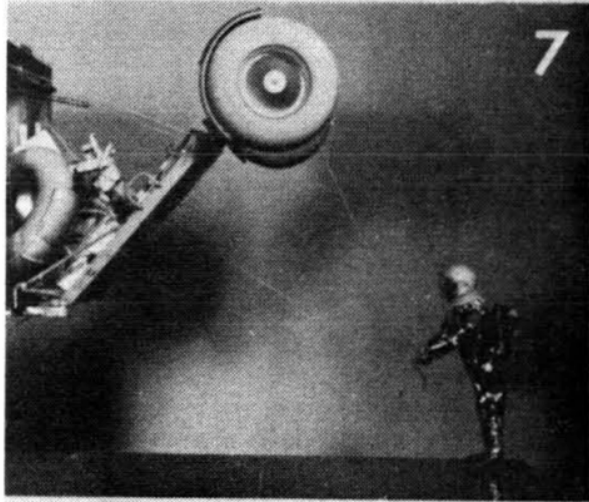
# LRV in Stowed Configuration



# LRV Deployment (1-6)

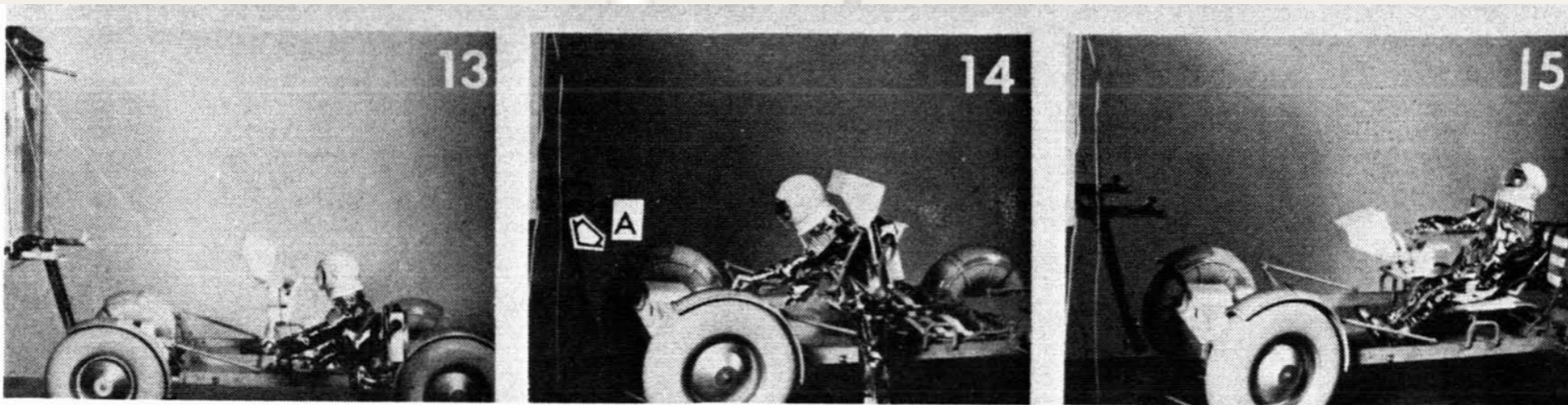


# LRV Deployment Sequence (7-12)

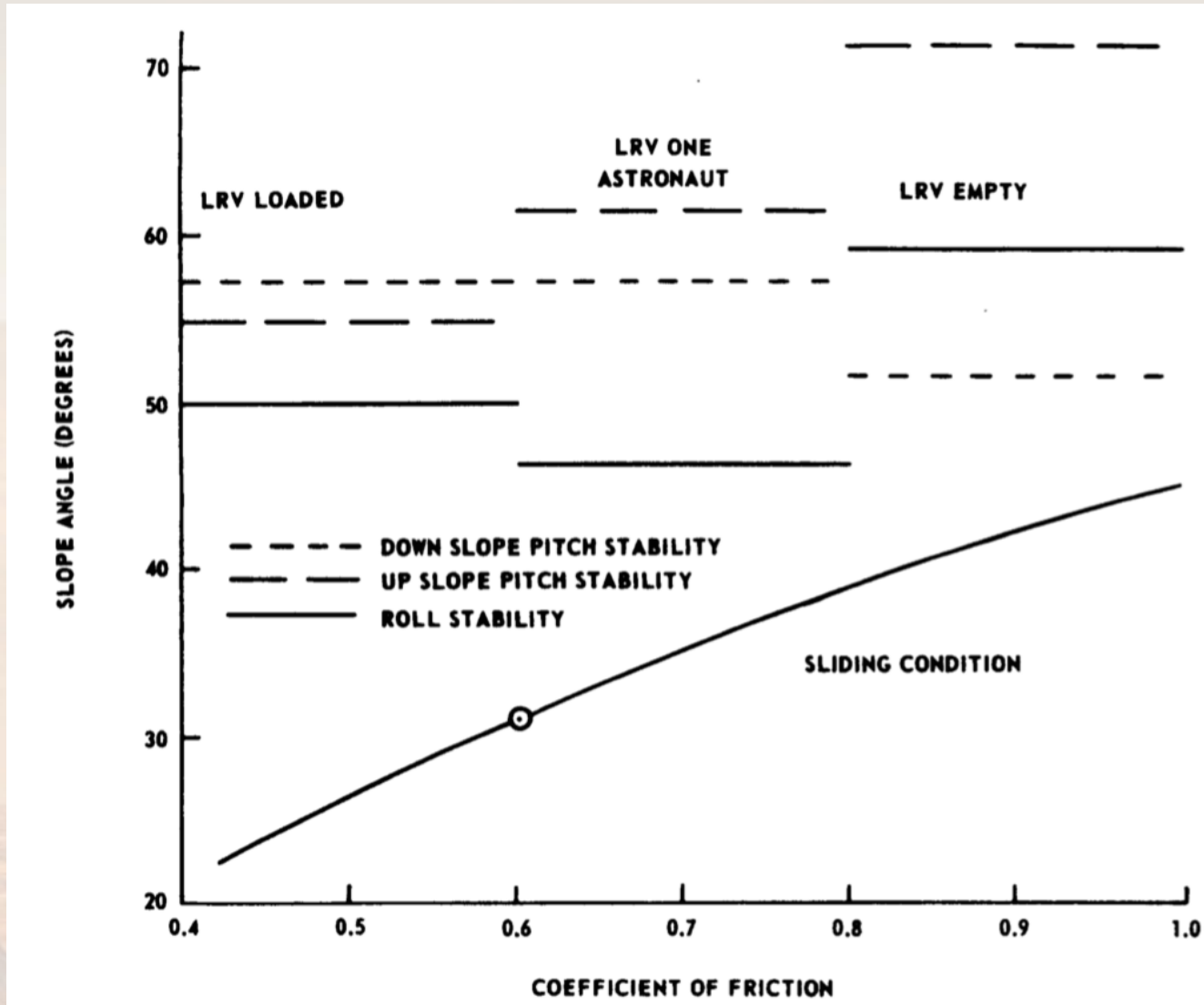




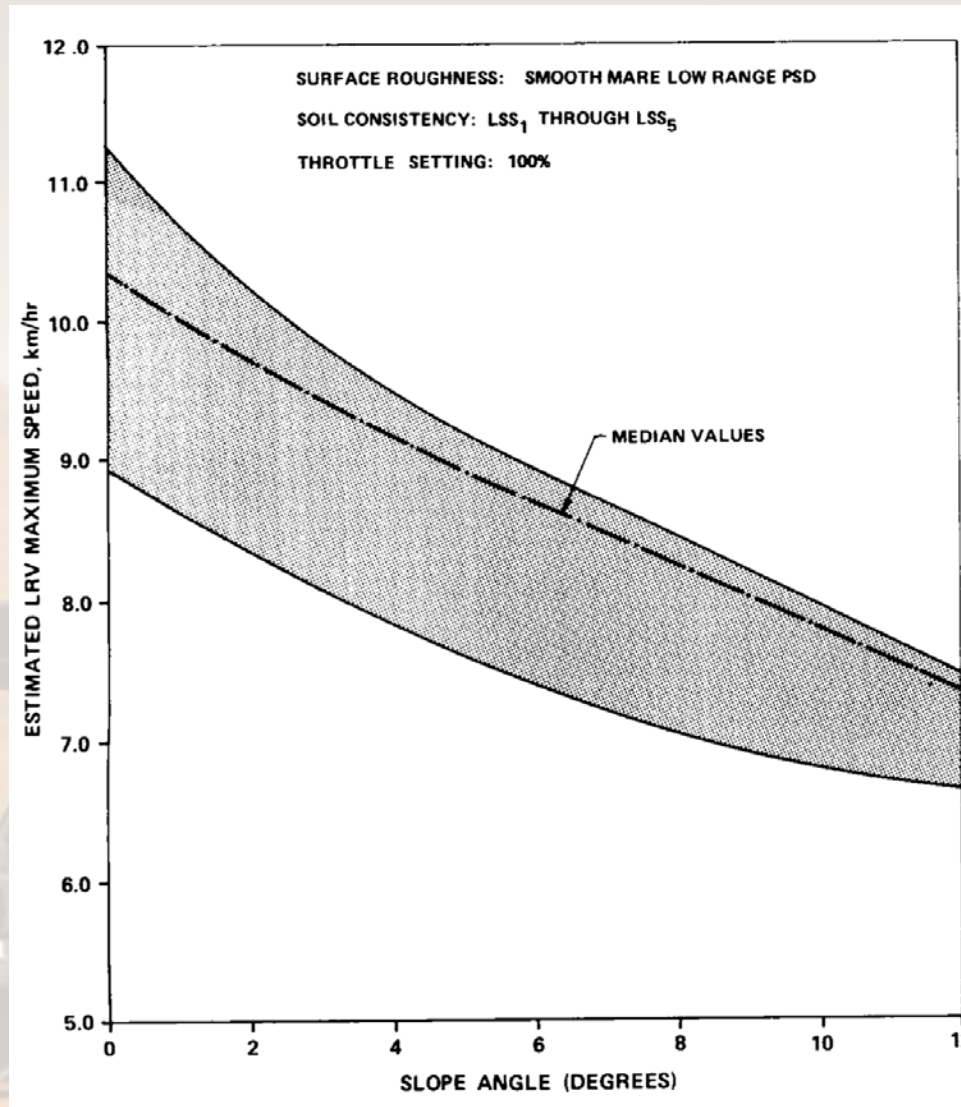
# LRV Deployment Sequence (13-15)



