Multi-Wheel Systems

- Obstacle climbing with multiwheel systems
- Planar rocker analysis
- Planar rocker-bogey analysis
- Suspension dynamics





788DF16L10.climbing

ENAE 788X - Planetary Surface Robotics

Slopes + Obstacles



wheel coefficient of friction with ground = M N= normal force to surface Z=urN=urWsild $T = \mu N = \frac{2}{r} = W \sin \alpha$ u Wess the = Wsing $tan \alpha = M$ T>Minit Nr (friction limited, not torque limited)

Assume

Longitudinal Dynamic Solutions

$$N_{1} = mg\left[\left(1 - \frac{a}{\ell}\right)\cos\theta - \left(\frac{h}{\ell} + \frac{r}{\ell}\right)\sin\theta - \frac{1}{g}\frac{dv}{dt}\right]$$
$$N_{2} = mg\left[\frac{a}{\ell}\cos\theta + \left(\frac{h}{\ell} + \frac{r}{\ell}\right)\sin\theta + \frac{1}{g}\frac{dv}{dt}\right]$$
$$T_{2} = \frac{N_{2}}{N_{1} + N_{2}}\left(mg\sin\theta + m\frac{dv}{dt}\right)$$
$$T_{1} = \frac{N_{1}}{N_{1} + N_{2}}\left(mg\sin\theta + m\frac{dv}{dt}\right)$$

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Slopes and Static Stability ENAE 788X - Planetary Surface Robotics

Normal and Shear Wheel Force w/Slope



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Slopes and Static Stability ENAE 788X - Planetary Surface Robotics

Four-Wheeled Vehicle Climbing a Wall



from Howard Eisen, "Scale and Computer Modeling of Wheeled Vehicles for Planetary Exploration" S.M. Thesis, Department of Aeronautics and Astronautics, Massachusetts Institute of Technology, May, 1990 UNIVERSITY OF Multi-Wheel Systems ENAE 788X - Planetary Surface Robotics

Wall Climbing EFvertical => N2 + MN, = W EFhorizontil => UN2 = N, EMrear axle => MN2r + MN, (l+r) = W(l-a) $N_2 + u^2 N_2 = W \Rightarrow N_2 = \frac{W}{1 + u^2} \Rightarrow N_1 = \frac{u}{1 + u^2} W$ $\frac{\mathcal{U}}{\mathcal{U}_{1+\mathcal{M}^{2}}}Wr + \frac{\mathcal{U}^{2}}{\mathcal{U}_{1+\mathcal{M}^{2}}}W(l+r) = W(l-q)$ $(r_{49}) M^2 + r M - (l_{-9}) = 0$

$$\mathcal{M} = \frac{-r \pm \sqrt{r^2 + 4(r+a)(l-a)}}{2(r+a)}$$
Let $d \equiv \frac{a}{r}$

$$\mathcal{M} = \frac{-1 \pm \sqrt{1 + 4(1+a)(\lambda-a)}}{2(1+a)}$$





Required Traction for Wall Climbing



from Howard Eisen, "Scale and Computer Modeling of Wheeled Vehicles for Planetary Exploration" S.M. Thesis, Department of Aeronautics and Astronautics, Massachusetts Institute of Technology, May, 1990 UNIVERSITY OF Multi-Wheel System

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Wheel Interaction with Slope



from Howard Eisen, "Scale and Computer Modeling of Wheeled Vehicles for Planetary Exploration" S.M. Thesis, Department of Aeronautics and Astronautics, Massachusetts Institute of Technology, May, 1990 UNIVERSITY OF Multi-Wheel Systems ENAE 788X - Planetary Surface Robotics

Equations for Slopes under Wheels



Sum of Horizontal forces:

 $\mu N_2 \sin \phi_2 + N_2 \cos \phi_2 + \mu N_1 \sin \phi_1 + N_1 \cos \phi_1 - W = 0$ Sum of vertical forces:

 $\mu N_2 \cos \phi_2 - N_2 \sin \phi_2 + \mu N_1 \cos \phi_1 - N_1 \sin \phi_1 = 0$ Sum of forces around the rear axle: $\begin{pmatrix} \mu N_2 r - W(L - a) + N_1 L \cos \phi_1 + \mu N_1 (r + L \sin \phi_1) = 0 \\ \text{WIVERSITYOF} \\ \text{MARYLAND} \\ \text{MARYLAND$

Bump/Slope Traction Requirements



from Howard Eisen, "Scale and Computer Modeling of Wheeled Vehicles for Planetary Exploration" S.M. Thesis, Department of Aeronautics and Astronautics, Massachusetts Institute of Technology, May, 1990 UNIVERSITY OF Multi-Wheel Systems ENAE 788X - Planetary Surface Robotics