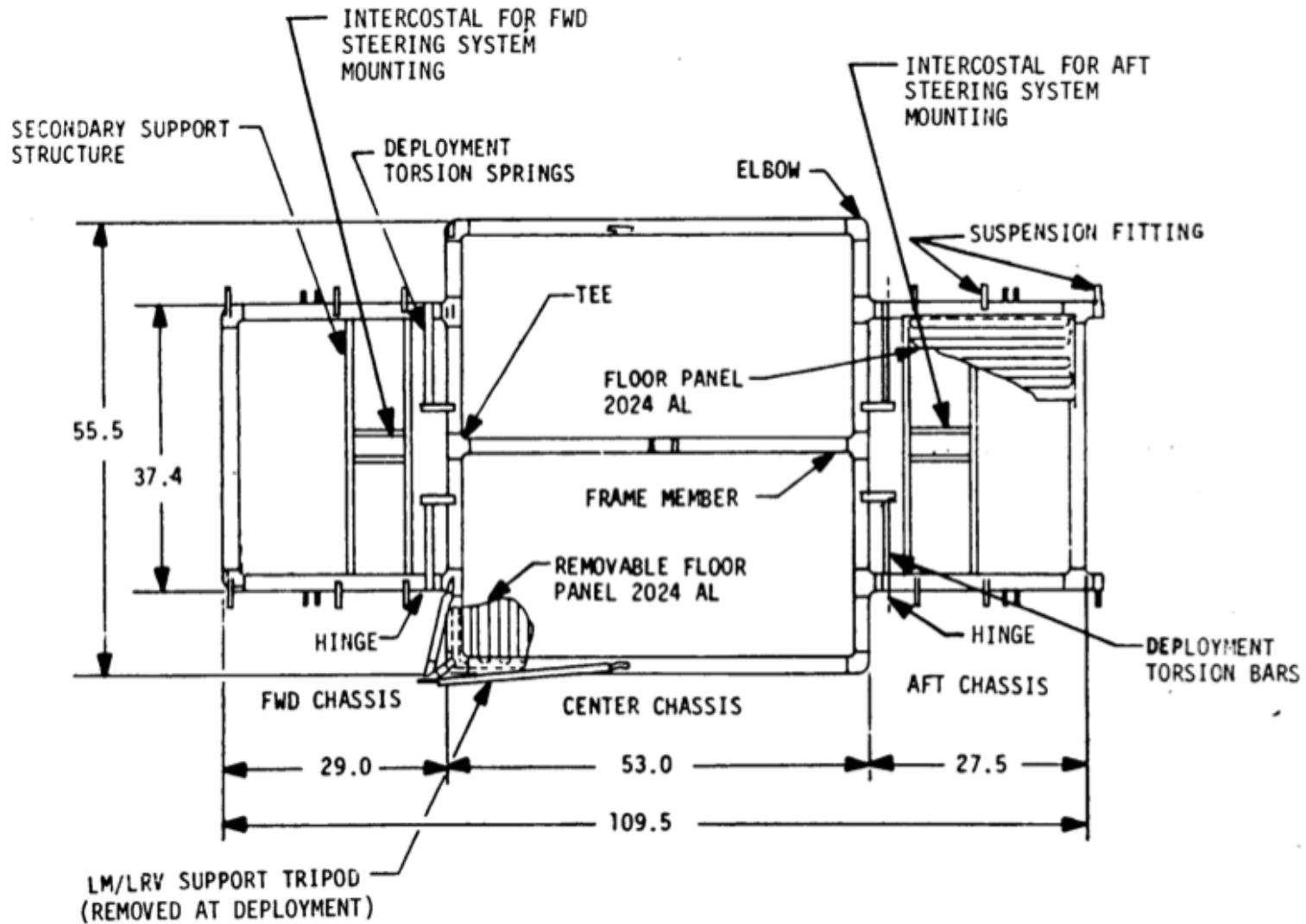
# LRV Static Stability



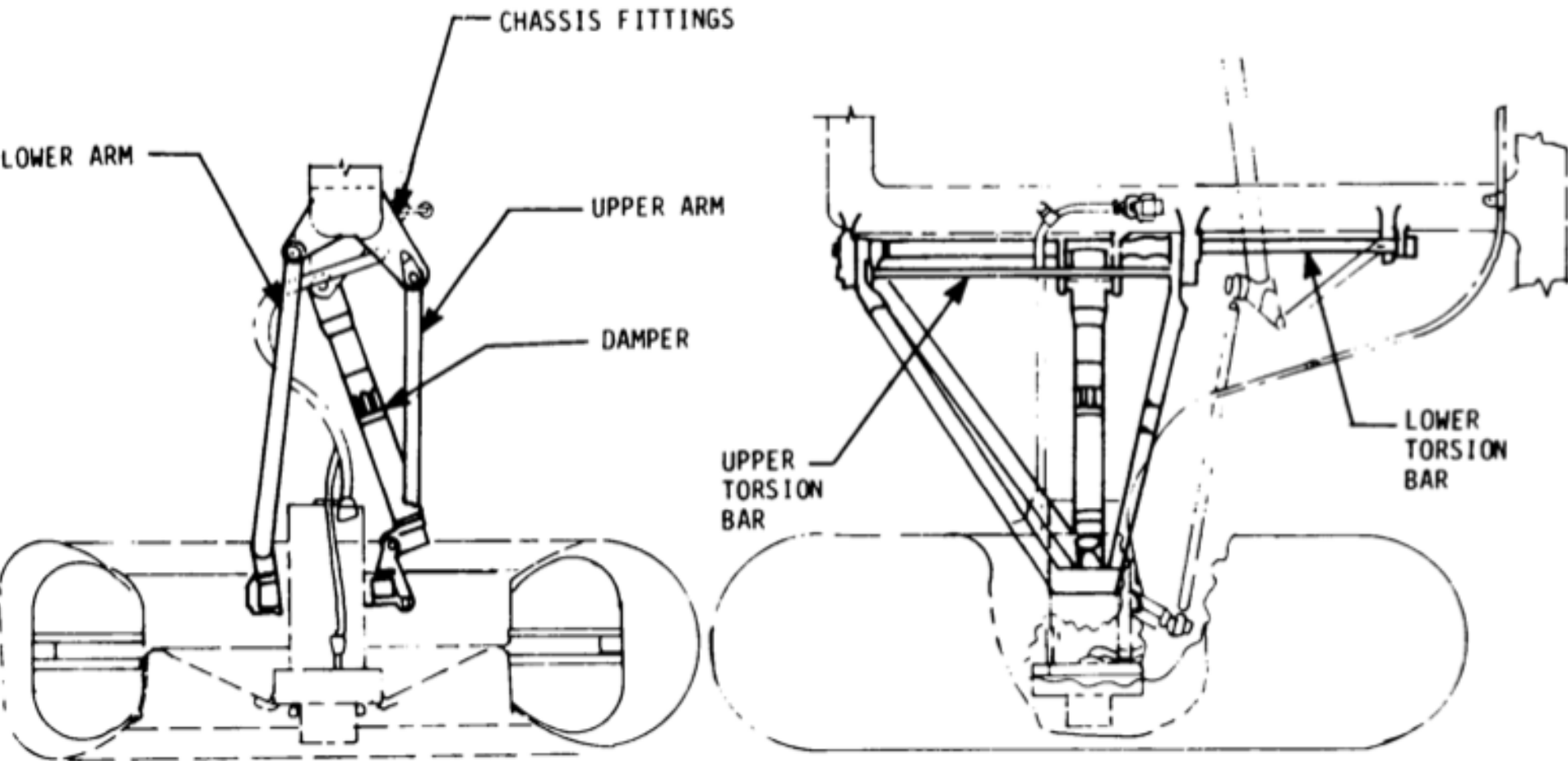
# LRV Speed Capabilities



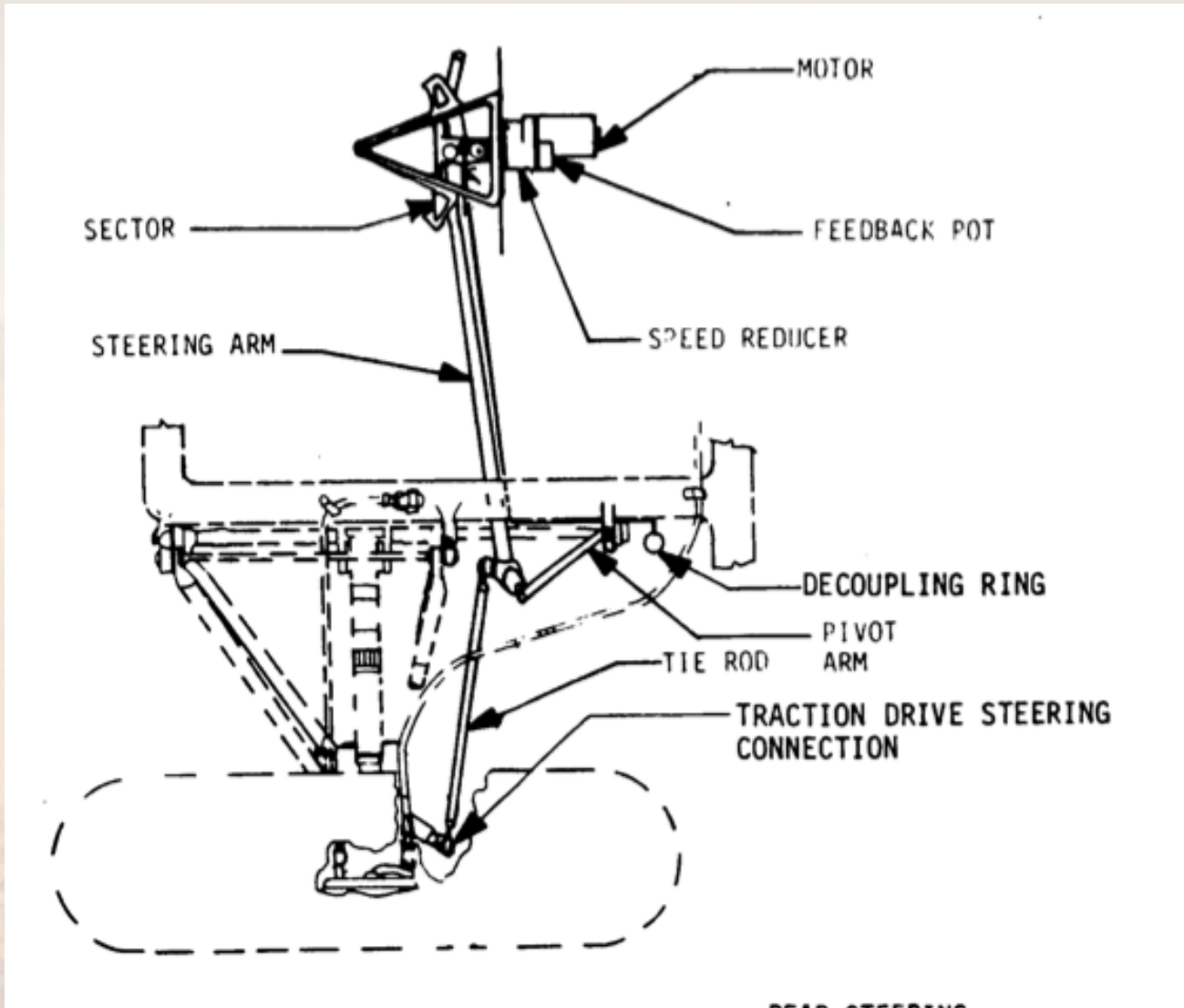
# LRV Chassis Structure



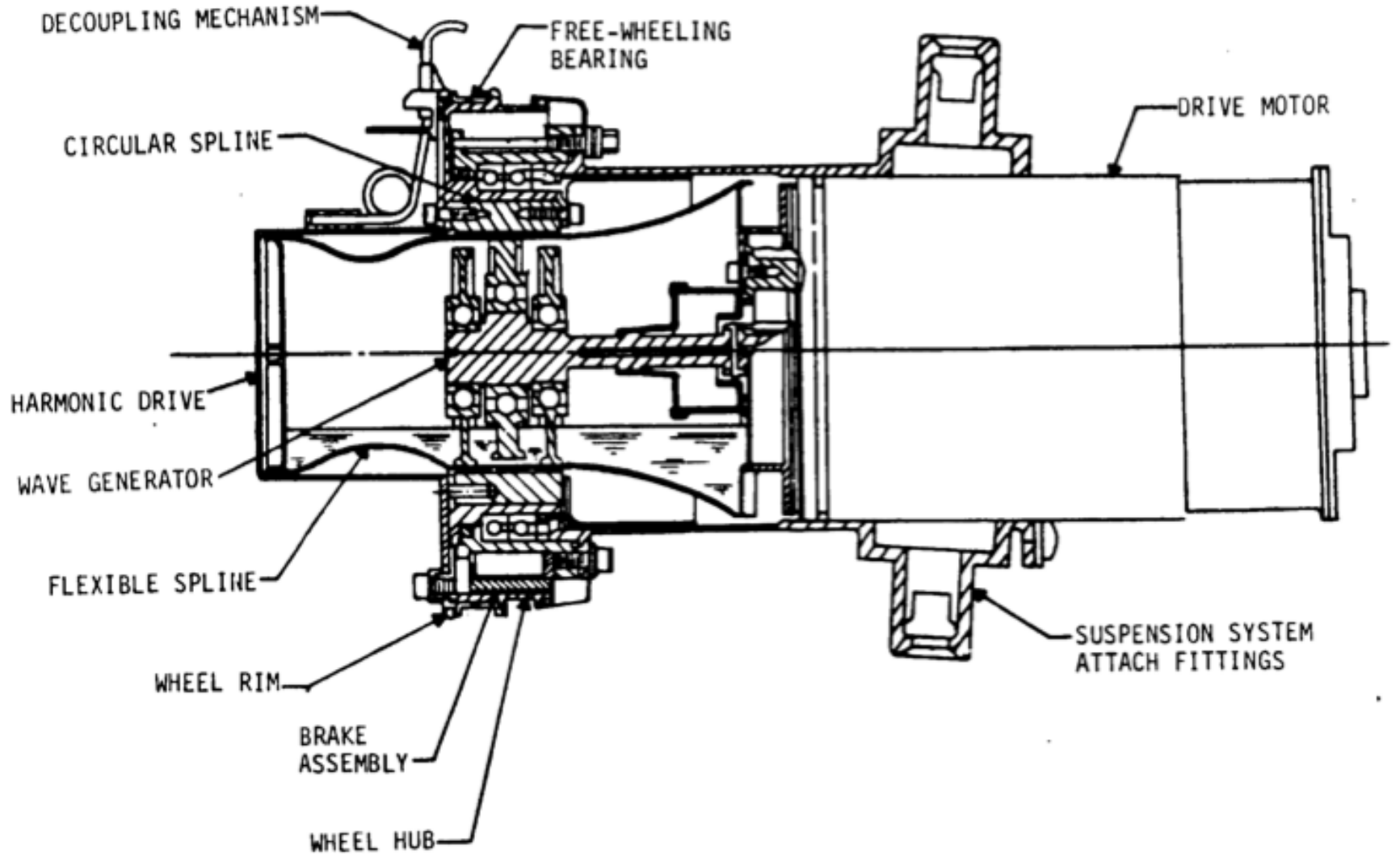
# LRV Wheel Suspension



# LRV Wheel Steering Connections



# LRV Wheel Motor and Gearing



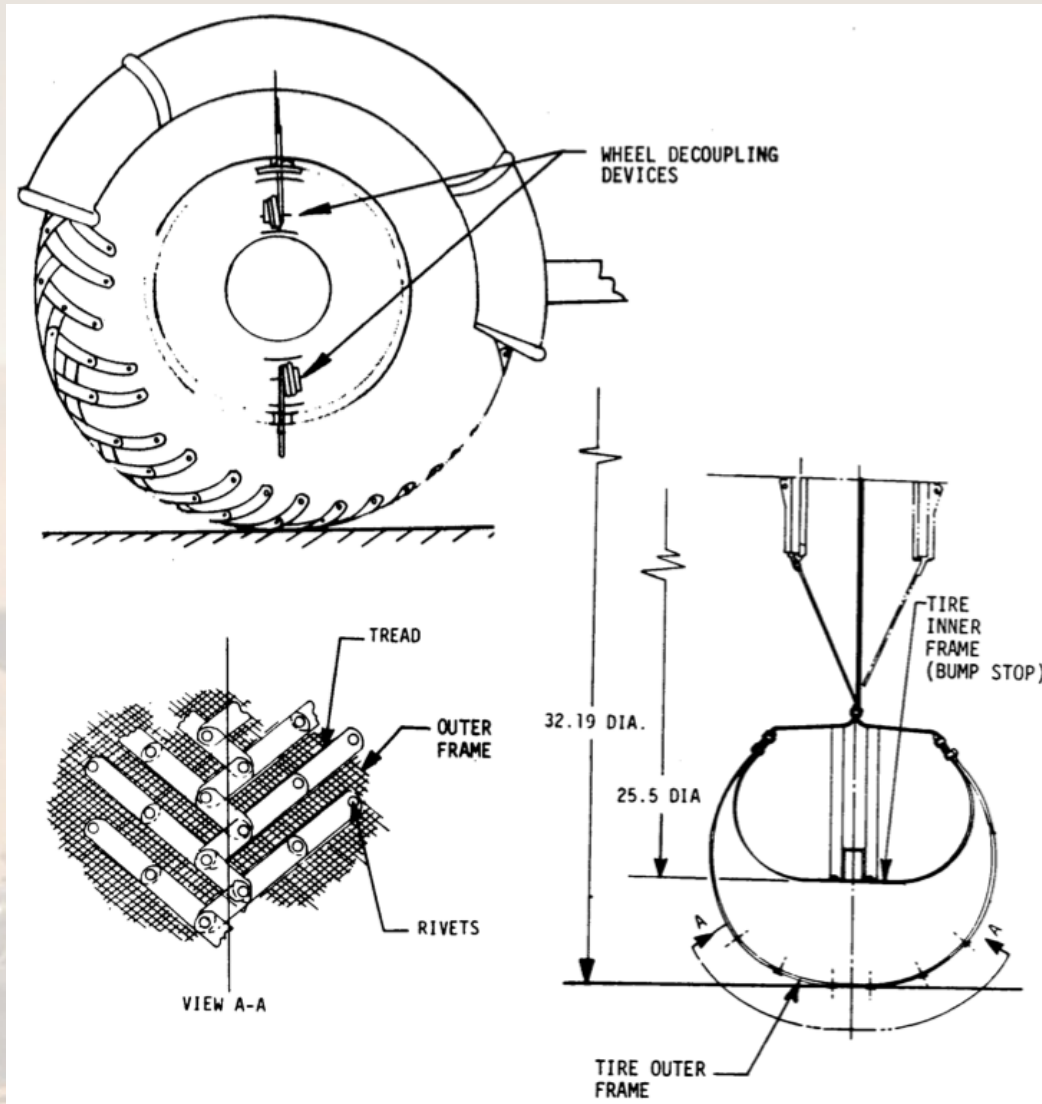
# Apollo 17 LRV Wheel and Fender



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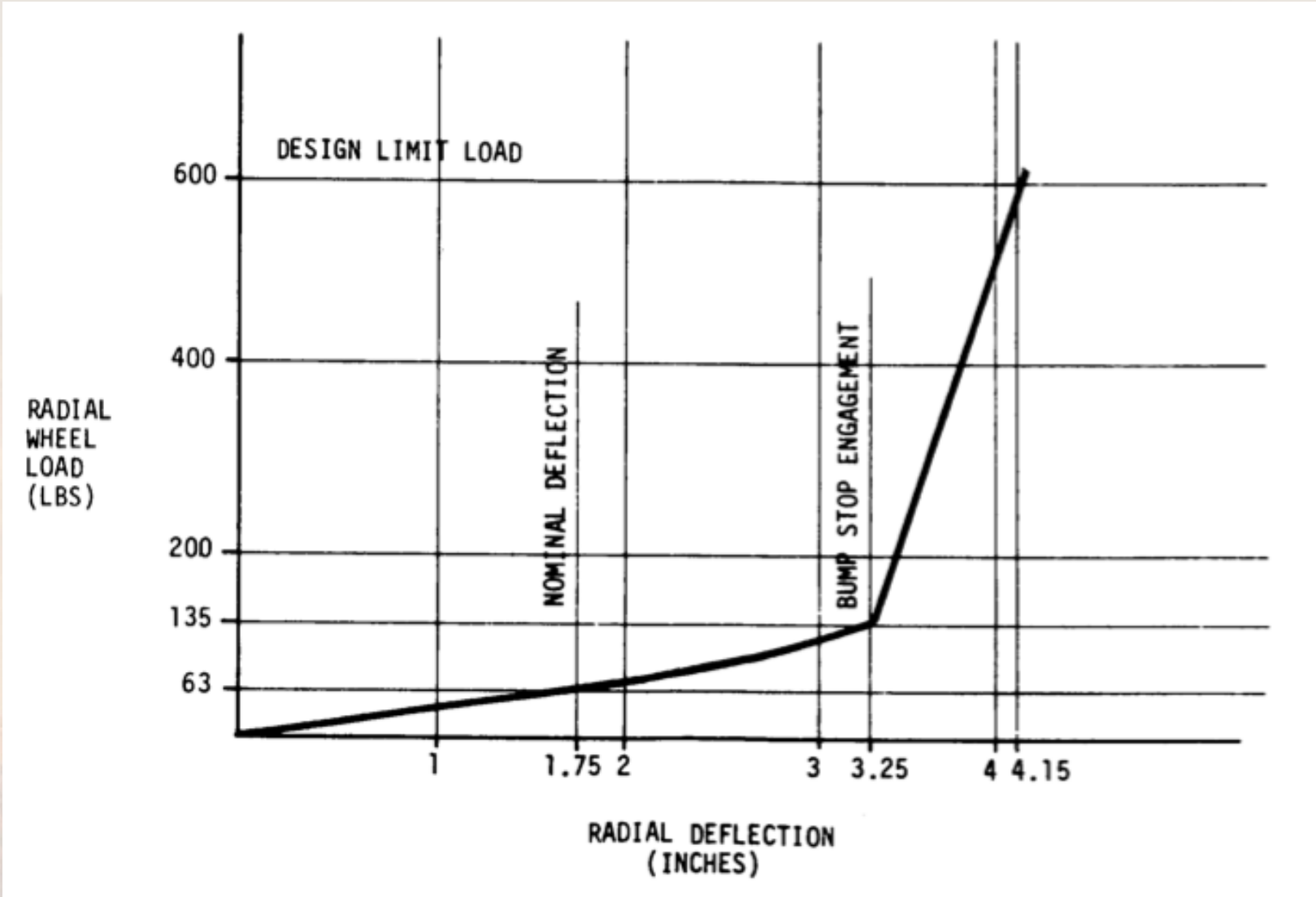
# LRV Wheel Design Details