Six-Wheel Articulated Body Rover



Model of Six-Wheel Vehicle



Sum of vertical forces:

$$N_3 + \mu N_2 + N_1 - Wf - Wb = 0$$

Sum of horizontal forces:

$$\mu N_3 - N_2 + \mu N_1 = 0$$

Sum of moments for front body around pitch axis

$$\mu N_1(r+e) + N_1(a+b+c) + \mu N_2(r+c) + -N_2e - Wf(b+c) = 0$$

Sum of moments for rear body around pitch axis

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$$Vbd + \mu N_3(r + e) - N_3(d + f) = 0$$



Navtest Rover with Walls and Slopes

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from Howard Eisen, "Scale and Computer Modeling of Wheeled Vehicles for Planetary Exploration" S.M. Thesis, Department of Aeronautics and Astronautics, Massachusetts Institute of Technology, May, 1990





Six-Wheel Rover, Slope Climbing



 Wall Angle (deg)

 from Howard Eisen, "Scale and Computer Modeling of Wheeled Vehicles for Planetary Exploration" S.M. Thesis,

 Department of Aeronautics and Astronautics, Massachusetts Institute of Technology, May, 1990

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Four-Wheel Rocker Suspension

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Planar Rocker Analysis



 \leq Forces: $N_1 + N_2 = W$ E Moment (rea axle) $\mathcal{T}_{1} + \mathcal{T}_{2} + N_{1} l \cos \theta = \mathcal{T}_{0} + \mathcal{W}[(l-a)\cos \theta + h\sin \theta]$

 $N_{i} l \cos \theta = T_{0} - Z_{i} - T_{2} + W \left[(l - q) \cos \theta - h \sin \theta \right]$ $N_{i} = \frac{T_{0} - 2_{i} - \overline{T_{2}}}{R_{cos} \Theta} + W \left(\frac{l - 9}{R} - \frac{h}{R} + cos \Theta \right)$ $N_2 = W \cdot N_1 = W \left(1 - \frac{l \cdot a}{R} + \frac{L}{T} t \cdot a \cdot \theta \right) - \frac{T_0 - 2_1 - 2_1}{I \cdot a \cdot \theta}$ $N_2 = W\left(\frac{9}{2} + \frac{h}{\ell} \tan \theta\right) + \frac{\tau_1 + \tau_2 - \tau_0}{\ell_0, \theta}$



Nondinensionalize To=WX = To=X

Tubel = Tr = Tubel = Tr $\overline{\cdot}$

 $\frac{N_{i}}{W} = 1 - \frac{9}{4} - \frac{h}{4} \tan \theta + \left(\frac{X}{4} - \frac{T_{i}}{W}\frac{r_{i}}{4} - \frac{T_{i}}{W}\frac{r_{i}}{4}\right) - \frac{T_{i}}{W}\frac{r_{i}}{4} - \frac{T_{i}}{W}\frac{$ $\frac{N_2}{W} = \frac{2}{7} + \frac{1}{7} t \cdot n \theta + \left(\frac{T_1}{W} + \frac{T_2}{T} + \frac{T_2}{W} + \frac{T_2}{T} + \frac{T_2}{T}\right) \frac{1}{c_0 \theta}$

Six-Wheel Rocker-Bogey Suspension



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Kinematics of Planar Rocker-Bogey

Planar Rocker - Bogey Analysis $N_{1} = \overline{P} N_{B} \left(\frac{I_{B} \cdot q_{B}}{I_{R}} - \frac{h_{B}}{I_{0}} + \frac{h_{B}}{I_{0}} + \frac{\theta_{B}}{I_{0}} \right) - \frac{\mathcal{T}_{1} + \mathcal{T}_{2}}{I_{B} \omega_{1} \theta_{R}}$ N2 = NB (To + TB to OB) + T, +T2 10 coile $(T_{B}=0 a lue y s)$ For rocker $N_{B} = \frac{T_{o} - T_{3}}{l_{o}\cos(\theta_{R} + \theta_{R})} + W\left(\frac{J_{R} - 9_{R}}{J_{R}} - \frac{h_{R}}{J_{R}} + \frac{h_{R}}{I_{R}} + \frac{h_{R}}{h_{R}} + \frac{h$ N3 = W ($\frac{q_R}{I_0} + \frac{h_R}{I_R} \stackrel{ton}{\longrightarrow} (\Theta_R + \Theta_R) + \frac{T_3 - T_0}{I_R \cos(\Theta_R + \Theta_{R_0})}$ $\int l_{R}^{2} - h_{B}^{2} - (l_{B} - q_{B}) = l_{2} \quad \theta_{R_{0}} = \sin^{-1} \frac{h_{B}}{l_{0}}$ $\frac{l_{B}}{l_{B}} = \frac{l_{R}}{l_{B}} = \frac{l_{R}}{l$