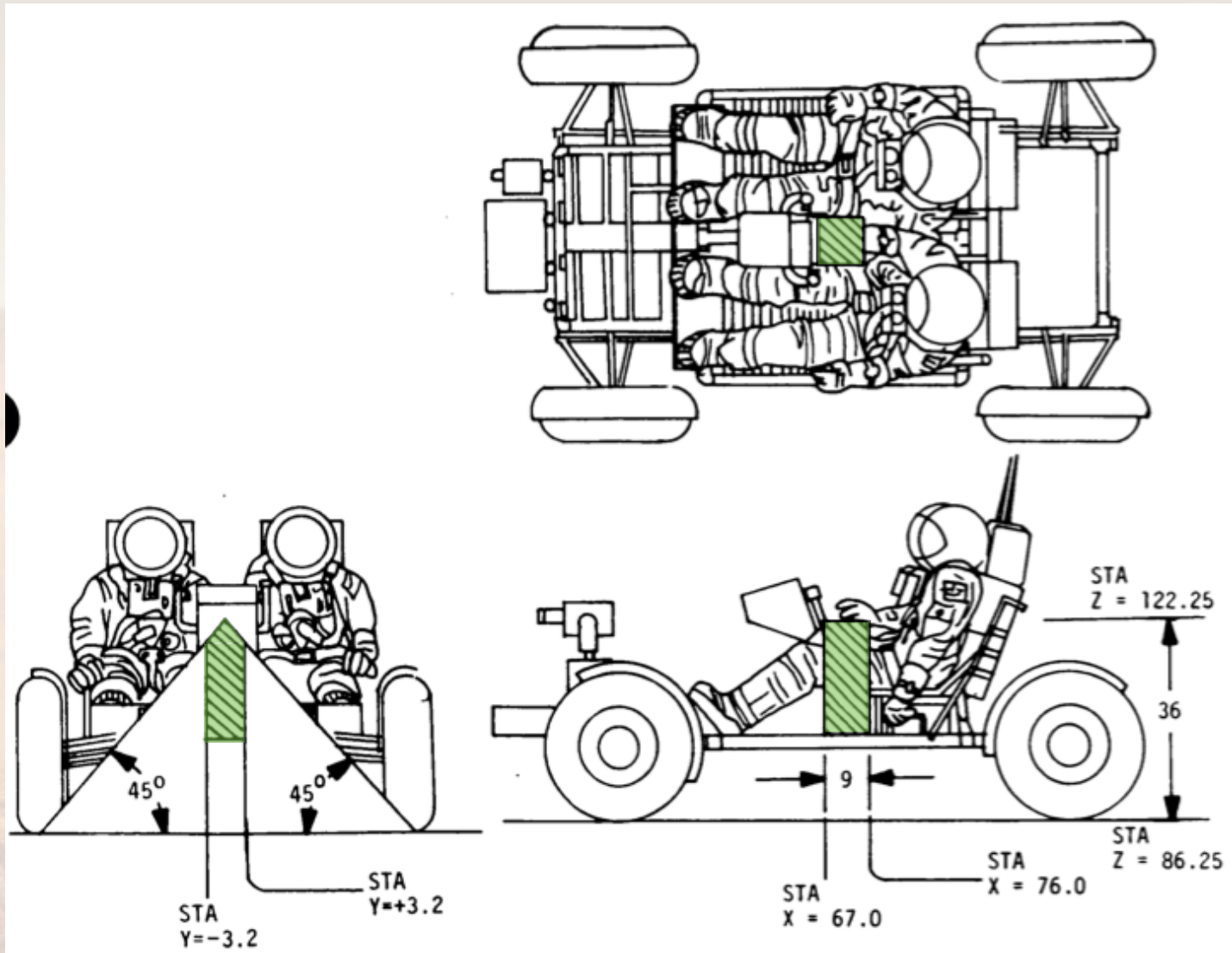
# LRV Wheel Testing



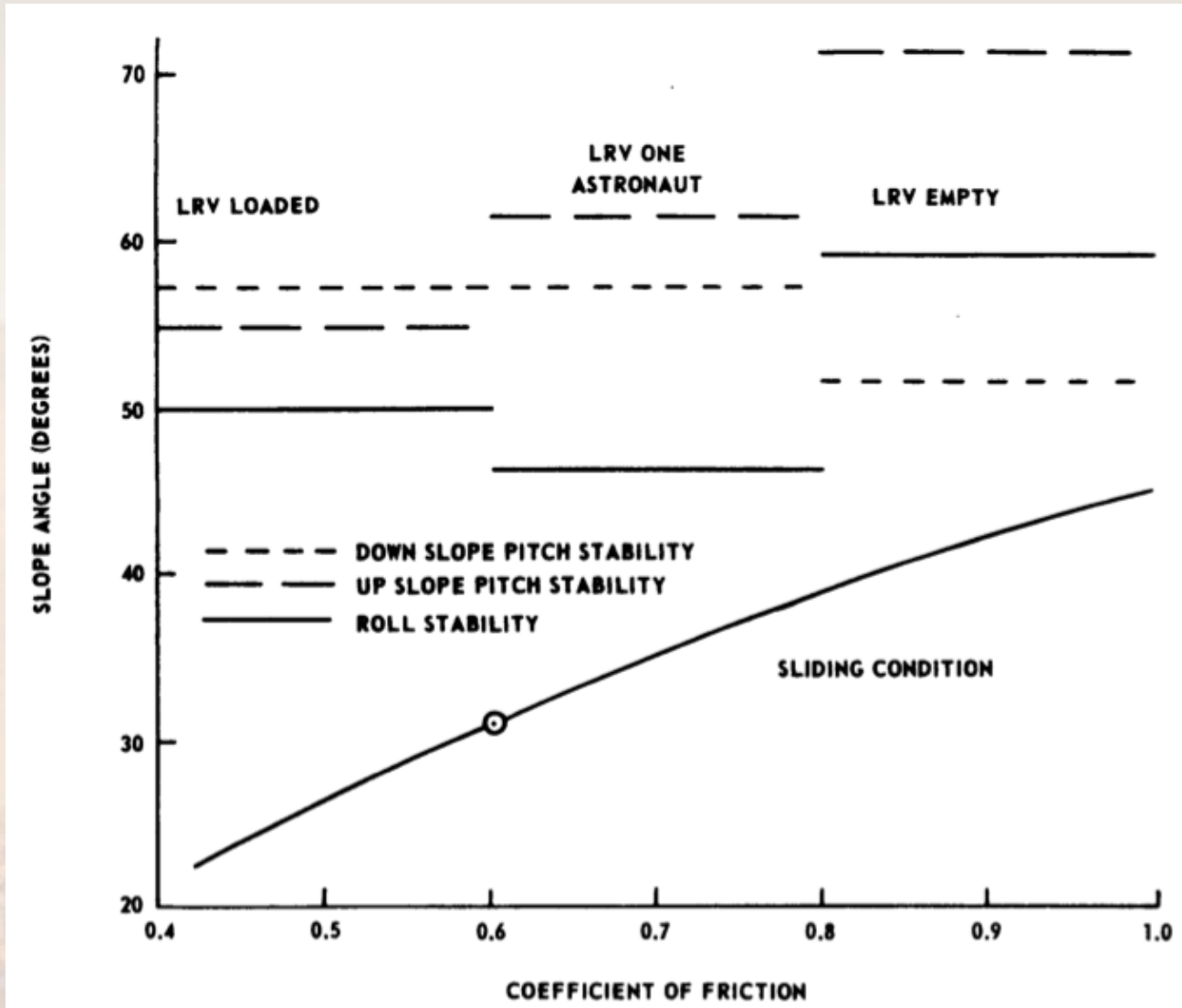
# LRV Wheel Deflection



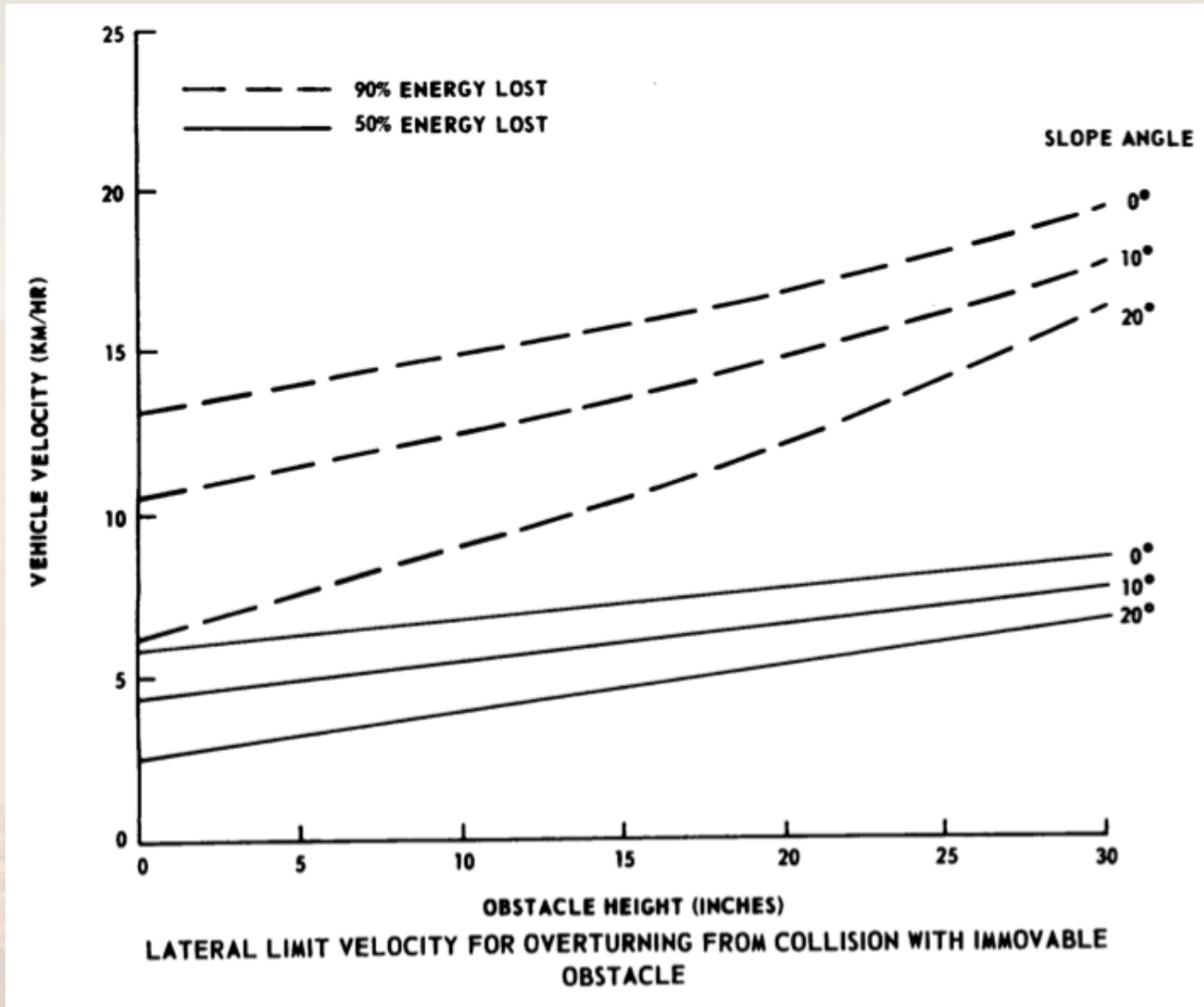
# Location of LRV Center of Gravity



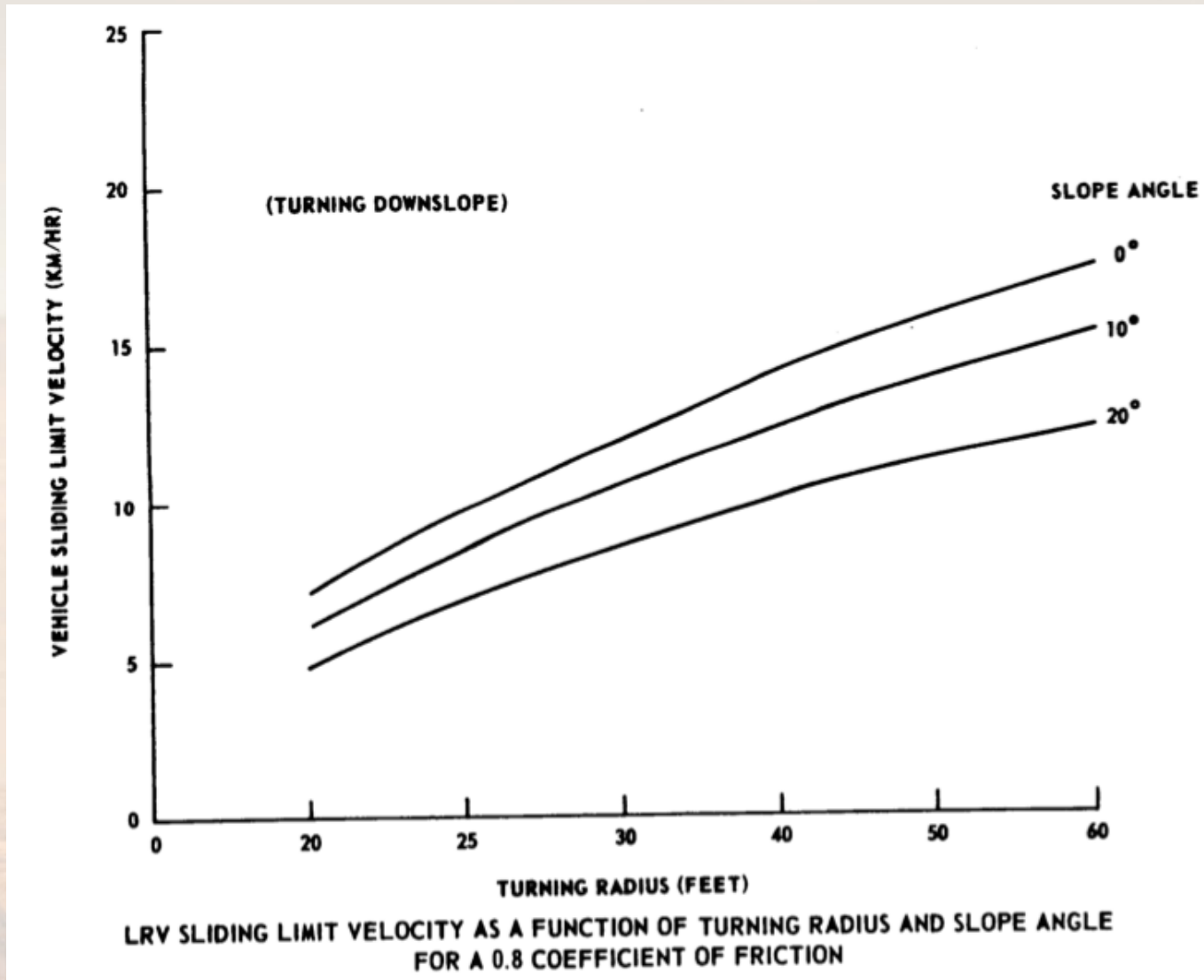
# LRV Static Stability



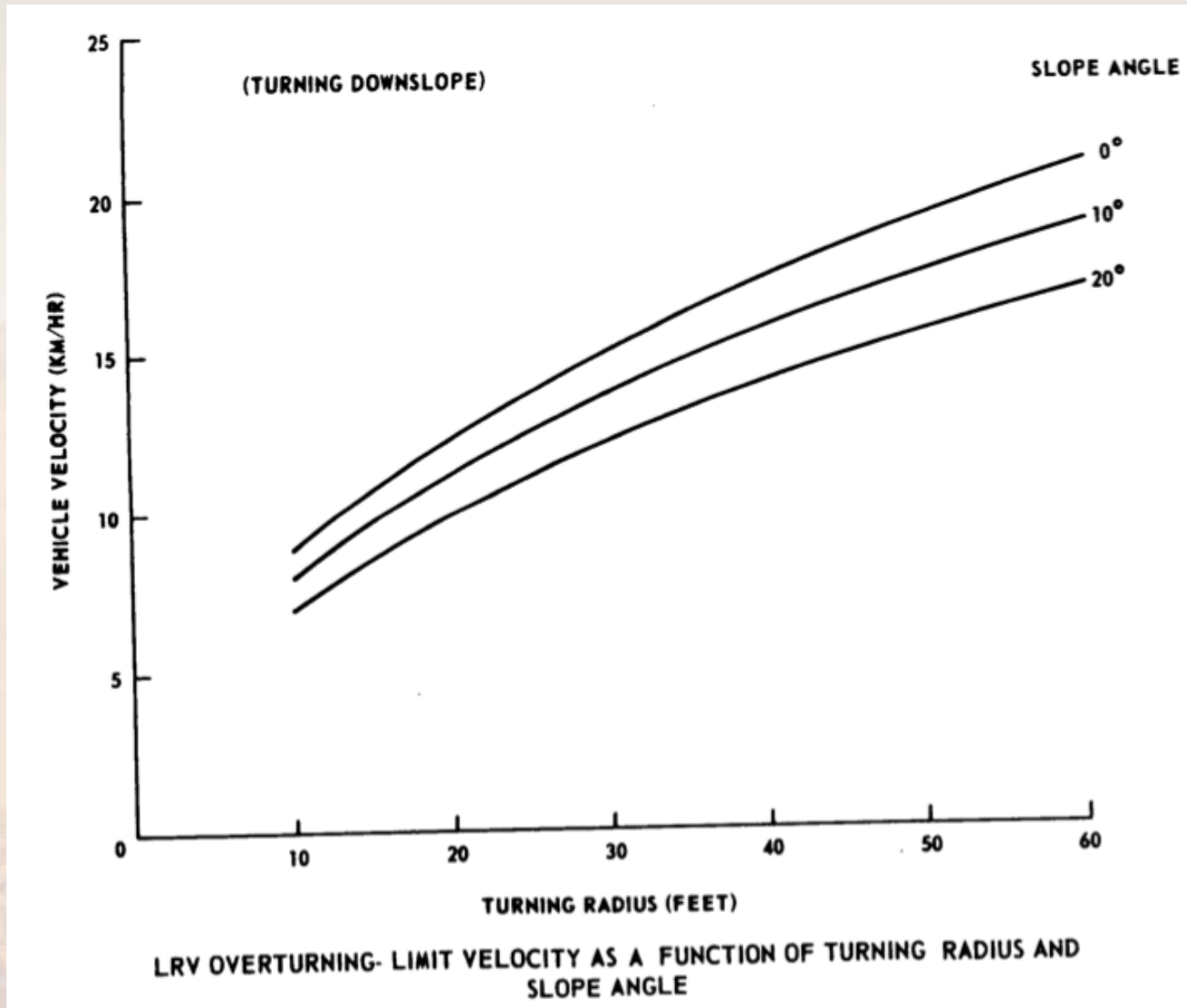
# Limiting Velocity for Obstacle Impact



# Sliding Limit in Turn ( $\mu=0.8$ )

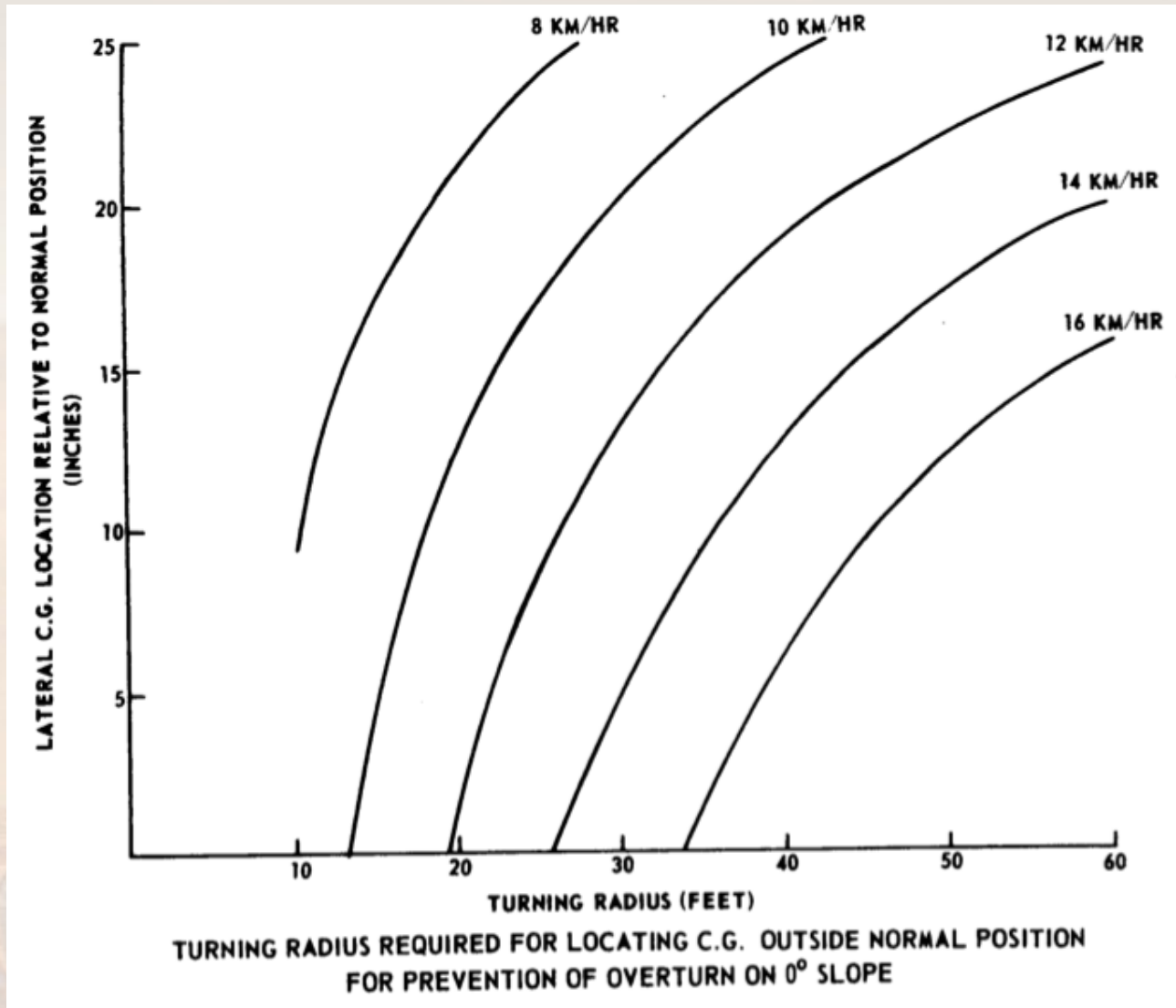


# LRV Overturn Limits in Turning

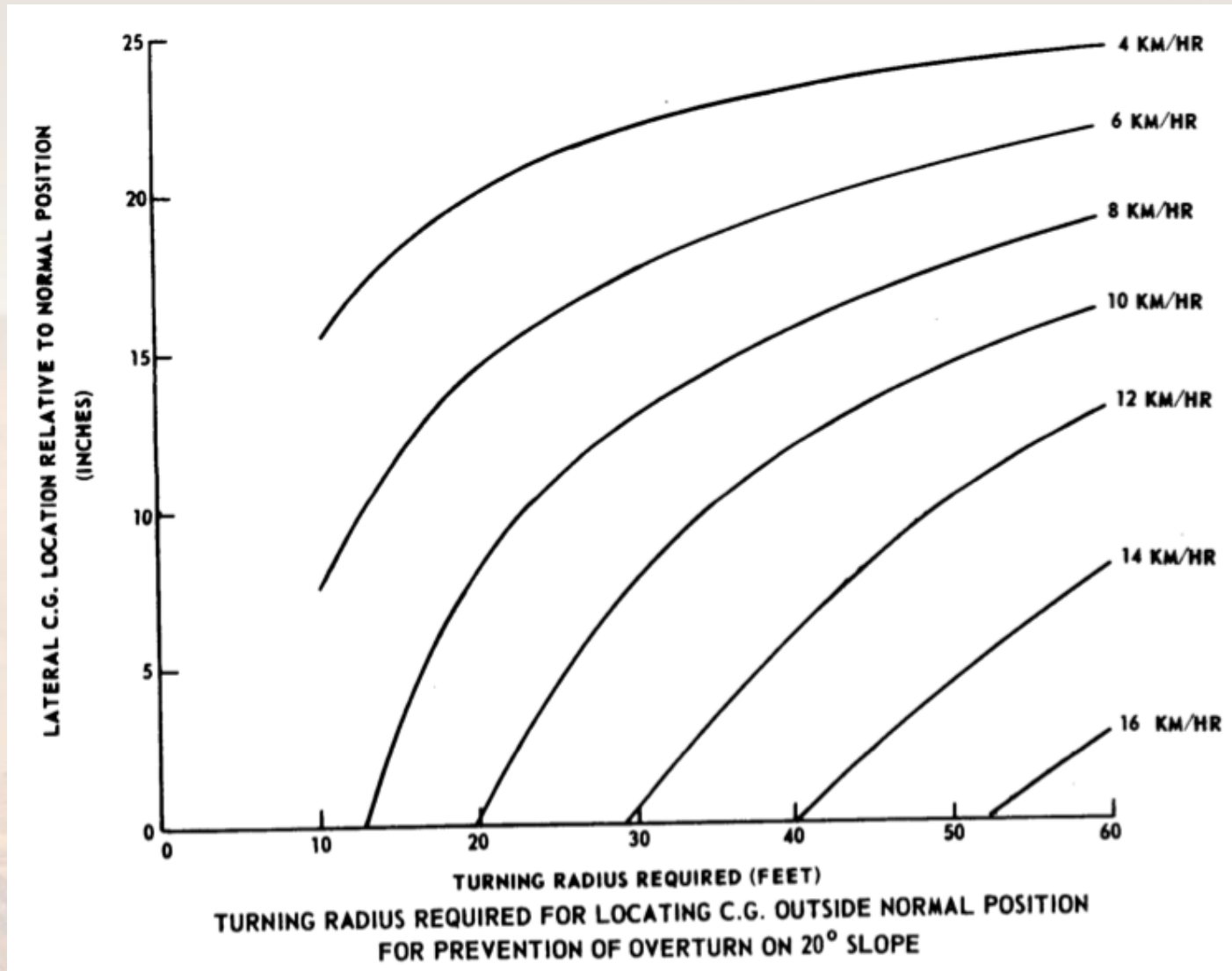




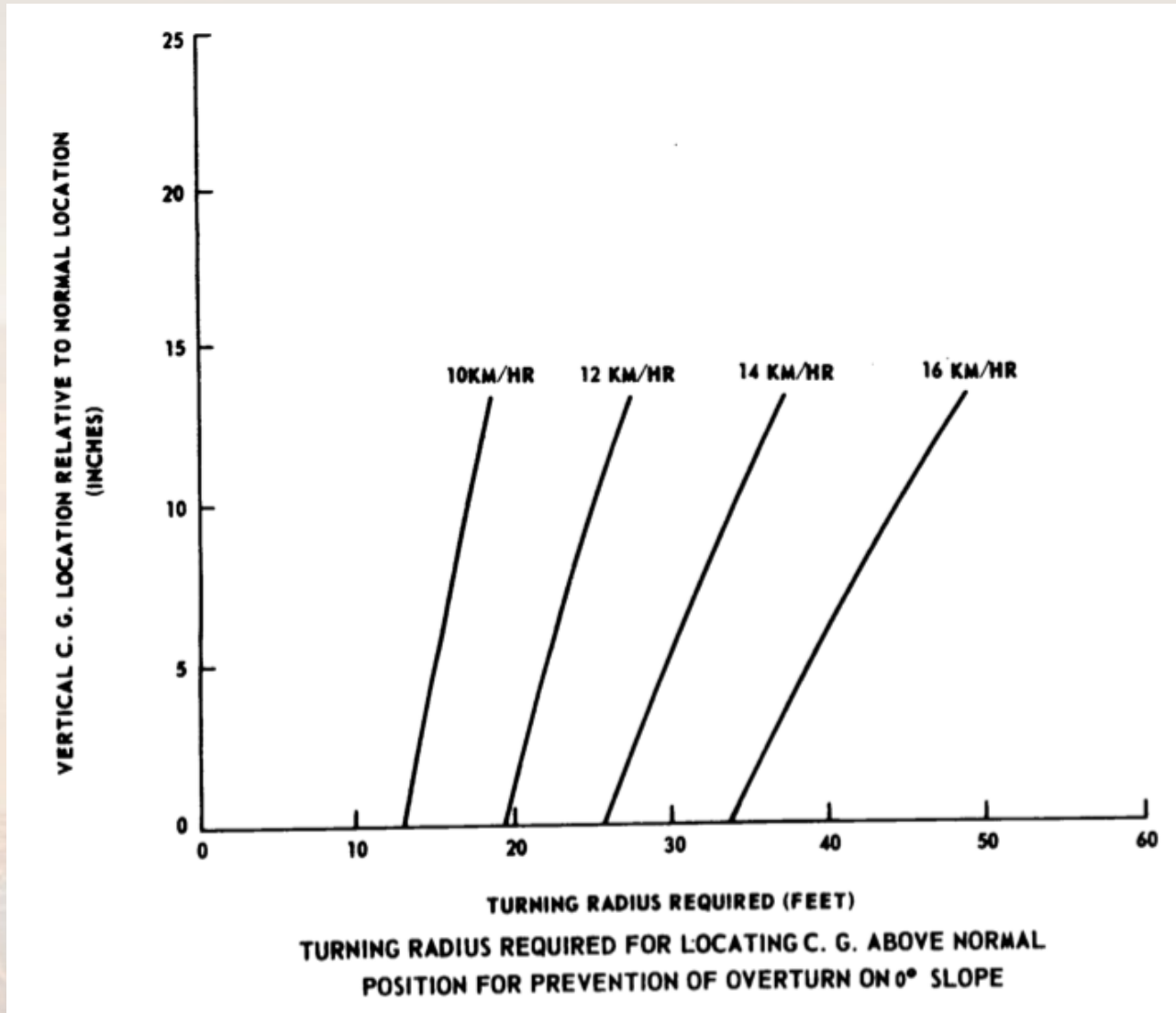
# Turning Radius with CG Shift - 0° Slope