Normalize by Wand la

 $\frac{N_3}{W} = \frac{a_R}{I_R} + \frac{h_R}{I_R} \stackrel{\text{ten}}{=} \left(\theta_R + \theta_{R_0} \right) + \left(\frac{T_3}{W} \frac{r_2}{I_B} \frac{I_B}{I_R} - \frac{\chi}{I_B} \frac{I_B}{I_R} \right) \frac{1}{\cos(\theta_0 + \theta_{R_0})}$

 $\frac{N_B}{W} = 1 - \frac{\alpha_R}{I_R} - \frac{h_R}{I_R} \tan \left(\Theta_R + \Theta_{R_0} \right) + \left(\frac{\lambda}{I_B} \frac{l_B}{I_R} - \frac{T_3}{W} \frac{r_3}{I_B} \frac{l_A}{I_R} \right) \frac{1}{\cos \left(\Theta_R + \Theta_{R_0} \right)}$

 $\frac{N_{i}}{W} = \frac{N_{B}}{W} \left(1 - \frac{q_{B}}{I_{B}} - \frac{h_{B}}{I_{B}} t_{a} - \theta_{B} \right) - \left(\frac{T_{i}}{W} \frac{r_{i}}{I_{B}} + \frac{T_{Z}}{W} \frac{r_{Z}}{I_{B}} \right) \frac{1}{\cos \theta_{B}}$ $\frac{N_2}{W} = \frac{N_B}{W} \left(\frac{q_B}{I_B} + \frac{h_B}{I_B} \tan \theta_B \right) + \left(\frac{T_1}{W} \frac{r_1}{I_B} + \frac{T_2}{W} \frac{r_2}{I_B} \right) \frac{1}{\cos \theta_B}$

Suspension Systems - Current planetary rovers (e.g., MER, MSL) have little or no shock absorption - Notional car suspension e damper)-wheel one wheel first Analyze mz+cz+kz=cz+kz. Undamped force-free solution Z=Z cos w,t m=+ k==0 $-m\omega_{n}^{2}+k=0$ $\ddot{z}=-\ddot{z}\omega_{n}^{2}\cos\omega_{n}t$ $\omega_n = \int_m^k k$

Rover Example MTOT = 500 kg = wheel m = MTOT = 125 kg d = deflection of suspension (at rest) ~ 0.1m Earth Moon $k = \frac{F}{d} = \frac{mg}{d}$ 2000 Nm k 12,250 m $\omega_n = \int_{m}^{k}$ Wn 9.9 rad/sec 4 rad/ $f_0 = \frac{\omega_n}{2\pi}$ 0.64 Hz 1.6 Hz fn derit = critical distance between bumps 4.3m @10kph levit 1.8m (2.8 m/ser) $= \frac{V}{f}$

Multicheel Analysis Responses to two wheels hitting a bump bounce Imo excite both of these mades Equation of Motion $\begin{array}{cccc}
 & Equation of Motion \\
 & (935 uming no damping) \\
 & k_1 & Bounce: m & +k_p (2-l, \theta) + k_r (2+l_2 \theta) = 0 \\
 & k_1 & (5mall angles) \\
 & front \end{array}$ Pitch: Iy & + k, l, (2-l, 0) + k, l2 (2+ 120)=0 front let Iy = mry ry = radius of gyration Solve this set of coupled differential equs $D_{1} = \frac{k_{p} + k_{r}}{m}$ $D_{2} = \frac{k_{r} J_{2} - k_{p} J_{1}}{m}$ $D_{3} = \frac{k_{p} J_{1}^{2} + k_{r} J_{2}^{2}}{I_{y}}$

Rewrite in terms of Di, Dz, P3 Dz = coupling coefficient \vec{z} + $D_1 \neq + D_2 \hat{\theta} = 0$ Equations are independent if $D_2 = 0 \Rightarrow k_f l_1 = k_r l_2$ $\ddot{\Theta} + D_3 \Theta + \frac{P_2}{r_y^2} = 0$ If Dz=0, force @ CG only produces bounce an = /D, force elsewhere produces pitch $\omega_n = \sqrt{P_3}$ Assume D2 70 $\Theta = \Theta \cos \omega_n t$ Z= Z cos Wat $\begin{pmatrix} D_1 - \omega_n^2 \end{pmatrix} \begin{pmatrix} Z_1 + D_2 \end{pmatrix} \begin{pmatrix} D_2 \end{pmatrix} = 0 \\ D_1 - \omega_n^2 \end{pmatrix} \begin{pmatrix} D_1 - \omega_n^2 \end{pmatrix} \\ \frac{D_2}{r_1^2} \begin{pmatrix} Z_1 + (D_3 - \omega_n^2) \end{pmatrix} \begin{pmatrix} D_2 \end{pmatrix} = 0 \end{pmatrix} \begin{pmatrix} D_2 \\ r_1^2 \end{pmatrix}$ $\begin{array}{c} D_2 \\ D_3 - \omega_n^2 \end{array} = 0$