# Turning Radius with CG Shift - 20° Slope



# Turning Radius Limit for Higher CG



# LRV Top Speed Limits (Struct. Fatigue)

## LURAIN TYPE (MIDRANGE)

## MAX ALLOWABLE SPEED

SMOOTH MARE

13 KM/HR

ROUGH MARE

8.5 KM/HR

HUMMOCKY UPLAND

8 KM/HR

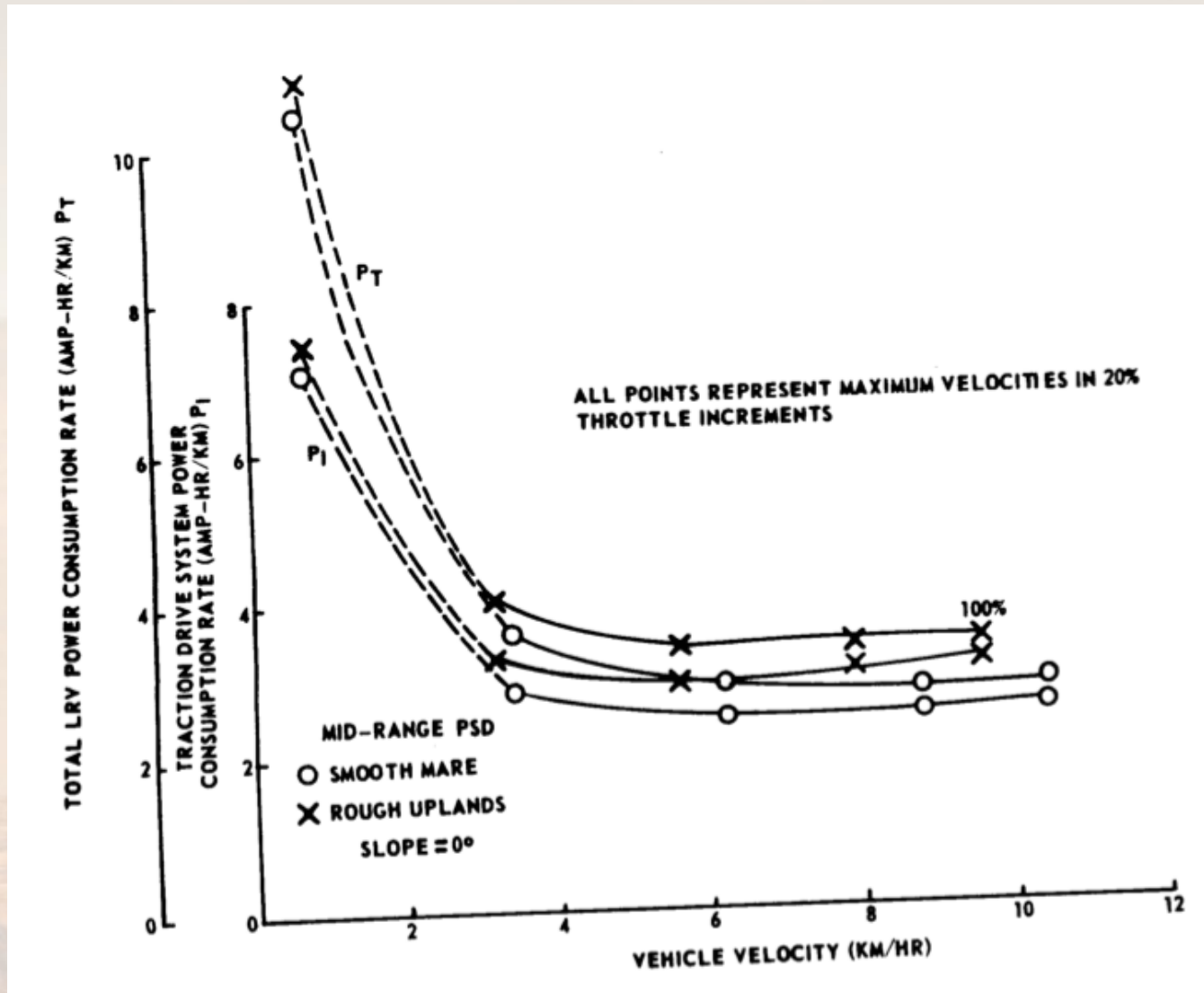
ROUGH UPLAND

7 KM/HR

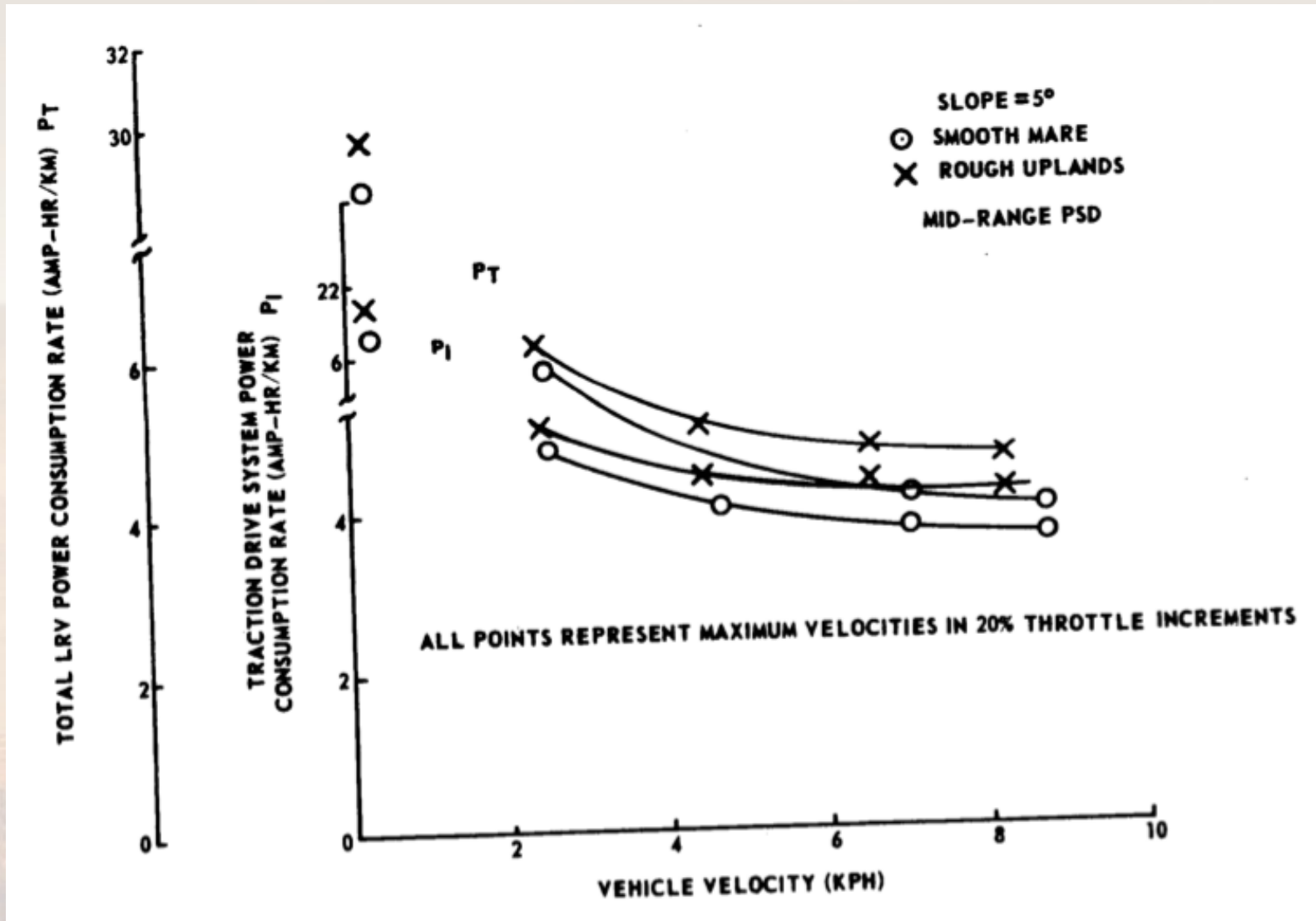
\*BASED ON CEI REFERENCE MISSION



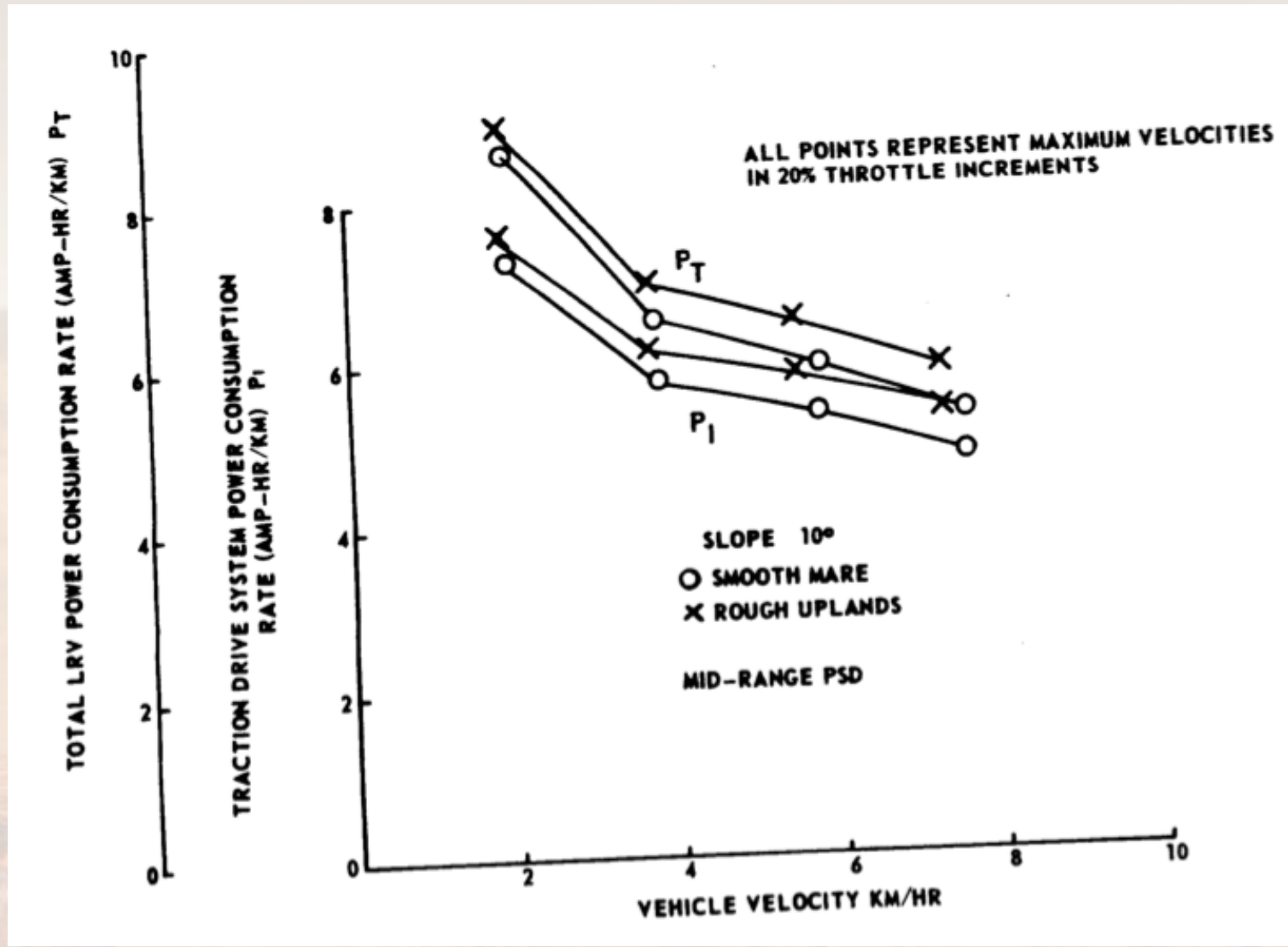
# LRV Power Requirements - 0° Slope



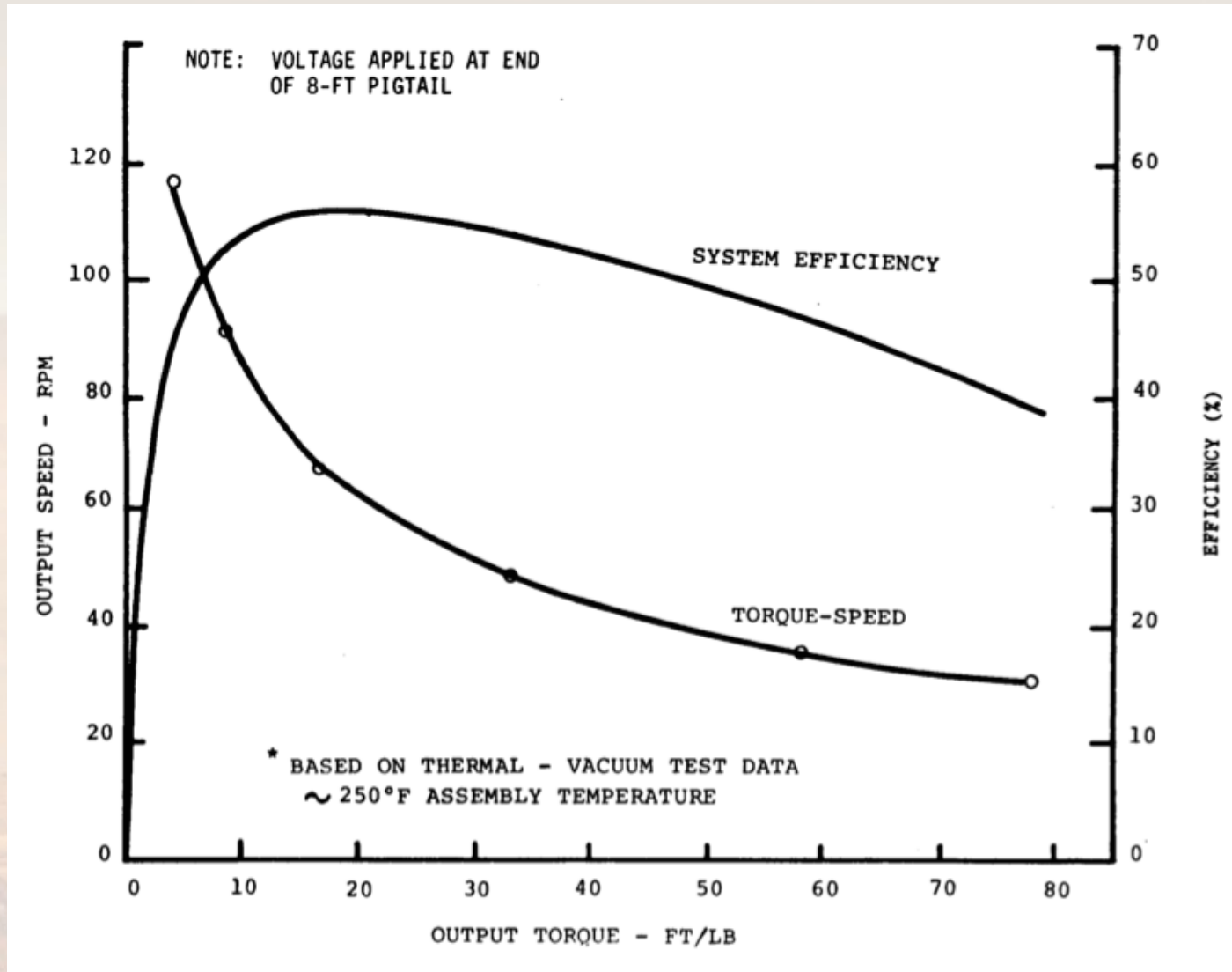
# LRV Power Requirements - 5° Slope



# LRV Power Requirements - 10° Slope

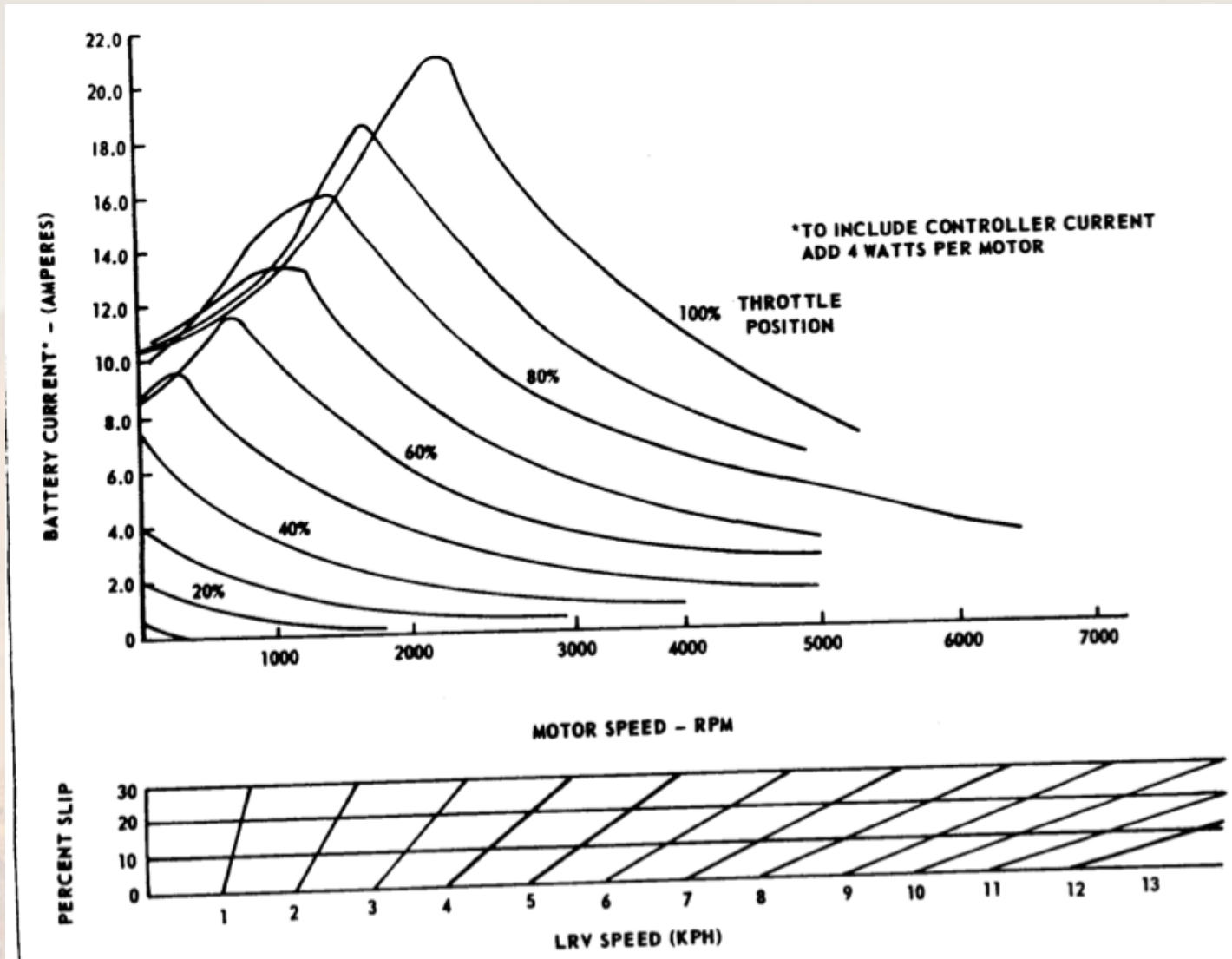


# LRV Traction Drive Performance

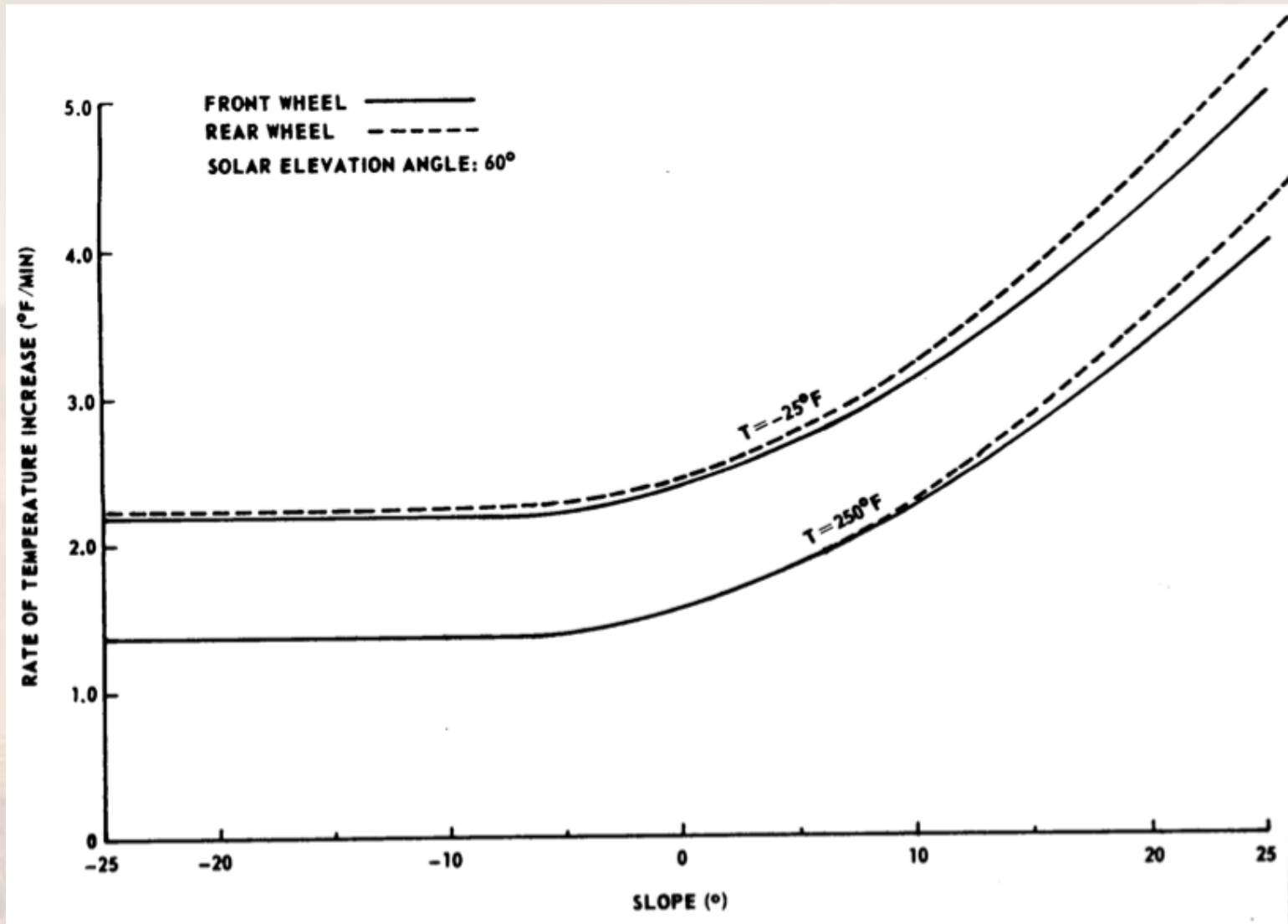




# Battery Current vs. Speed



# Wheel Motor Temperature vs. Slope



# Power Usage by System



<u>COMPONENT</u>	<u>POWER</u>	<u>TIME</u>
CONTROL & DISPLAY	10 WATTS	ENTIRE SORTIE
NAVIGATION (WARM UP)	90 WATTS	3 MINUTES
NAVIGATION (AFTER WARM UP)	40 WATTS	ENTIRE SORTIE AFTER WARMUP
DRIVE CONTROLLER (STANDBY)	23 WATTS	DURING PARKED PERIOD WITH DRIVE MOTORS ON



# LRV Wheel Loading



LRV-1 LOADED WEIGHT DISTRIBUTION:

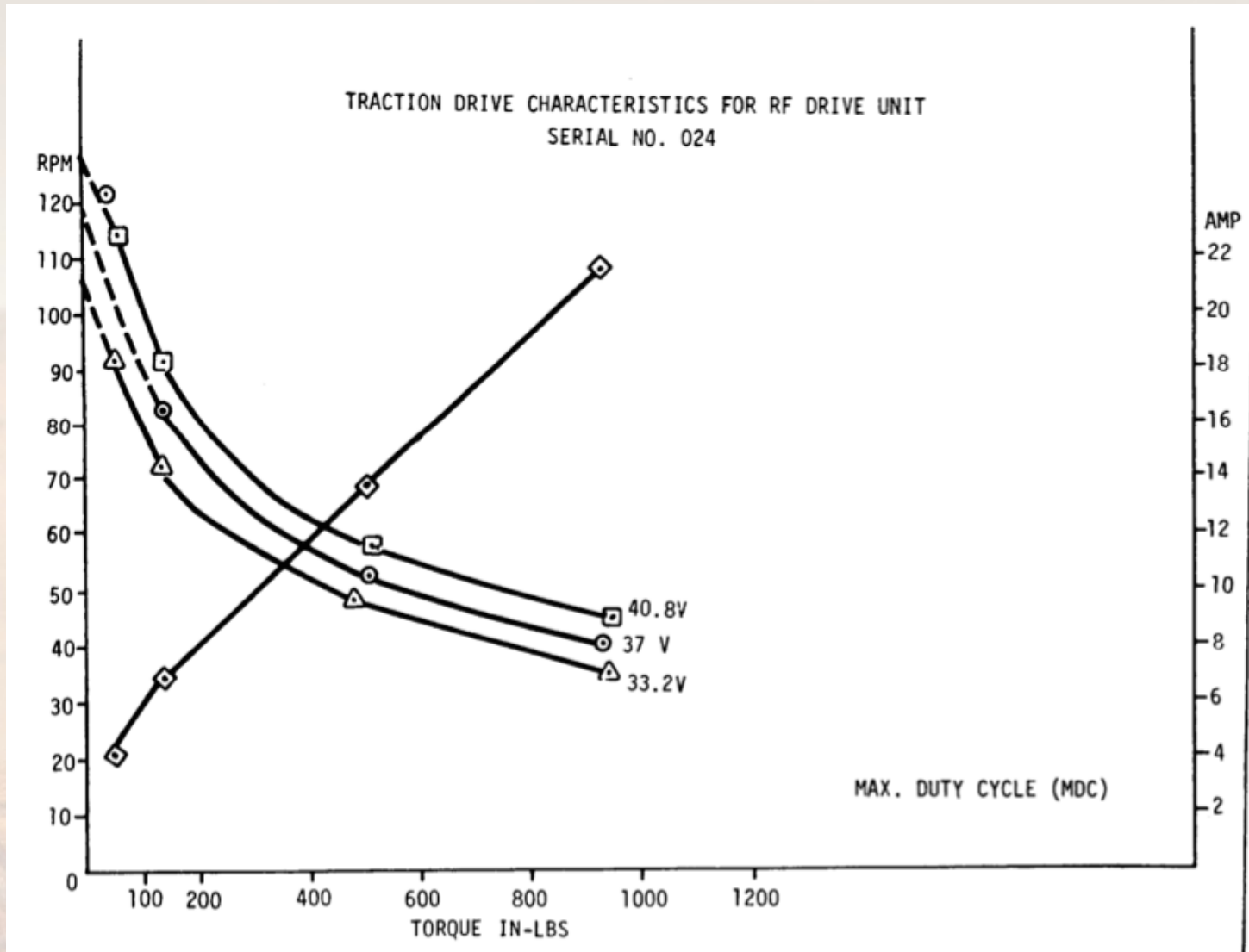
FRONT WHEELS	48.4%
REAR WHEELS	51.6%

LRV-1 LOADED WHEEL LOADING:

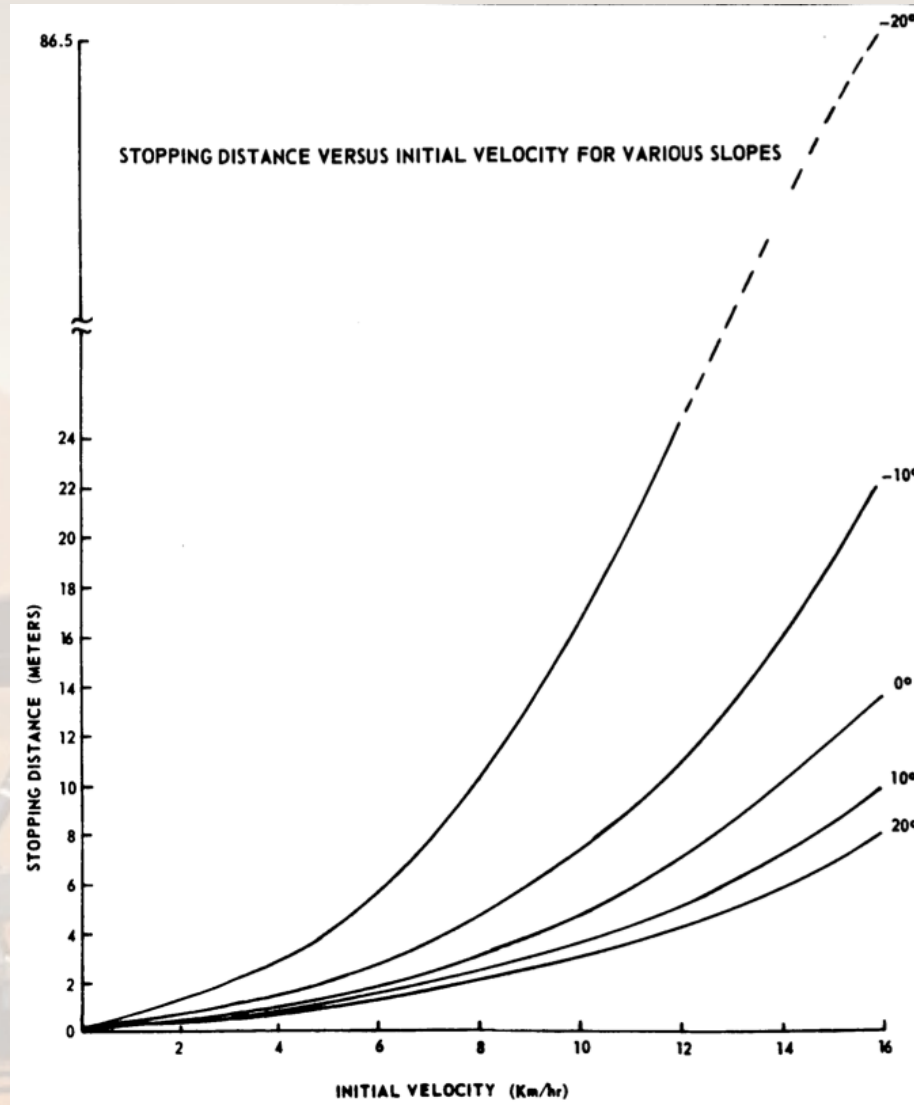
RIGHT FRONT	365.5 LBS	(24.0%)
LEFT FRONT	369.9 LBS	(24.3%)
RIGHT REAR	390.2 LBS	(25.7%)
LEFT REAR	394.8 LBS	(26.0%)



# Drive Motor Characteristics



# LRV Stopping Distance vs. Speed/Slope



# LRV Terrain Design Cases

- Crevasse crossing capability 70 cm
- Step obstacle climbing capability 35 cm
- Clearance under chassis 35 cm



# LRV Mobility Parameters

PARAMETER	
1. GROSS VEHICLE MASS	47.2 SLUGS (1520 LBS)
2. SUSPENDED VEHICLE MASS	44.2 SLUGS (1424 LBS)
3. WHEEL MASS	.745 SLUG (24 LBS)
4. WHEEL ROTATIONAL MOMENT OF INERTIA	2.2 SL-FT <sup>2</sup>
5. VEHICLE MOMENTS OF INERTIA	See Table 6-I
6. CG LOCATION	See Table 6-I