 $\omega_{n}^{2} = 17.33 \pm 10.43$ $\omega_{n_{1}} = 2.63' \frac{4}{sec} \Rightarrow 0.42 H_{z}$ $\omega_{n_{2}} = 5.67' \frac{4}{sec} \Rightarrow 0.84 H_{z}$

Add in Tire Mass & Stiffness



Sprung Mass
$M_s \ddot{z}_s + C_s (\dot{z}_s - \ddot{z}_u) + k_s (z_s - \ddot{z}_u) = 0$
Unsprung Mass
$m_{y} = \frac{1}{2} + (g(z_{y} - z_{s}) + k_{s}(z_{y} - z_{s}))$
+ $C_4 = \frac{1}{2} + k_4 = F(E) = C_4 = \frac{1}{2} + \frac{1}{2} + \frac{1}{2}$
Undamped Force - Free Solutions
$M_s z_s + k_s (z_s - z_u) = 0$
$m_{u} = \frac{1}{2} + k_{s} (z_{u} - z_{s}) + k_{u} = 0$
Zs=Z, cos wht Zu=Zu cos wht
$k_s - m_s \omega_n^2 - k_u = 0$
$-k_s = k_s + k_y - m_y \omega_n^2$

$$\begin{split} \omega_{n}^{4} (m_{u} m_{s}) + \omega_{n}^{2} (-m_{s} k_{s} - m_{s} k_{u} - m_{u} k_{s}) + k_{s} k_{u} = 0 \\ \omega_{n_{1}} = \frac{+B_{u}^{4} - \sqrt{B^{2} + 4AC}}{2A} \qquad \omega_{n_{2}} = \frac{B + \sqrt{B^{2} - 4AC}}{2A} \\ A = m_{u} m_{s} \qquad B = m_{s} (k_{s} + k_{u}) + m_{u} k_{s} \qquad C = k_{s} k_{u} \\ E_{x onple} : \qquad m_{s} = 100 \ k_{g} \qquad m_{u} = 25 \ k_{s} \\ k_{s} = 2000 \ N/m \qquad k_{u} = 10,000 \ N/m \\ A = 2500 \ k_{g}^{2} \qquad B = 1.25 \times 10^{6} \ k_{g}^{2}/s_{er}^{2} \qquad C = 2 \times 10^{7} \ N_{m}^{2} \\ \omega_{n_{1}} = 4.76 \ \frac{c_{ad}}{s_{ec}} \Rightarrow 0.8 \ H_{2} \iff Suspension \\ frequency \\ \omega_{n_{2}} = 28.8 \ \frac{r_{ad}}{s_{ec}} \Rightarrow 3.5 \ H_{2} \iff w_{heel} \ \frac{s_{s}^{4} (A^{2} + A^{2})}{s_{er}^{4}} \end{split}$$

Wheel- Soil Interaction in Turn Slip ratios $S = \frac{\Gamma \omega - V_X}{\Gamma \omega} \quad \text{for driving: } |\Gamma \omega| > |V_X|$ S= TW-Vx for broking: (IW/</Vx) $Sh_{p} \quad \text{angle} \quad \alpha = t = n' \left(\frac{V_{y}}{V_{x}} \right)$ -15551 wheel sinkeye P(z) = (ke + ky) z" $Z(\theta) = r(\cos \theta - \cos \theta_r)$ θ_{λ} for static sinkage Of = Of = Os Z(0)=r(cos 0 - cos 0s) $P(\Theta) = \left(\frac{k}{b}c_{+} \neq g\right) r^{n} \left(c_{0} \Theta - c_{0} \Theta \Theta_{s}\right)^{n}$ Given weight Won wheel, $W = \int \frac{\Theta_s}{P(\Theta) \, b \, r} \, \cos \Theta \, d\theta = r \frac{n + i}{(k_c + k_s b)} \int \frac{\Theta_s}{(i \circ s \, \Theta - c \circ s \, \Theta_s)} \frac{1}{(i \circ s \, \Theta - c \circ$ cos Odo Static sinkage Zs = r (1-cos Os)