7. VERTICAL SUSPENSION RATE	14 LB/IN (0-9 INCHES) 500 LB/IN (< 0 OR > 9 IN)
8. VERTICAL DAMPING RATE	17.3 LB-SEC <sup>2</sup> /FT <sup>2</sup>
9. HORIZONTAL SUSPENSION RATE	51,000 LB/FT
10. HORIZONTAL SUSPENSION DAMPING RATE	2420 LB/(FT/SEC)
11. WHEEL RADIAL SPRING RATE	400 LB/FT (0-1.5 IN) 680 LB/FT (1.5-3 IN) 7300 LB/FT (3 IN)
12. WHEEL DAMPING RATE	2.5 LB/(FT/SEC)
13. WHEEL DIAMETER	32 INCHES
14. VEHICLE WHEEL BASE	90 INCHES

TABLE 6-IV SUMMARY OF LRV-1 MOBILITY PARAMETERS





# Apollo 15 Terrain (“Lurain”)



# Apollo 15 LRV



# Apollo 16 “Lunar Grand Prix”



# Apollo 16 LRV (image stabilized)



# References

- Glenn C. Miller, “The Lunar Cart” *7th Aerospace Mechanisms Symposium*, Houston, Texas, Sept. 2-3, 1972
- Alex B. Hunter and Bryan W. Spacey, “Lunar Roving Vehicle Deployment Mechanism” *7th Aerospace Mechanisms Symposium*, Houston, Texas, Sept. 2-3, 1972
- Boeing Company, “LRV Operations Handbook Appendix A (Performance Data)” NASA TM-X-66816, April 19, 1971

